

G4S

Fully relativistic positioning for the Galileo for Science (G4S) project

GALILEO FOR SCIENCE

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The Galileo for Science project

G4S

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The full project is described in another presentation: **POSTER X3.105**



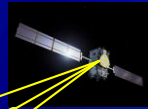
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Accurate orbit determination for Doresa and Milena

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Use of a Relativistic Positioning System (RPS) based on emitters fixed to the surface of the earth



This represents a Fully RPS based on the full exploit of Special and General Relativity and on the concept of space-time



In the case of GNSS satellites the positioning is traditional, very close to that of Nautical Astronomy, and the effects of Special and General Relativity are introduced as perturbations, i.e. as corrections to a Newtonian formulation

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The Principle of the measurement

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The counting of the pulses for a set of different emitters whose positions on the earth and periods are assumed to be known, is used to provide null emission, or light, coordinates for the receiver on a satellite



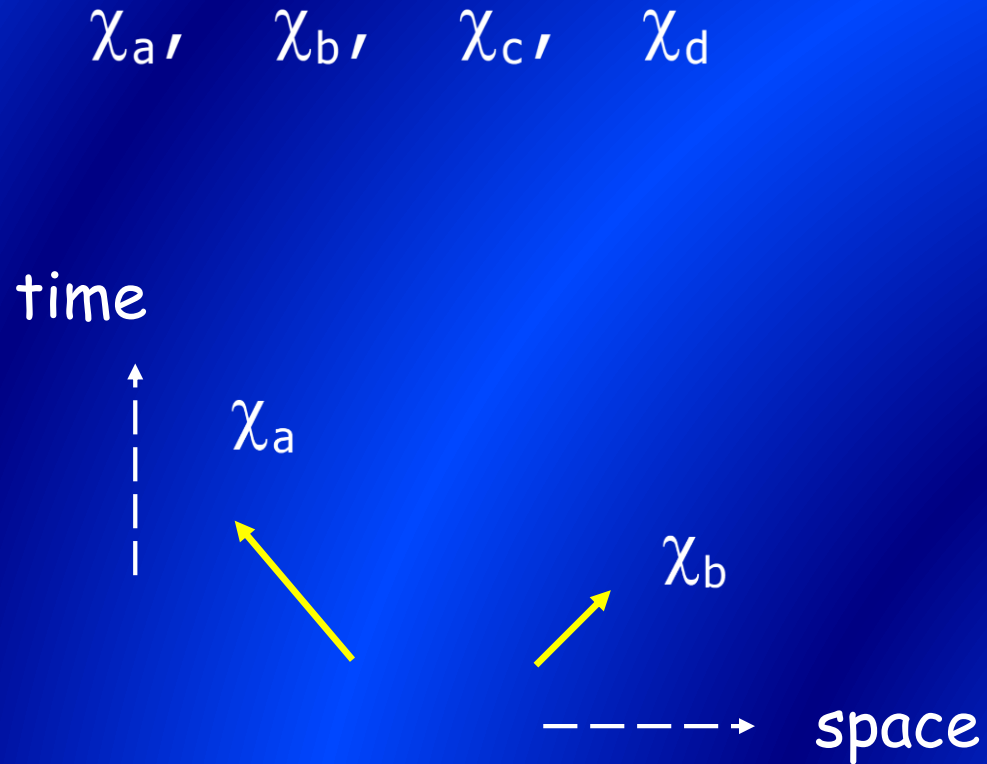
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The measurement of the proper time intervals between successive arrivals of the signals from the various emitters is used to give the final localization of the receiver, within an accuracy controlled by the precision of the onboard clock

