

Development of the NIM Mass spectrometer for Exploration of Jupiter's Icy Moons Exospheres

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- Mission Introduction
- NIM Instrument
- Instrument Performance Tests
 - Storage capability of Ion Source
 - Density enhancement behaviour of Antechamber
 - Mass Range
- Tests of different Electronic Boards
 - Low Voltage Board
 - High Voltage Board
 - Motor Board
- Ion Source Redesign and Tests

- JUICE*: Jupiter Icy Moons Explorer
 - Investigation of Jupiter and its environment
 - Characterisation of Jupiter's icy moons Ganymede, Europa and Callisto
- PEP: Particle Environment Package
 - Investigation of the icy moons' atmospheres and plasma environment
 - Determine global surface composition and chemistry



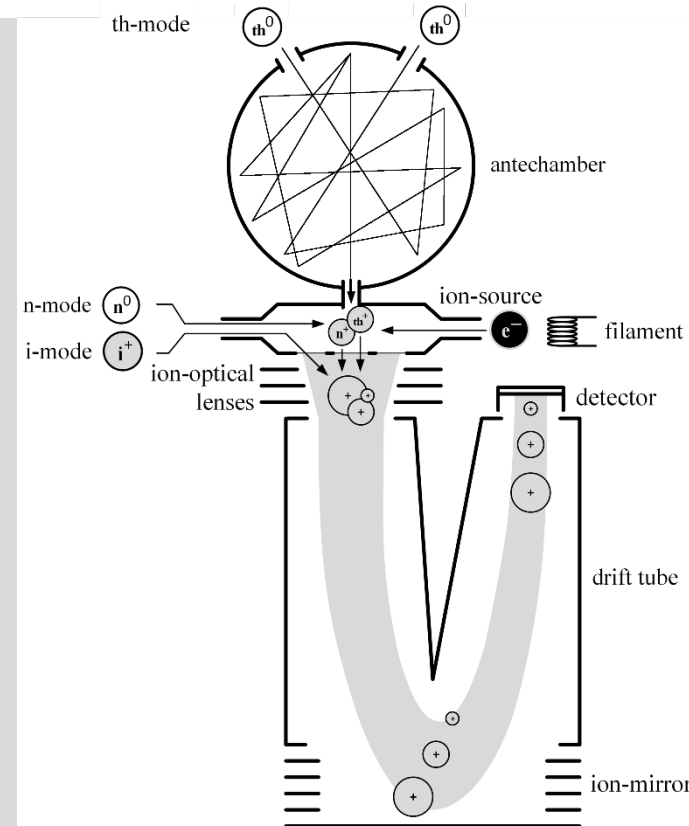
*<http://sci.esa.int/juice>

Neutral Gas and Ion Mass Spectrometer (NIM)

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- 3 modes: **t**hermal-, **n**eutral- and **i**on-mode
 - Closed source: Thermalisation of neutral gas. FoV of $10/3 \pi$ sr
 - Open source: Ions and neutral gas enter ion-source directly. FoV of 300° azimuthal direction and 10° elevation
- Electron ionisation
- Ion mirror to increase the flight distance

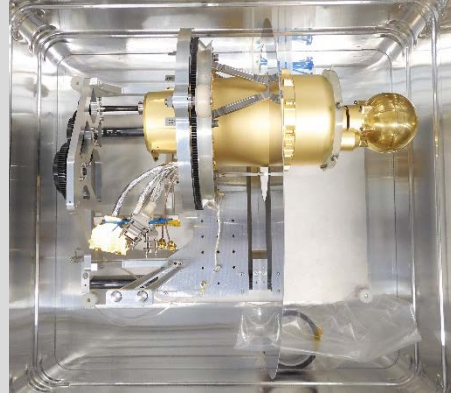


NIM ProtoFlight Model (PFM)

