

Development of a small unmanned aircraft system to derive CO₂ emissions of anthropogenic point sources

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Atmospheric Measurement Techniques

Introduction

- CO₂ emissions are the primary cause of man-made climate change.
- Large parts of the human CO₂ emissions come from point sources such as power plants.
- On a global level, a monitoring by means of satellites is envisaged (e.g., CO₂M).
- Regionally, this is going to be complemented and validated with airborne measurements.
- UAV (unmanned aerial vehicle) based measurements can provide a cost-effective way to contribute to these activities.
- **For this purpose, a sUAS (small unmanned aircraft system) has been developed to quantify the CO₂ emissions of a nearby point source from its downwind mass flux without the need for any ancillary data.**

Outline

- Hardware setup and instrumentation
- CO₂ sensor characterization
- Anemometer calibration
- In-flight validation using ICOS measurements
- Demonstration flights downwind of an industrial facility

UAV and instrumentation

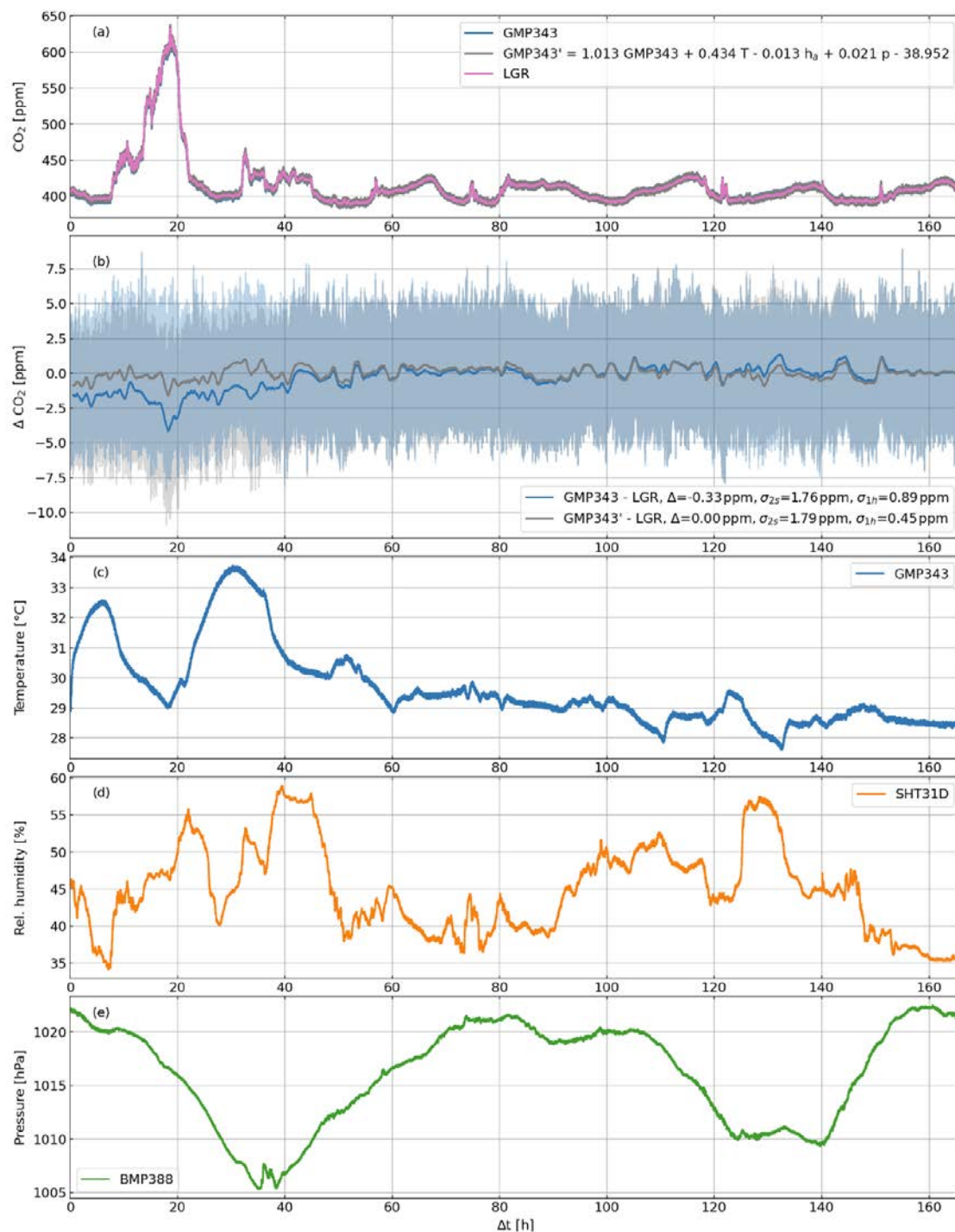


Device	Weight [g]	Power [W]	Flight time [min]
UAV DJI M210v2	4800	836.1 [#]	34.0
Camera DJI X4S	253	-	-1.9 ^b
CO ₂ Vaisala GMP343	360	2.4	-2.7
Wind FTTech. FT205	100	0.6	-0.8
Raspberri Pi [♫]	143	2.8	-1.1
Telemetry Dronee Zoon	36	0.2	-0.3
Wiring, converters, etc.	143	-	-1.1
Plattform and pole mount	187	-	-1.4
Total	6022	842.1	24.7



- Usage of off the shelf components. UAV: DJI Matrice 210v2. CO2: Vaisala GMP343. Wind: FT Technologies FT205. Pressure: Bosch BMP388. Temperature and humidity: Sensirion SHT31-DIS.
- Total weight: 4.8kg UAV + 1.2kg Payload (including camera)
- Maximum flight time: 24.7min.

Characterization of the CO2 Instrument



- We compared measurements of the Vaisala GMP343 CO₂ sensor with those of a highly accurate (± 0.3 ppm) ABB LGR-ICOS ultra-portable greenhouse gas analyzers
- The comparison period was about one week
- Ambient pressure, temperature, and humidity varied larger than expected for the measurement flights
- The single sounding precision is about 1.8 ppm (1σ)
- Drifts are in the order of 0.9 ppm (1σ)
- A linear correction using temperature, pressure, absolute humidity, and the measured CO₂ concentration can reduce drifts to 0.4 ppm
- The linear correction mainly corrects for an underestimation of the CO₂ variability of 1.3%



Calibration of the anemometer

- Mounting the anemometer aboard an UAV makes a **calibration necessary**
- Straight forward approach: wind tunnel
 - Hovering in place not easy
 - Large wind tunnel not easily available
- **Our approach: use head wind for calibration**