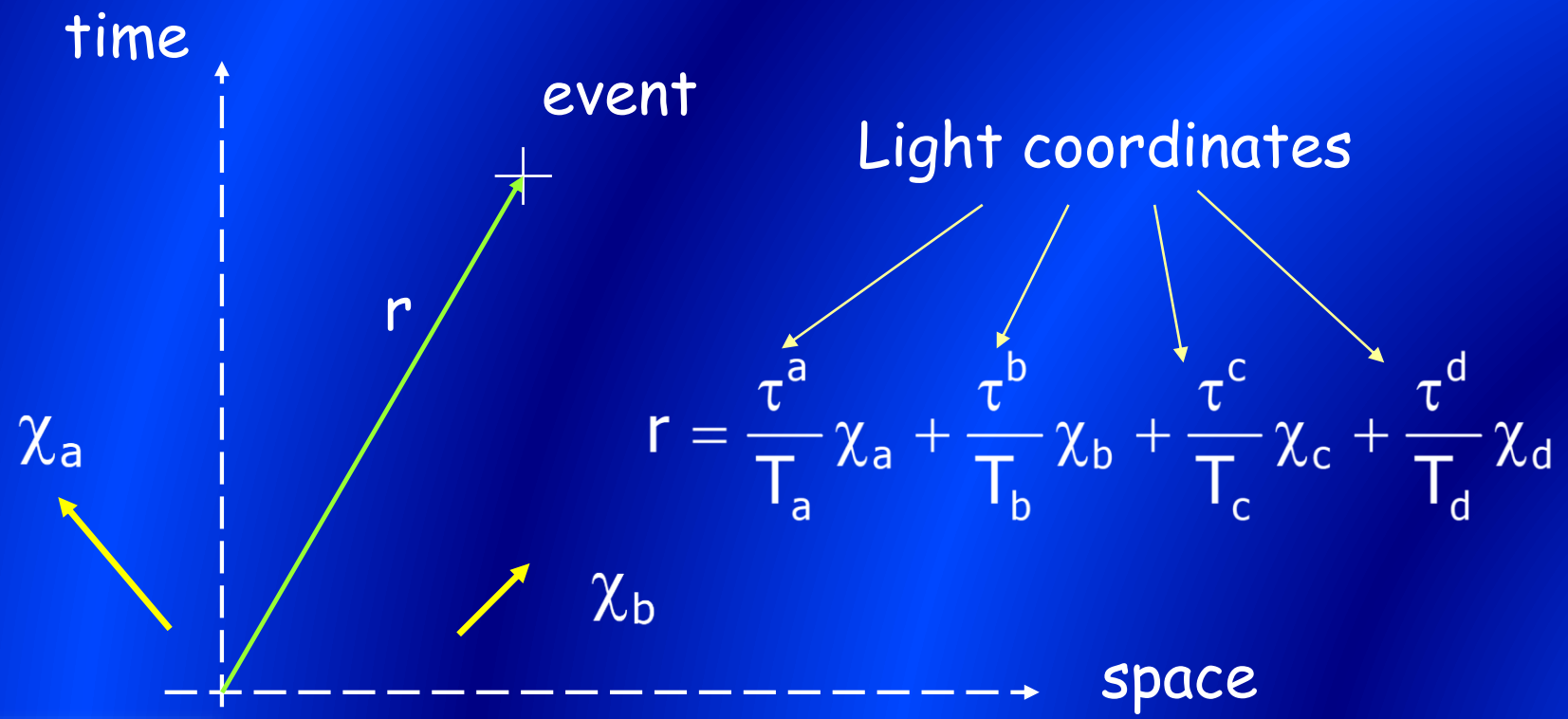


Null bases for space-time

Null vectors corresponding to the space-time directions of electromagnetic signals emitted from four independent sources comprise a null base



A position in space-time in terms of null wave-vectors



Wavevectors in terms of an ordinary timelike base

Null geodesics have null tangent wave-vectors

$$\chi = cT(1, \cos \alpha, \cos \beta, \cos \gamma) = cT(1, \hat{n})$$



 Period of the signal Direction cosines

$$\chi^2 = 0$$

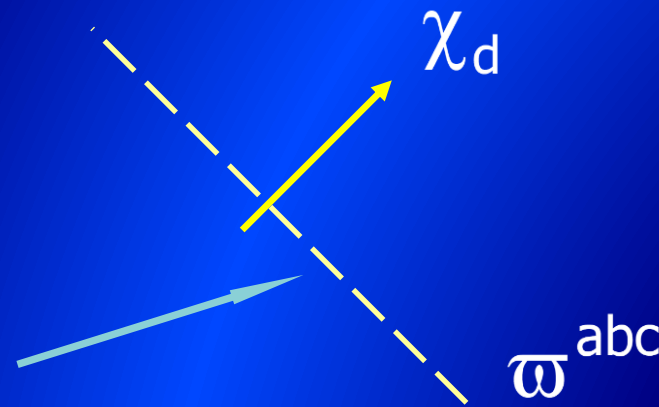


Null wave fronts

Null hyperplane
(in flat space-time
and source at infinity)