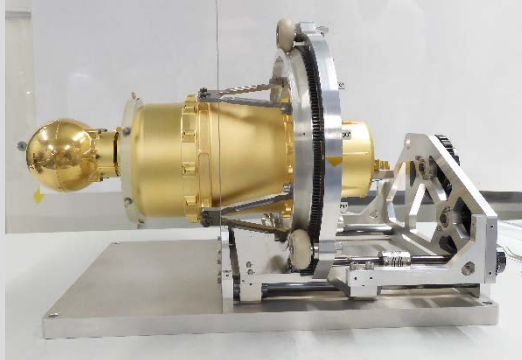


PFM

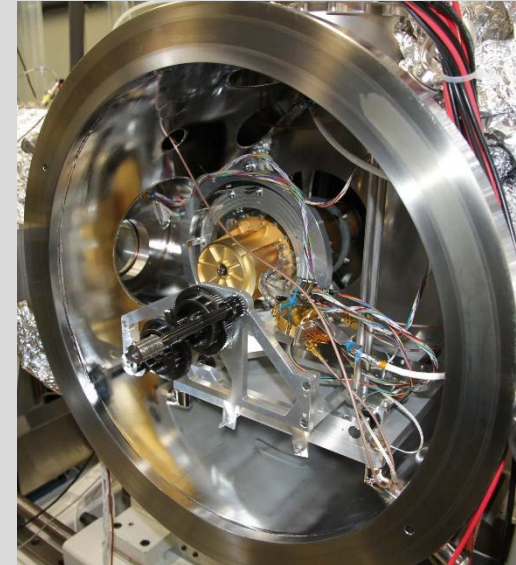
PFM integrated in
the test setup



Transport to the
vacuum chamber



Integration into the
test chamber



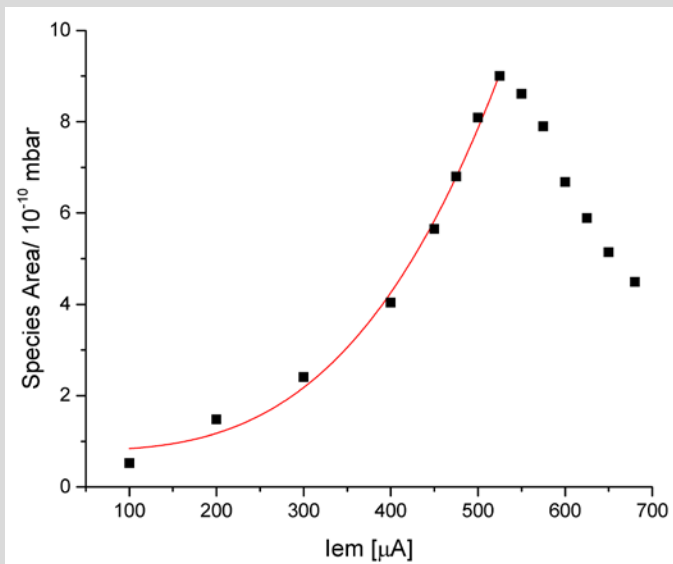
Ion Storage Capability of NIM

Objective

- Investigation of the NIM ion storage capability

Conclusions

- $S \sim I_{em}^{3.2} \rightarrow$ Ion storage capability is very good of this source. Usual storage values, $S \sim I_{em}^2$ (Abplanalp, 2009)
- Signal decrease for $I_{em} > 500 \mu A$ due to space charge effects



Signal height depends on the electron emission current (I_{em}). Red curved is a fit with the function : $S = a + b \cdot I_{em}^c$

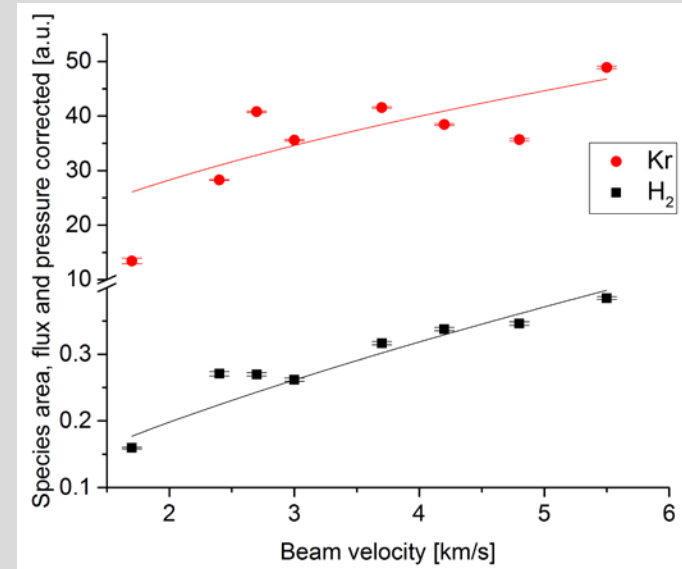
Density enhancement behaviour of Antechamber