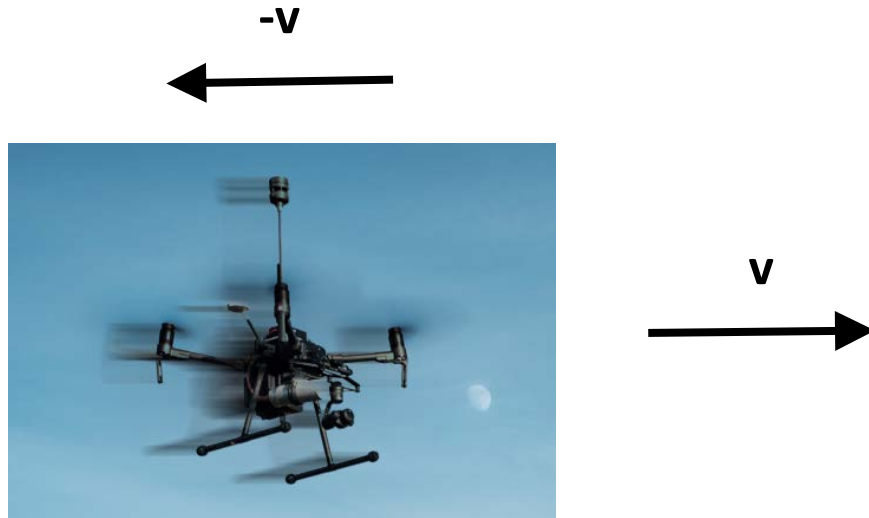


Without ambient wind ($w=0$)

Calibration function: $u' = A m'$

u' = calibrated measurement
 m' = raw anemometer measurement
 A = calibration coefficient

Calibration of the anemometer



Calibration

$$u' = -v = A m'$$

$$A = -v/m$$

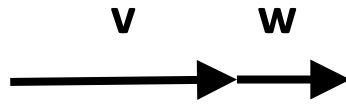


With ambient wind ($w \neq 0$)

Calibration function: $u' = A m'$

u' = calibrated measurement
 m' = raw anemometer measurement
 A = calibration coefficient

Calibration of the anemometer



Calibration

$$w - v = A m'$$

UNDERDETERMINED

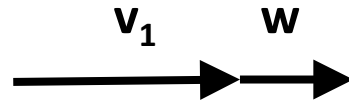
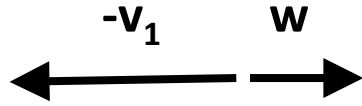


With ambient wind ($w \neq 0$)

Calibration function: $u' = A m'$

u' = calibrated measurement
 m' = raw anemometer measurement
 A = calibration coefficient

Calibration of the anemometer

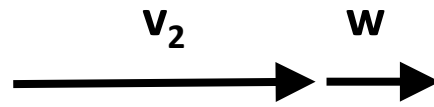
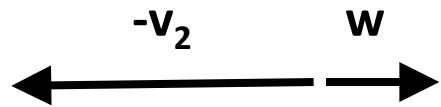


Calibration

$$w - v_1 = A m_1'$$

$$w - v_2 = A m_2'$$

$$A = (v_2 - v_1) / (m_1' - m_2')$$





With ambient wind ($w \neq 0$)

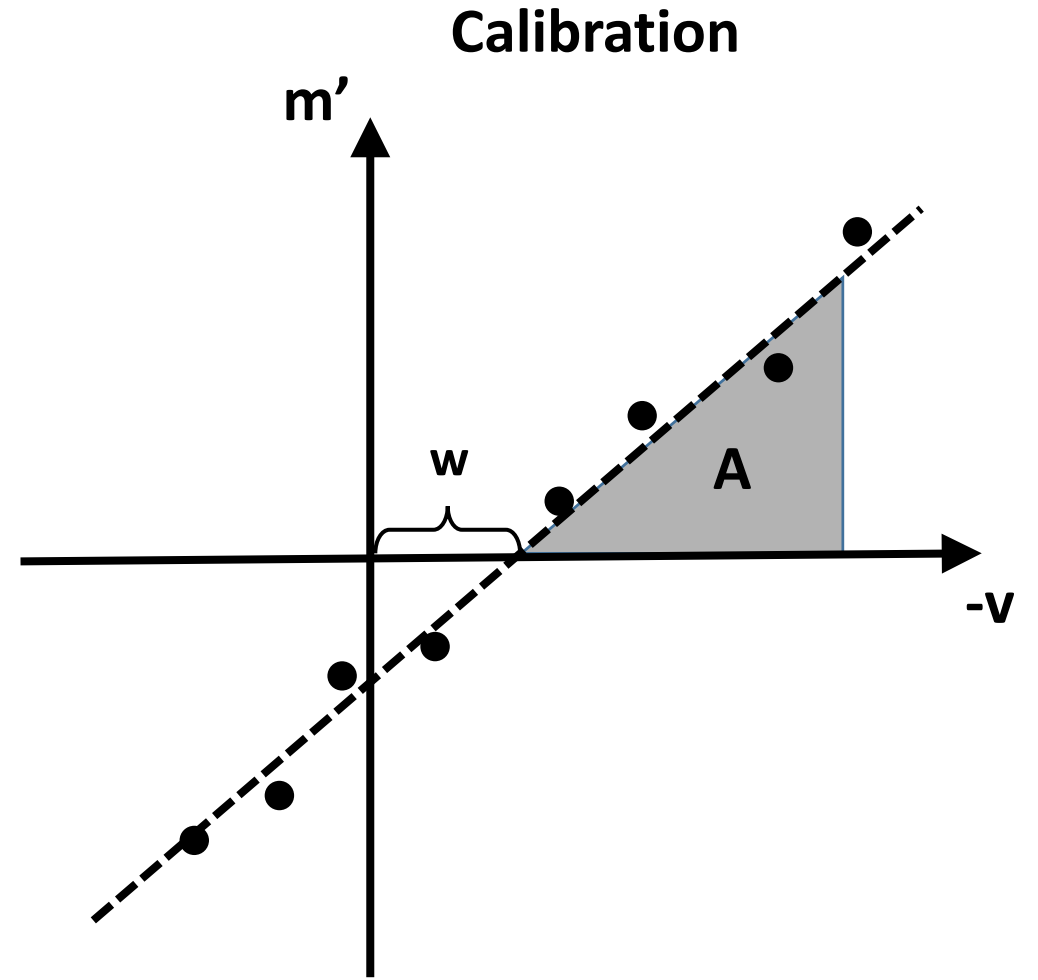
Calibration function: $u' = A m'$

u' = calibrated measurement
 m' = raw anemometer measurement
 A = calibration coefficient