$$\omega^{abc} = \varepsilon^{abcd} \chi_d$$

Null hypersurface in
three dimensions



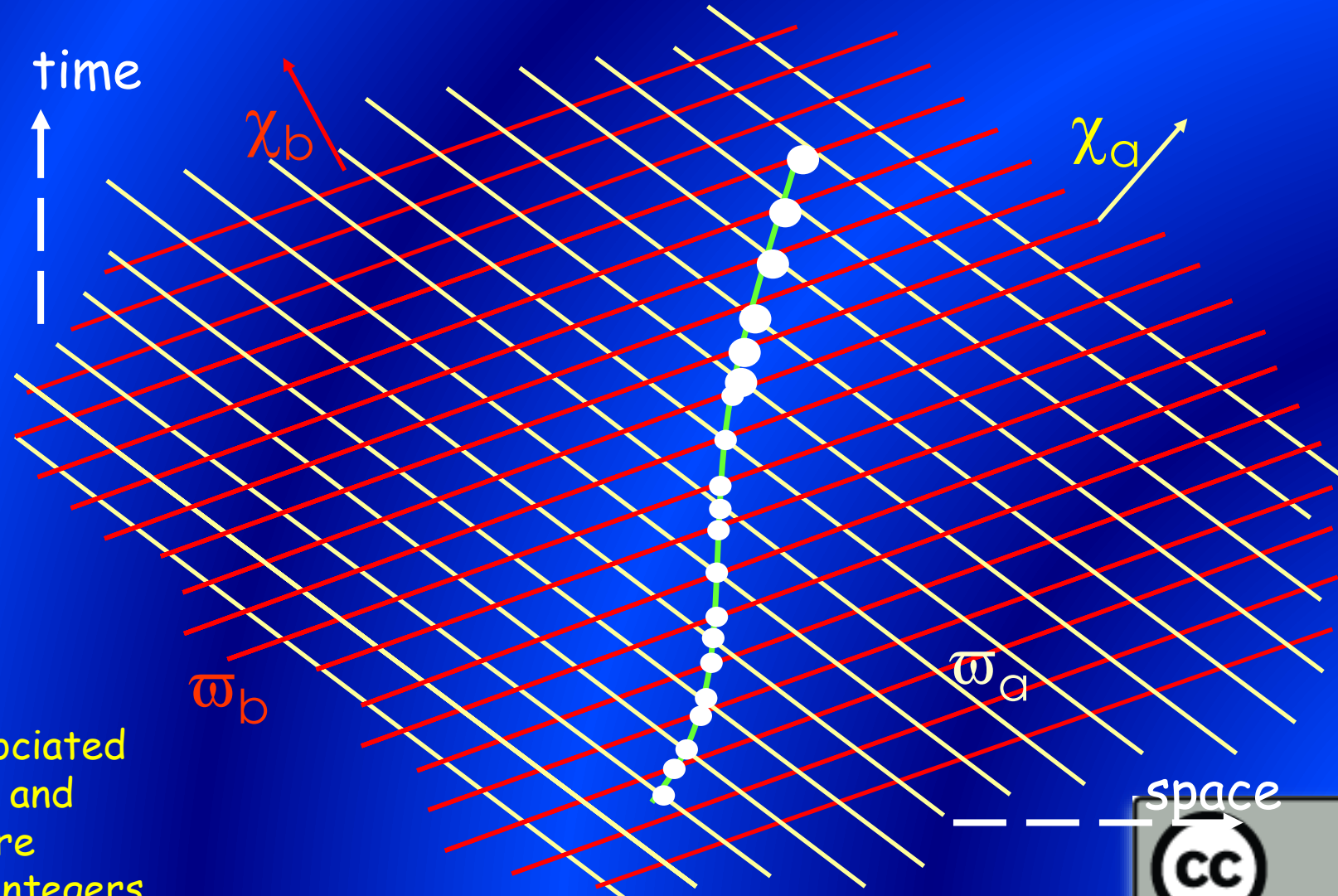
A null grid mapping space-time

— world-line
of a receiver

● arrival of
a signal

Interval between
arrivals measured
by proper time of
the receiver

Every wavefront is associated
with an integer number and
the knots of the grid are
identified by a pair of integers