Objective

- Verification of the functionality of the antechamber by measuring the density enhancement behaviour of the antechamber

Conclusion

- Signal increase with increasing beam velocity as expected according to (Wurz et al., 2007)



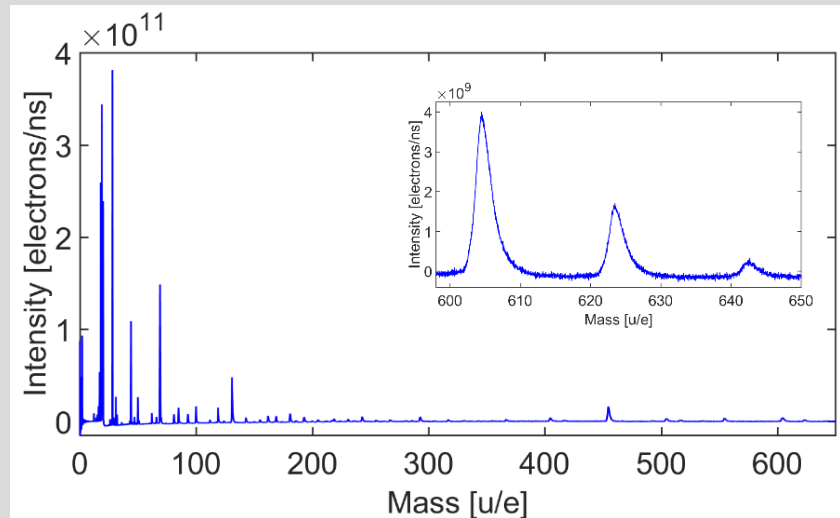
Mass Range

Objective

- NIM is designed to measure masses up to 1'000 u. Expected masses up to ~100 u
- FC5311 was used as a test sample because it has tabulated masses up to 624 u. This was the sample with the highest mass range available.