Various velocities, forward+backward



Calibration of the anemometer



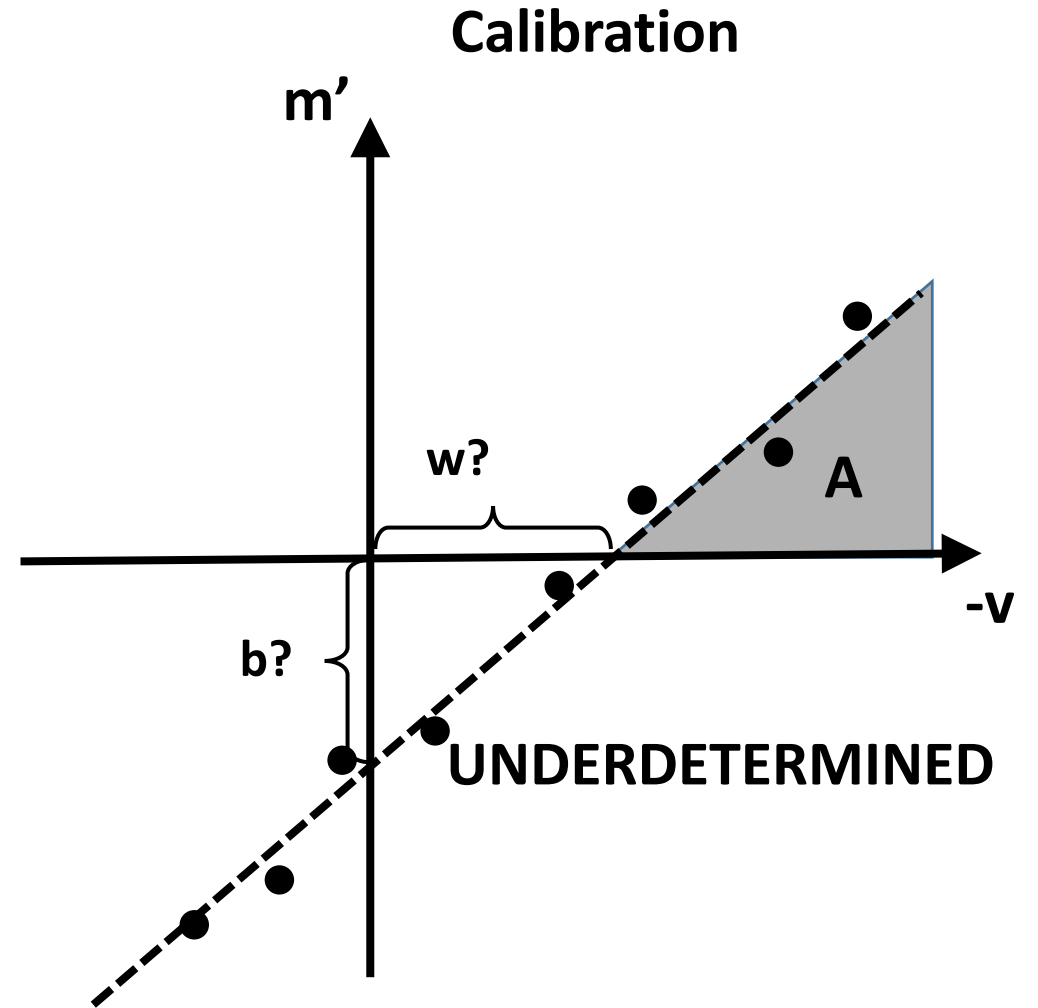
FT205
Wind sensor

With ambient wind ($w \neq 0$)
Calibration function: $u' = A m' + b$

u' = calibrated measurement
 m' = raw anemometer measurement
 A = calibration coefficient
 b = constant bias

Calibration of the anemometer

Various velocities, forward+backward





With ambient wind ($w \neq 0$)