Light coordinates of an event

$$\tau_{a,b,c,d} = [(n + x)T]_{a,b,c,d}$$

integer

fractional

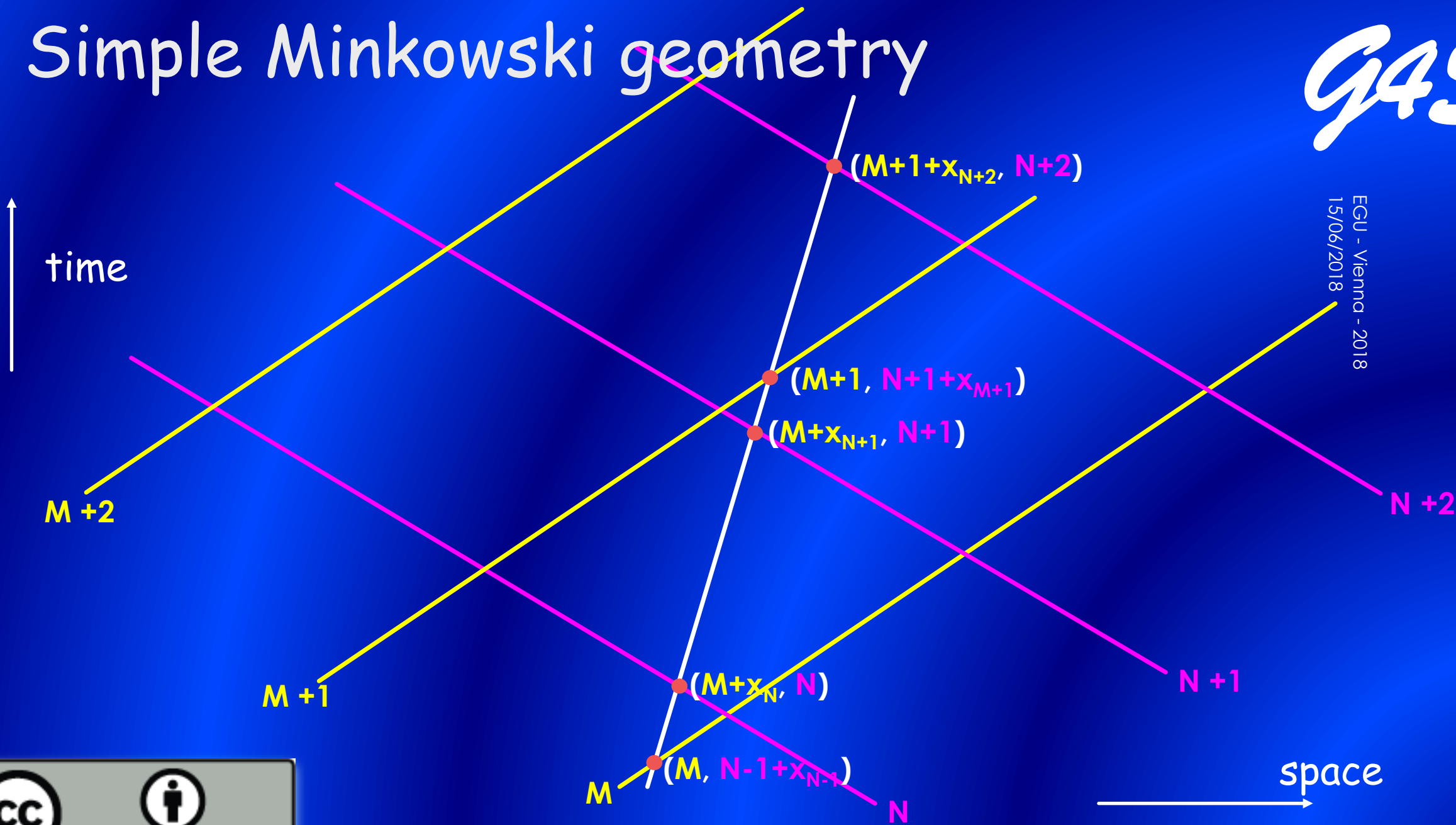
$$r_i = X_{ai} \chi^a \quad X_{ai} = \frac{\tau_{ai}}{T_a} = n_{ai} + x_{ai}$$

$$r = \frac{\tau^a}{T_a} \chi_a + \frac{\tau^b}{T_b} \chi_b + \frac{\tau^c}{T_c} \chi_c + \frac{\tau^d}{T_d} \chi_d$$