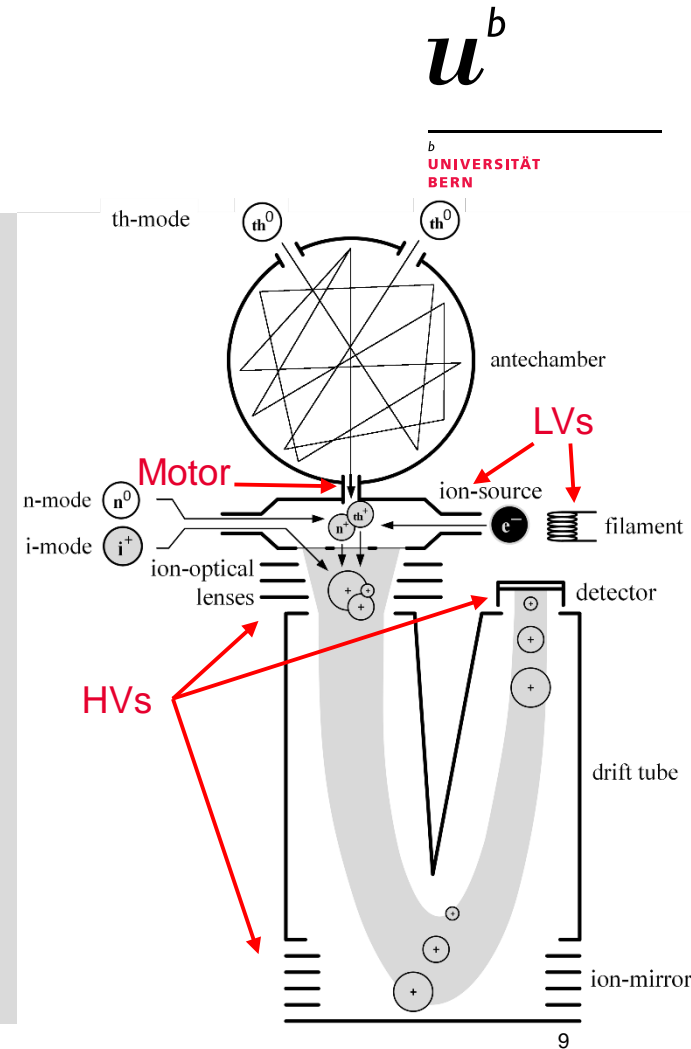
Conclusion

- Masses visible up to 642 u which is even higher than the highest mass tabulated (624 u) of this test sample



Electronic Board Tests

- Low Voltages (LVs)
 - Used in the ion-source and around the electron emitting filament
- High Voltages (HVs)
 - Used for the focusing lenses in the ion-source, for the ion-mirror and the detector
- Shutter Motor
 - Located between the antechamber and the ion-source. Used when measuring in n-Mode to block particles coming from the antechamber



Low Voltage Board

Electronic Board Tests

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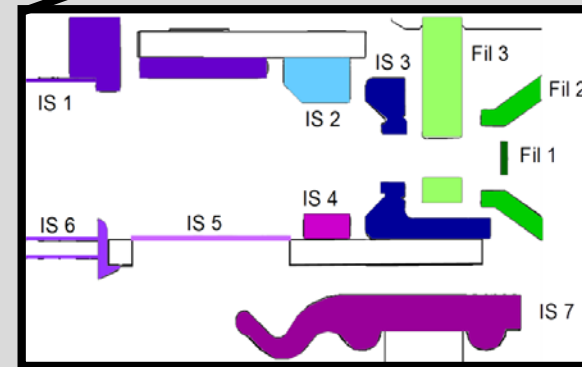
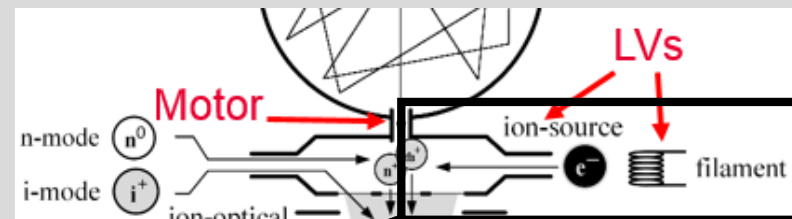
Measurement configurations

- Lab electronics: all voltages were provided by lab electronics. In case separate electrodes are listed, the voltages of these electrodes are provided by the flight low voltage board

Configuration	m/ Δm				SNR			
	O ₁₆	F ₁₉	N ₂	CO ₂	O ₁₆	F ₁₉	N ₂	CO ₂
Fil1	90 ±3	174 ±10	419 ±47	471 ±47	35	498	243	116
Lab electronics	103 ±4	156 ±8	419 ±47	448 ±43	62	838	544	263
Fil 1, 2, 3, IS 1, 2, 4	93 ±3	174 ±10	419 ±47	448 ±43	36	471	182	94
Lab electronics	105 ±4	164 ±9	419 ±47	471 ±47	74	789	654	325
IS 1, 2, 4	107 ±4	164 ±9	419 ±47	471 ±47	75	771	679	339

Conclusion

- Same mass resolution for all different configurations
- LV board not calibrated → lower SNR