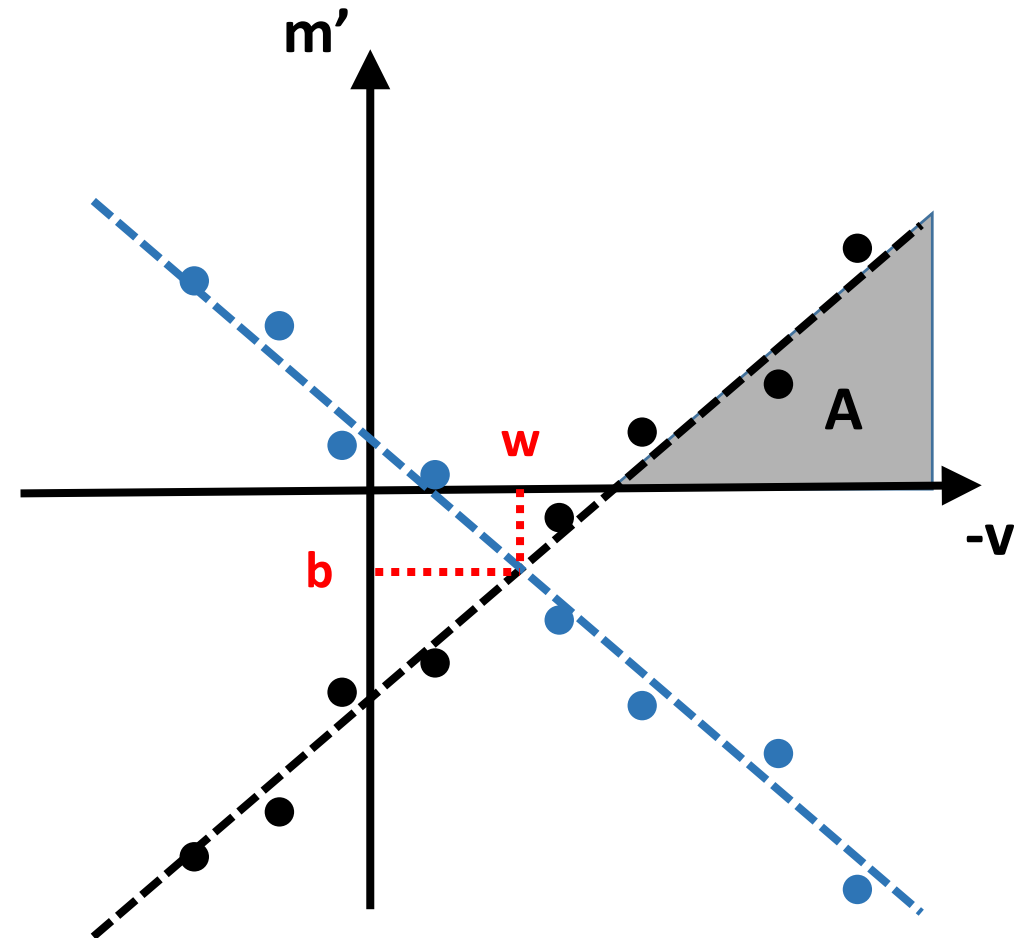
Calibration function: $u' = A m' + b$

Various velocities, forward+backward,
different orientation



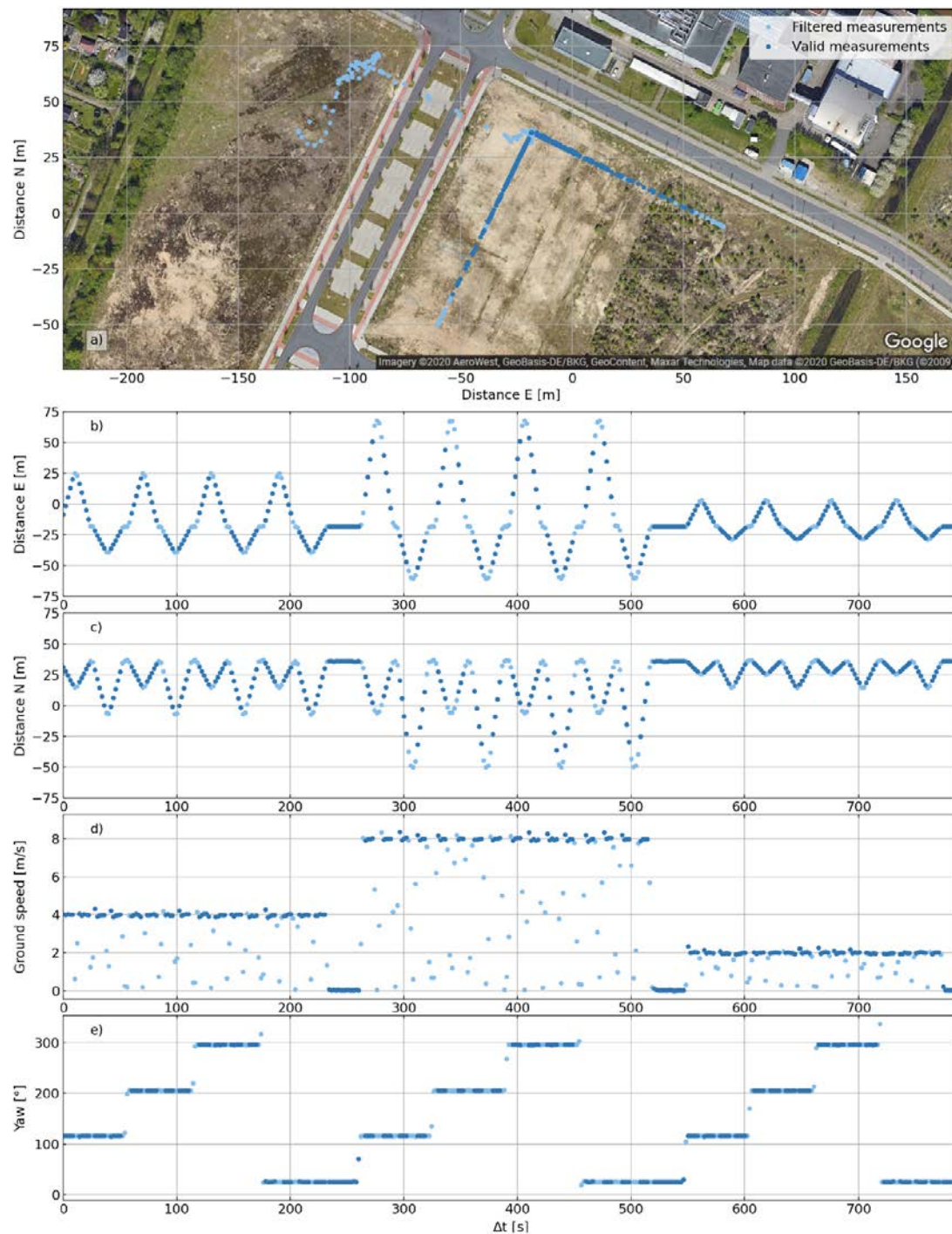
u' = calibrated measurement
 m' = raw anemometer measurement
 A = calibration coefficient
 b = constant bias

Calibration



Calibration of the anemometer

Calibration of the anemometer: Flight pattern



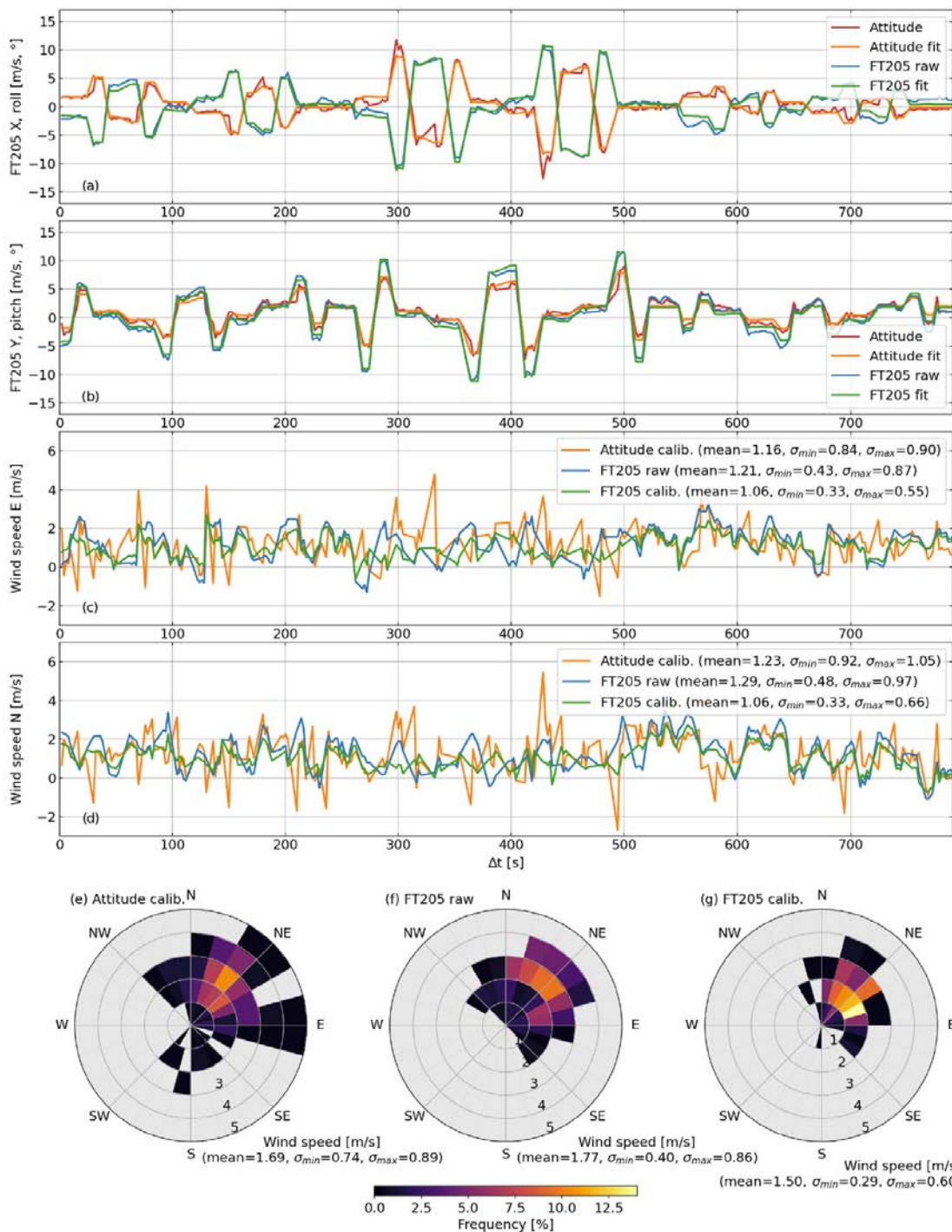
- In **2D**, the calibration function becomes a **vector function**:
$$u' = \mathbf{A} m' + b$$

(m' = raw anemometer measurement in UAV coordinates, \mathbf{A} = calibration matrix accounting for scaling and rotation, b = offset)
- Solving for m' :
$$m' = \mathbf{A}^{-1} [u' - b]$$
- Substitute u' :
$$m' = \mathbf{A}^{-1} [\mathbf{R} u - b]$$

(\mathbf{R} = rotation matrix, u = wind at UAV in geo.c.)
- Substitute u :
$$m' = \mathbf{A}^{-1} [\mathbf{R} (w - v) - b]$$

(w = wind, v = velocity of UAV)
- Find coefficients of \mathbf{A} , b , and w by least squares fitting and assuming stationary winds
- This requires a flight pattern with two ideally perpendicular paths and varying ground speeds, backward/forward motions, and orientations of the UAV.