Simple Minkowski geometry

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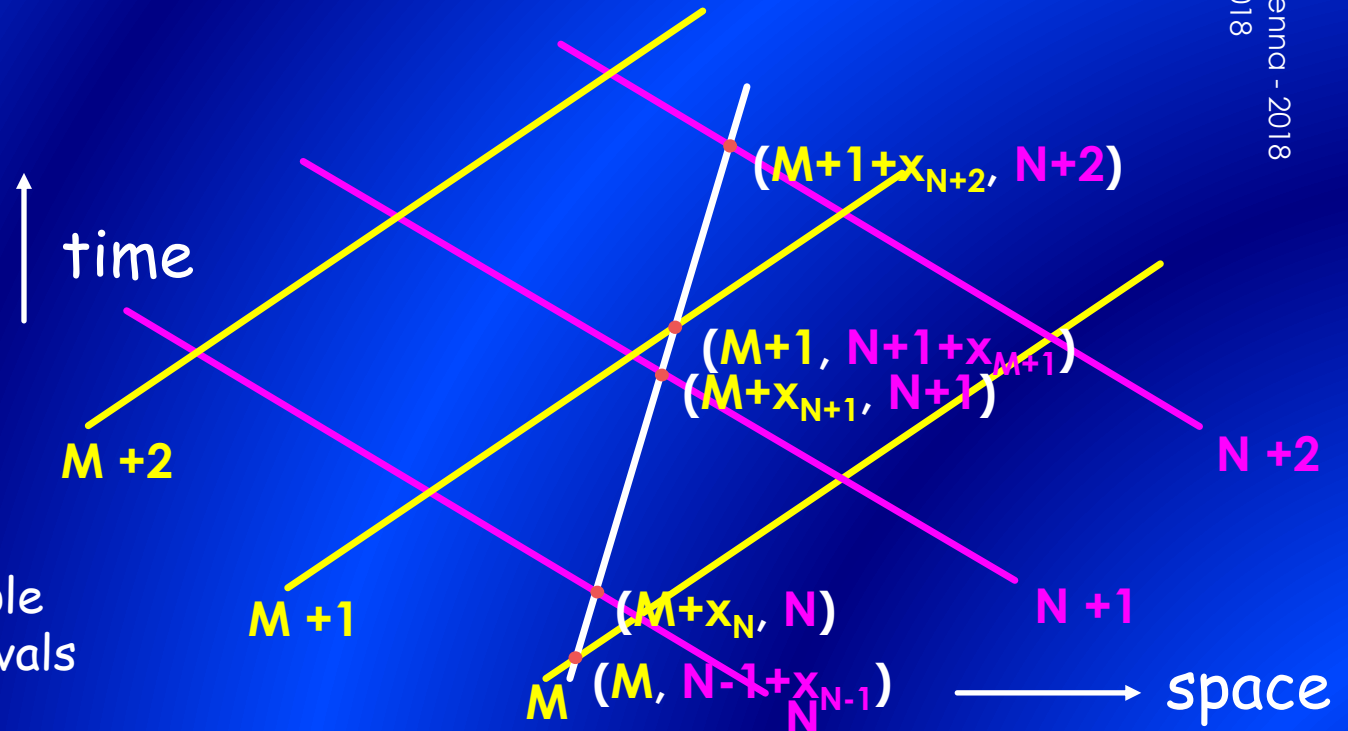
A linear algorithm

i.e. a practical way to evaluate the fractional part x

When the worldline of the user is straight, or may be thought to be approximately straight, the knowledge of the time span between the arrivals of the pulses is sufficient to uniquely localize the reception events, and then the position of the user

Given a sequence of eight arrivals, simple linear relations between the time intervals and the x 's hold

Solving the corresponding system of linear equations leads to the complete definition of the positions in spacetime of the first four reception events



With a moving sequence of events it is then possible to fully reconstruct the whole worldline of the receiver