Ion-Source (IS) and Filament (Fil) electrodes

High Voltage Board

Electronic Board Tests

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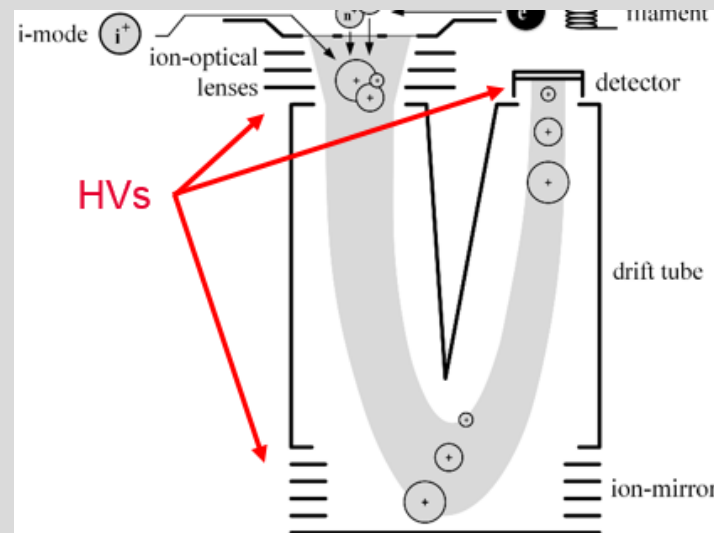
Measurement configurations

- Lab electronics: all voltages were provided by lab electronics. In case separate electrodes are listed, the voltages of these electrodes are provided by the flight high voltage board. Tested electrodes were electrodes from the ion-mirror
- R2 reaches -7 kV, R15 is a bipolar high voltage electrode reaching up to +2.3 kV

Configuration	m/ Δ m				SNR			
	N ₂	CO ₂	¹²⁹ Xe	¹³² Xe	N ₂	CO ₂	¹²⁹ Xe	¹³² Xe
R15	307 ± 25	330 ± 23	361 ± 16	365 ± 16	422	266	383	385
Lab electronics	295 ± 23	330 ± 23	396 ± 17	365 ± 16	350	225	310	313
R2	295 ± 23	330 ± 23	361 ± 16	373 ± 17	250	159	227	229
R2, R15	295 ± 23	330 ± 23	361 ± 16	373 ± 17	293	189	268	269
Lab electronics	295 ± 23	319 ± 21	369 ± 17	373 ± 17	316	204	287	289

Conclusion

- Same mass resolution and signal-to-noise ratio for the different configurations



Motor Boards

Electronic Board Tests

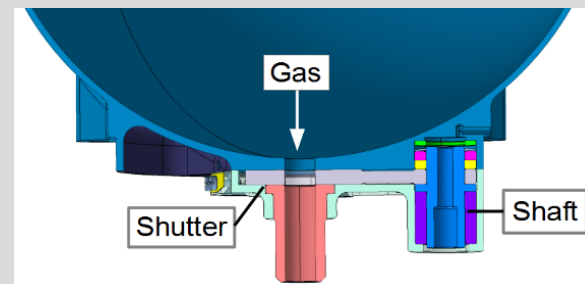
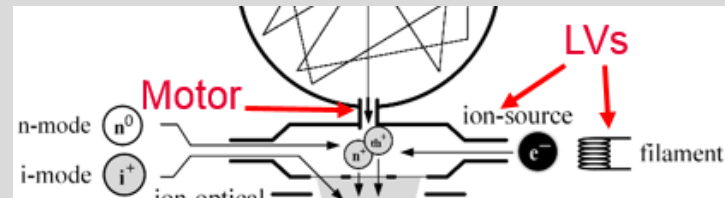
Measurement configurations

- Shutter open/ closed: neutral particle beam enters the ion source through the antechamber;
- Background: neutral particle beam points on the antechamber outer wall and scatters into the ion-source
- N_2 and CO_2 are residual gas and not part of the beam

Configuration	m/ Δ m				SNR			
	N_2	CO_2	^{129}Xe	^{132}Xe	N_2	CO_2	^{129}Xe	^{132}Xe
Shutter open	307 \pm 25	330 \pm 23	361 \pm 16	373 \pm 17	550.8	309.9	735.7	740.2
Shutter close	307 \pm 25	330 \pm 23	361 \pm 16	365 \pm 16	540.3	315.4	120.8	121.6
background	307 \pm 25	330 \pm 23	353 \pm 15	357 \pm 16	631.6	360.2	85.4	84