Calibration of the anemometer: Results



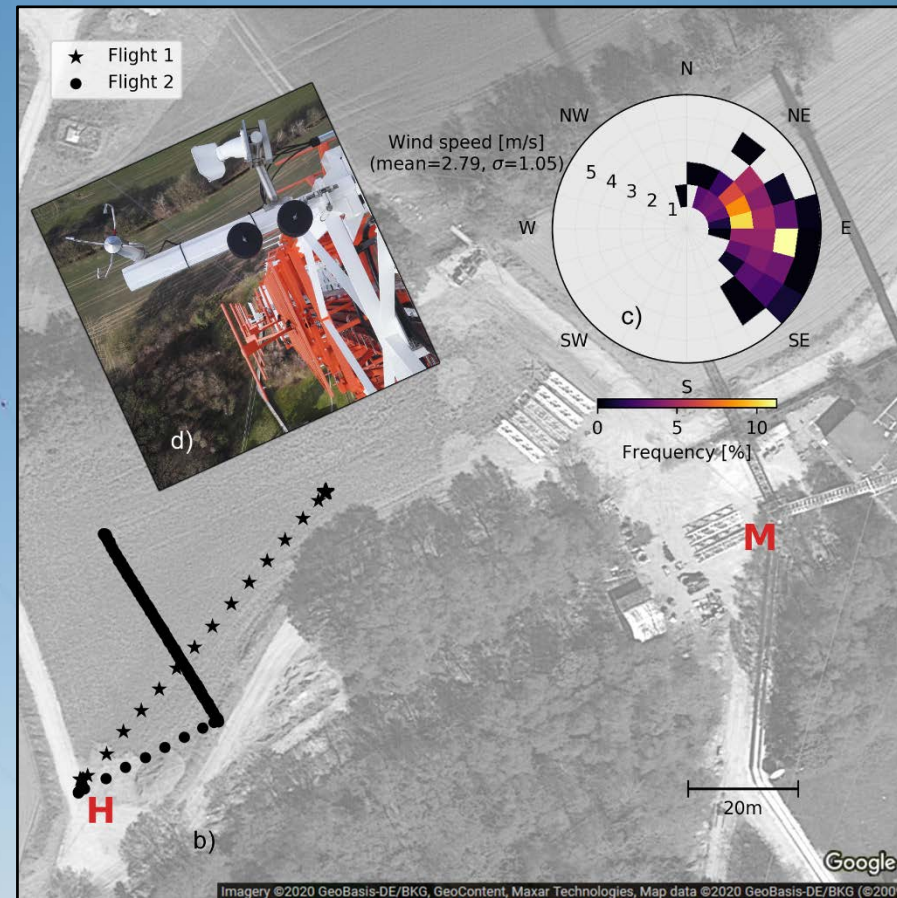
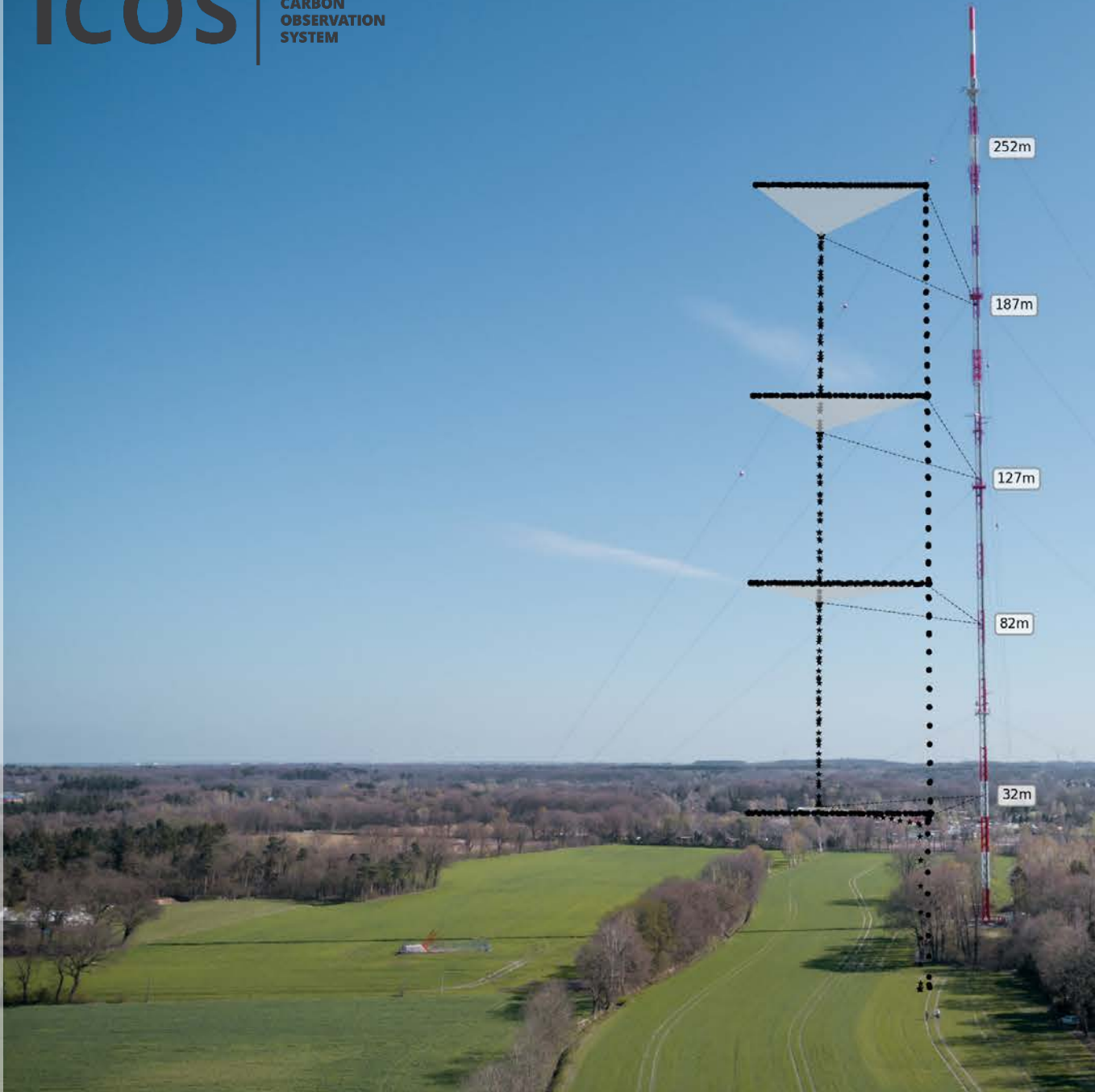
- $A = \begin{pmatrix} 0.88 & 0.02 \\ -0.02 & 0.83 \end{pmatrix}$, $b = \begin{pmatrix} 0.07 \\ -0.06 \end{pmatrix} m/s$, $w = \begin{pmatrix} 1.06 \\ 1.06 \end{pmatrix} m/s$
- Correction of an overestimation by 10-20%.
- Basically no rotation.
- Little constant offset below 0.1m/s.
- Average wind speed of 1.5m/s during the calibration flight.
- After calibration, the expected wind at the sensor agrees well with the measured wind.
- The calibrated wind relative to ground scatters less around the average compared to the un-calibrated wind.
- The calibration narrows the wind histogram.
- After calibration, we expect the precision of the wind speed relative to ground to be in the range of 0.3-0.6m/s.
- In principle, UAV attitude can also be used for wind measurement, but with larger noise (0.7-0.9m/s) despite stricter filter requirements.