A linear algorithm

The arrival times of the signals in the sequence will correspond to the following space-time positions:

$$r_1 = (n_1 T)^a \chi_a + [(n_1 + x_1) T]^b \chi_b + [(n_1 + x_1) T]^c \chi_c + [(n_1 + x_1) T]^d \chi_d$$

$$r_2 = [(n_2 + x_2) T]^a \chi_a + (n_2 T)^b \chi_b + [(n_2 + x_2) T]^c \chi_c + [(n_2 + x_2) T]^d \chi_d$$

$$r_3 = [(n_3 + x_3) T]^a \chi_a + [(n_3 + x_3) T]^b \chi_b + (n_3 T)^c \chi_c + [(n_3 + x_3) T]^d \chi_d$$

$$r_4 = [(n_4 + x_4) T]^a \chi_a + [(n_4 + x_4) T]^b \chi_b + [(n_4 + x_4) T]^c \chi_c + (n_4 T)^d \chi_d$$

$$r_5 = [(n_1 + 1) T]^a \chi_a + \dots$$

$$r_6 = \dots$$

$$r_7 = \dots$$

$$r_8 = \dots$$

$$r_{ij} = r_j - r_i = (X_{aj} - X_{ai}) \chi^a = \Delta X_{aij} \chi^a$$

$$\frac{\tau_{ij}}{\tau_{jk}} = \frac{\Delta X_{1ij}}{\Delta X_{1jk}} = \frac{\Delta X_{2ij}}{\Delta X_{2jk}} = \frac{\Delta X_{3ij}}{\Delta X_{3jk}} = \frac{\Delta X_{4ij}}{\Delta X_{4jk}}$$

Let us next consider a sequence of arrival times from respectively sources a,b,c,d and label the corresponding events as 1,2,3,4,5,6,7,8. We choose the first event as coinciding with the crossing of a hypersurface belonging to the a family, so that $x_{a1} = 0$. The second event will be at the next crossing of a b-hypersurface, the third will be at the encounter with the subsequent chypersurface, and so on



At least 8 events (two series of four)

Coordinates in the null frame

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We may easily count the n 's but have no direct means to measure the x 's. However if we suppose that the acceleration of the receiver is small enough to allow for the identification of the world line, in a couple of periods of the sources, with a straight line and if we carry on board the receiver a clock, we are able to measure the proper time intervals between the i -th and j -th arrival events, t_{ij} . With all this, trivial geometric considerations lead to the values of our interest. It is



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Coordinates in the null frame

$$x_{a1} = 0, x_{b1} = 1 - \frac{\tau_{12}}{\tau_{26}}, x_{c1} = 1 - \frac{\tau_{13}}{\tau_{37}}, x_{d1} = 1 - \frac{\tau_{14}}{\tau_{48}}$$

$$x_{a2} = \frac{\tau_{12}}{\tau_{15}}, x_{b2} = 1, x_{c2} = 1 - \frac{\tau_{13}}{\tau_{37}} + \frac{\tau_{12}}{\tau_{37}}, x_{d2} = 1 - \frac{\tau_{14}}{\tau_{48}} + \frac{\tau_{12}}{\tau_{48}}$$

.....

$$\tau_{ij} = \tau_j - \tau_i$$

Difference of the proper arrival times between the j_{th} and the i_{th} event



Uncertainties in the light coordinates

$$\left| \frac{\delta x}{x} \right| \leq 4 \left(\frac{1}{\tau_{i,i+4k}} + \frac{\tau_{i,i+1}}{\tau_{i,i+4k}^2} \right) \delta \tau$$

$\delta \tau$: accuracy of the onboard clock (a few ns)

$$\left| \frac{\delta x}{x} \right| \sim 10^{-9}$$



Conversion from null to practical coordinates

Straightforward when sources at infinity (pulsars)

More elaborate when sources are at finite distance

In any case: very high accuracy when positioning is with respect to a starting point along the space-time trajectory of the receiver



Double aim of the application of the RPS to G4S

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- a) To offer an additional approach to the orbit determination
- b) To test the RPS in inverse configuration, in view of other applications



References

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2. Angelo Tartaglia, Matteo Luca Ruggiero, Emiliano Capolongo, *Advances in Space Research*, 47, 645-653 (2011)
3. Angelo Tartaglia, *Acta Futura*, issue 7, 111-124 (2013)