Conclusion

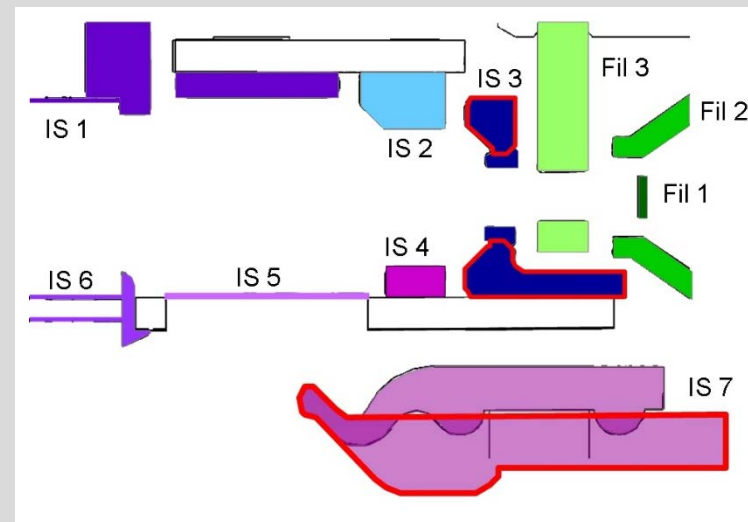
- Same mass resolution for all different configurations
- Shutter performance lower than expected due to differences between laboratory and flight environment



Redesign of IS

Objective

- Redesign of ion-source due to mechanical failure during vibration test
- New design:
 - ring to narrow the entrance hole of the electron beam had to be removed due to mechanical reasons
 - IS 7 was shifted to increase its distance to the entrance (IS 3) to increase high voltage stability
 - In red the redesigned electrodes IS 3 and IS 7



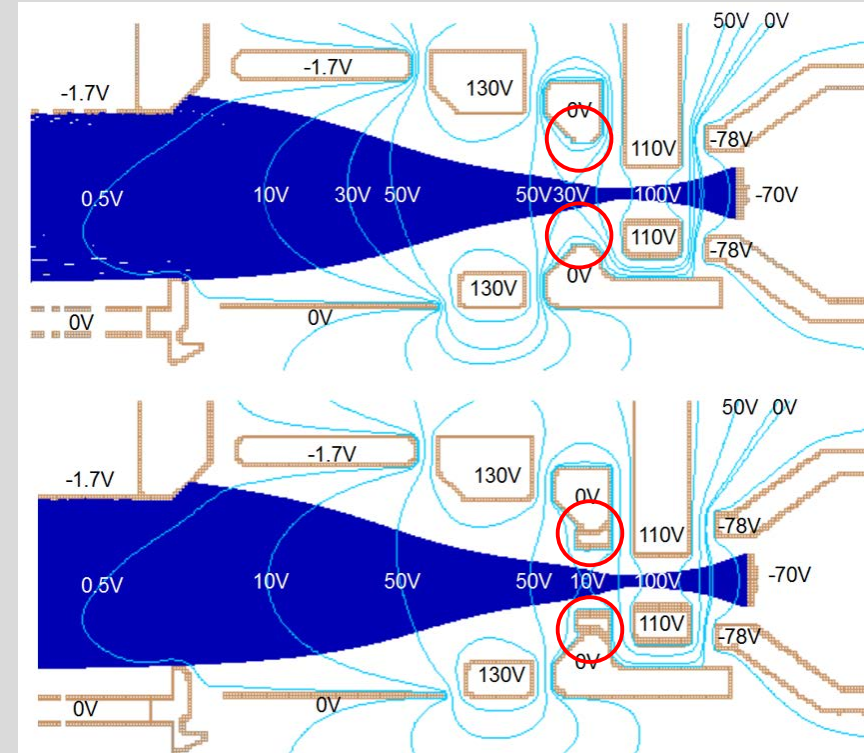
Redesign of Ion Source

Objective

- Simulations to evaluate the impact of the redesign on the performance of the ion source