Validation with ICOS in Steinkimmen:
Overview

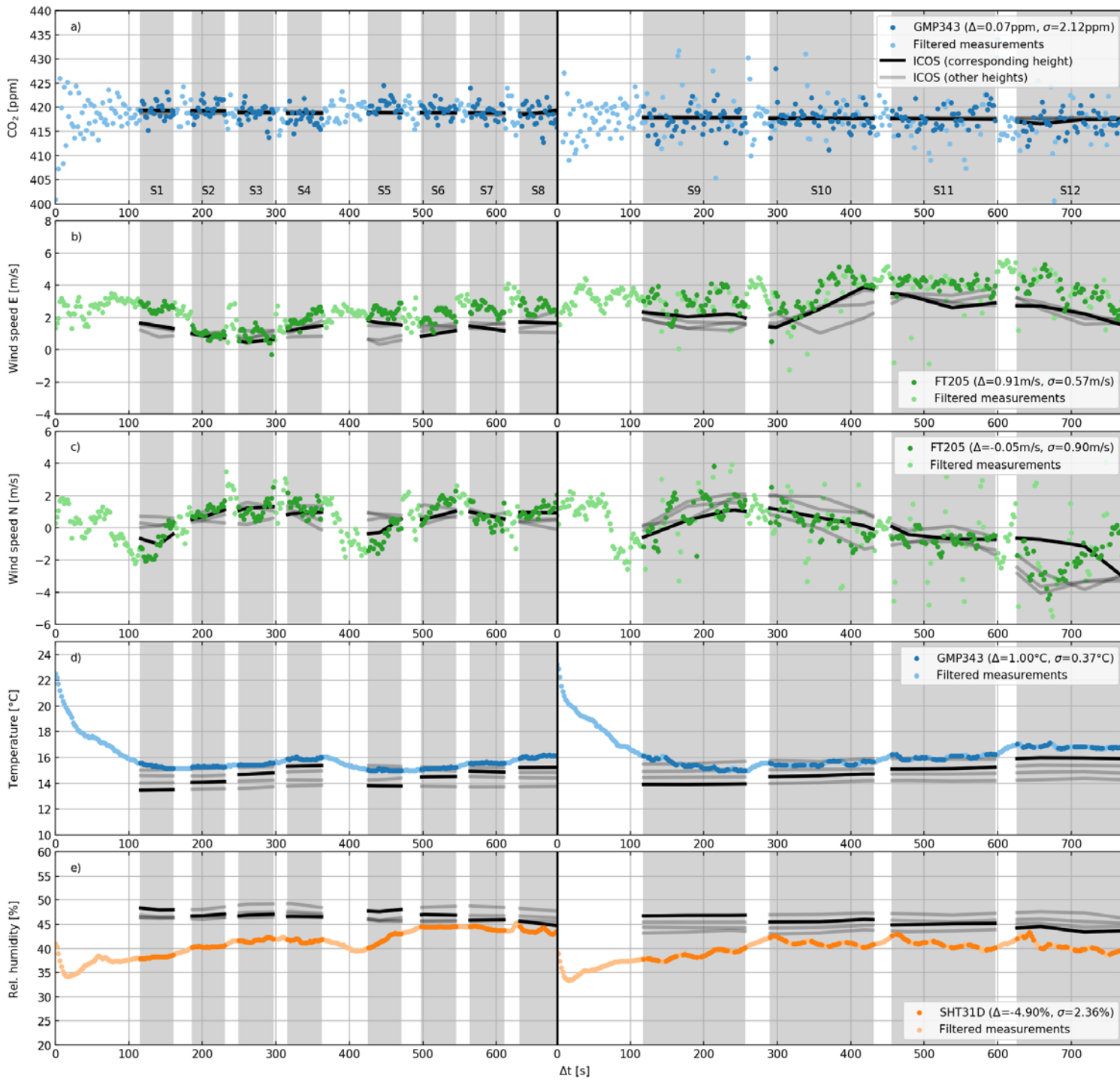
ICOS

INTEGRATED
CARBON
OBSERVATION
SYSTEM



- ★ Flight 1 (take-off: 09.04.2020 09:06:23 UTC)
- Flight 2 (take-off: 09.04.2020 09:30:47 UTC)

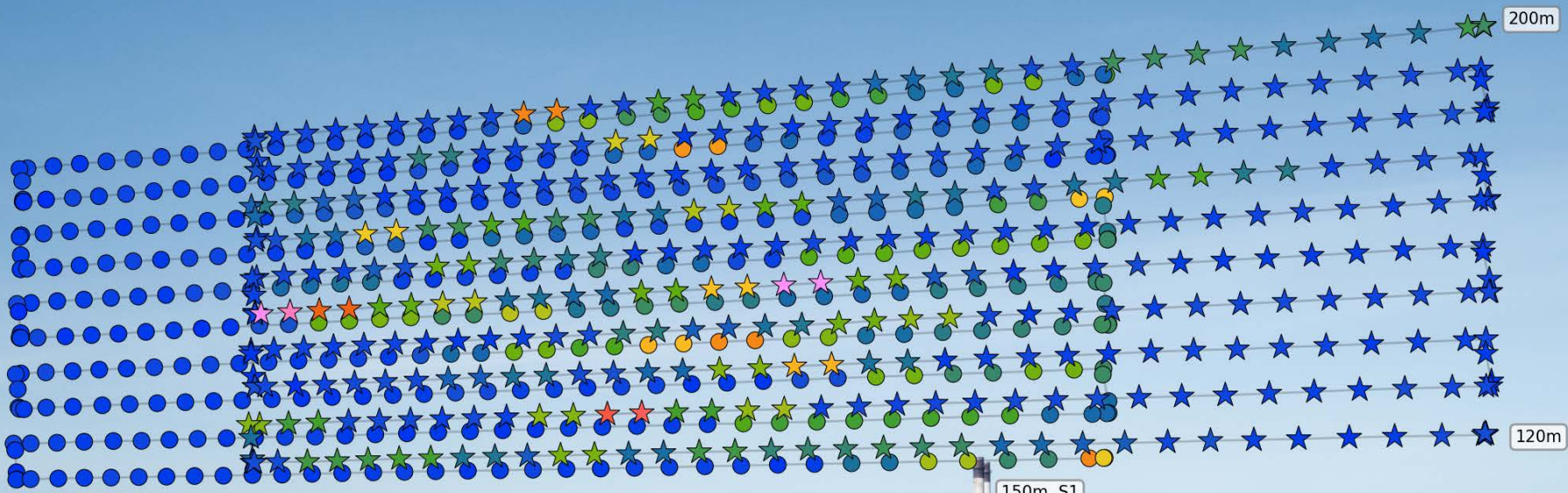
Validation with ICOS in Steinkimmen: Results