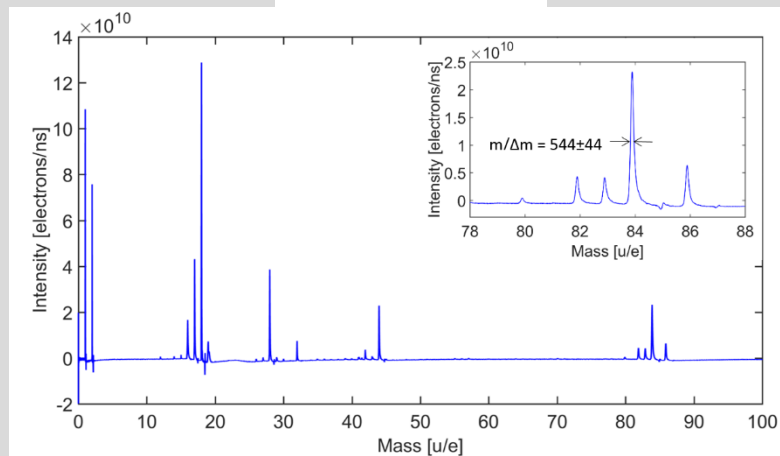
Conclusion

- The redesign did not change the flight path of the electrons significantly, therefore expected performance of the ion source should be the same

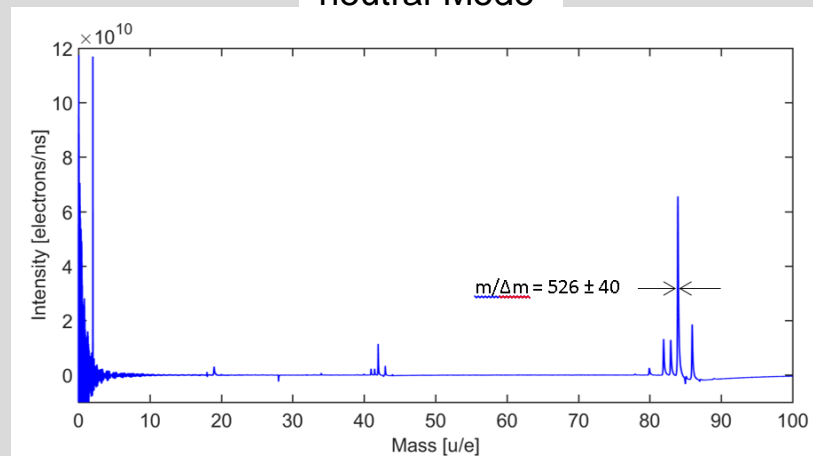


Test Results

thermal Mode



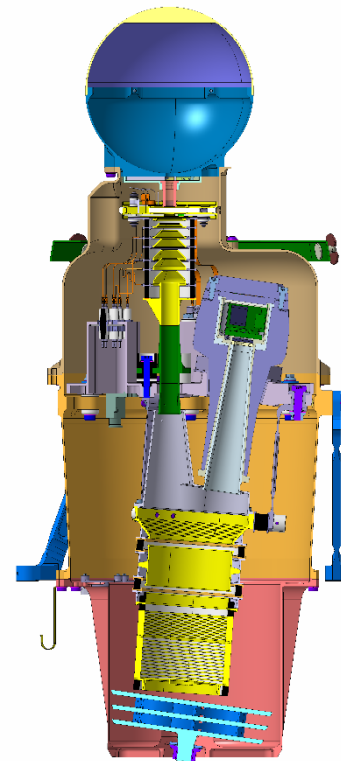
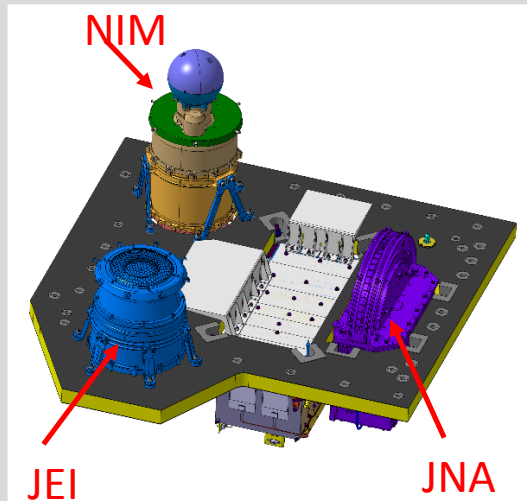
neutral Mode



- Mass resolutions could be improved by a redesign of the old source

Next Steps

- Finalise work with PFM
- Commissioning, testing and calibration of the Proto Flight Model (PFM) sensor with flight electronics
- Integration and final testing completed by end of July 2020



Have a nice day