- CO_2 : Bias 0.1ppm, scatter 2.1ppm, no flight-to-flight differences, somewhat larger scatter when UAV moves
- Wind north component: Bias 0.10m/s, scatter 0.90m/s (0.67m/s without last sequence at 32m)
- Wind east component: Bias 0.91m/s, scatter 0.57m/s. Bias likely due to ICOS (wind blockage). UAV operated with different orientations but bias of east component persisted. ICOS instrument mounted in wind direction.

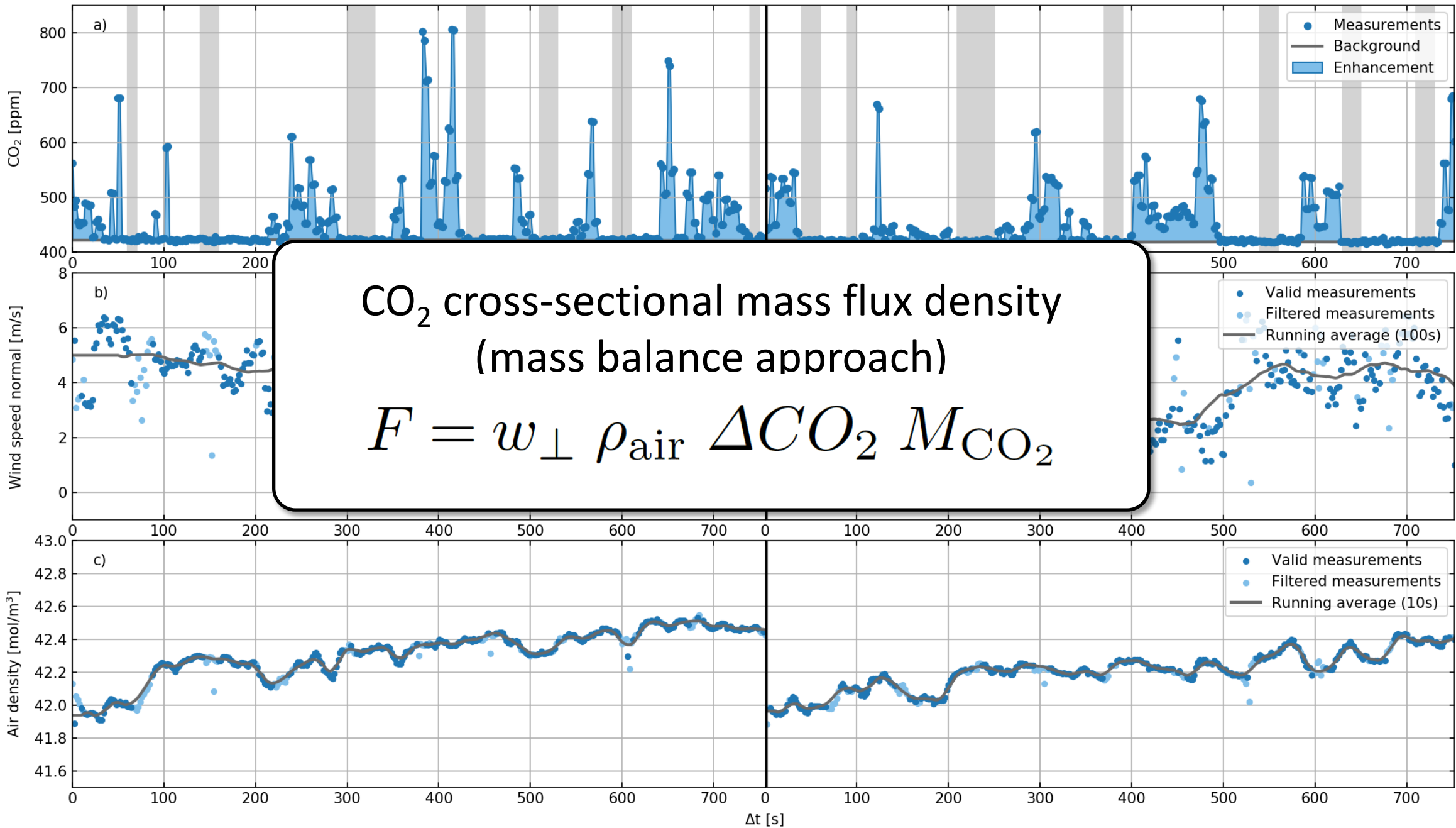


ExxonMobil Natural gas processing facility in Großenkneten: Overview

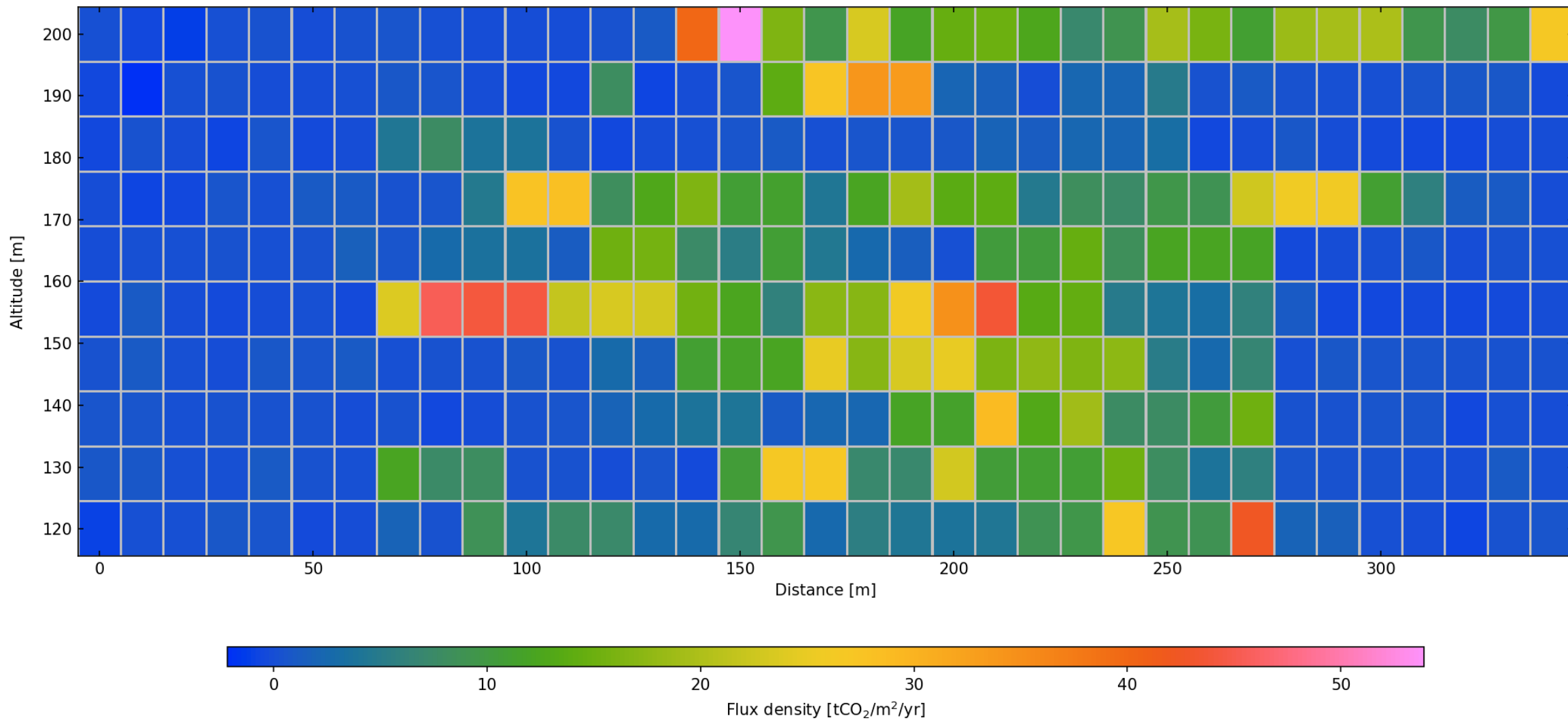


- ☆ Flight 1 (take-off: 10.04.2020 12:06:03 UTC)
- Flight 2 (take-off: 10.04.2020 12:30:21 UTC)

ExxonMobil Natural gas processing facility
in Großenkneten: **Measurements**

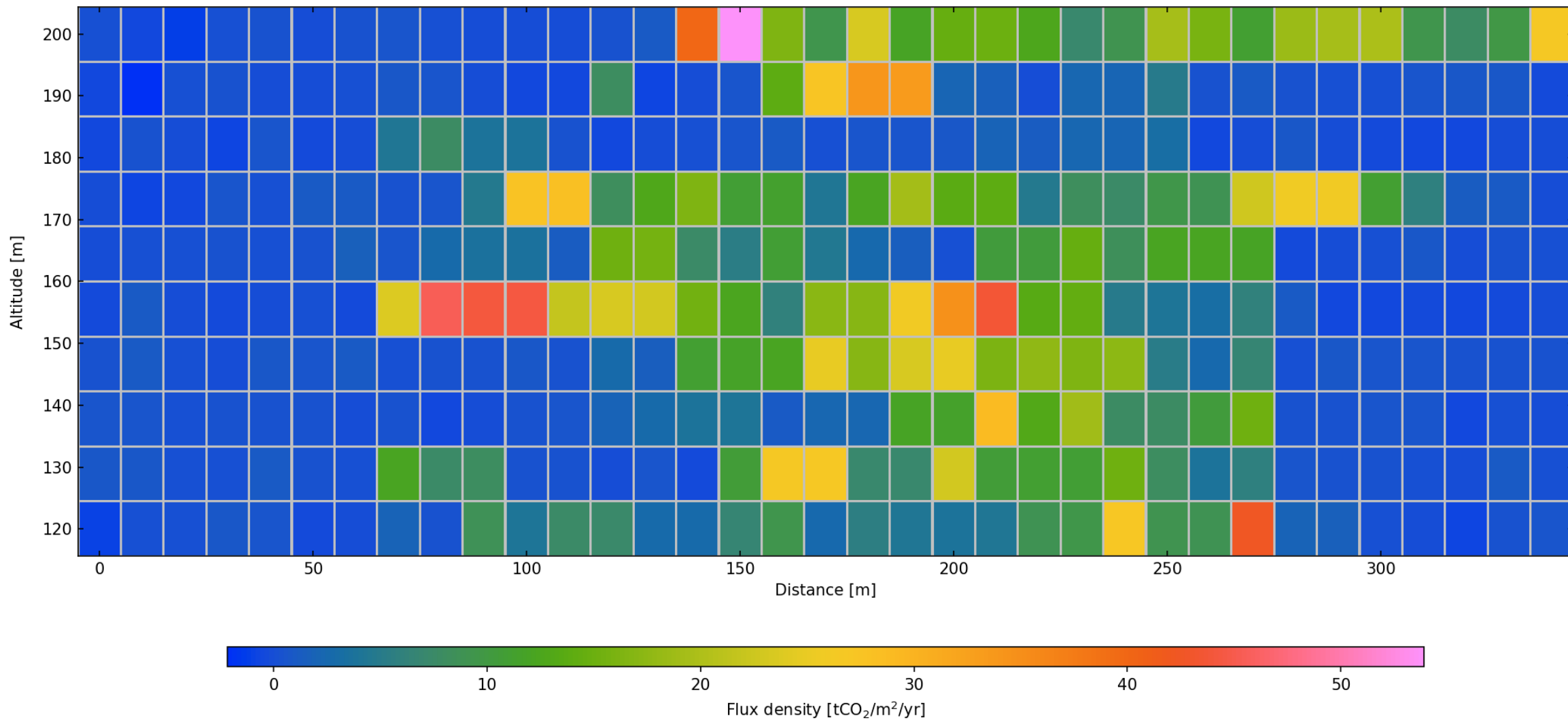


ExxonMobil Natural gas processing facility
in Großenkneten: Cross-sectional flux



- Good horizontal position of the flight pattern.
- Very scattered plume. The Gaussian plume approximation applies only on average.
- Integral of flux density results in a total flux of 196 ± 17 ktCO₂/year.
- Reported annual flux is about 3-4 times larger.
- Discrepancy likely due to plume above the flight pattern which ends at 200m because of flight regulations.

ExxonMobil Natural gas processing facility in Großenkneten: Uncertainties



- **Total flux uncertainty** (only considering instrumental effects): **8.7%**
(systematic uncertainties: 8.3% wind, 1.3% CO₂, 3.5% temperature; stochastic uncertainties 1.3%)
- **Turbulence** and under sampling of the plume can introduce **larger uncertainties**
- Such uncertainties can average out with increasing number of flights or increasing flight time

- **A sUAS has been developed** which is capable to determine atmospheric CO₂ mass fluxes from its own sensor data independently from external data sources.
- The CO₂ sensor has been characterized by comparisons with a lab instrument.
- The wind measurements have been calibrated using the head-wind.
- The calibration and validation measurements show that **all sensors perform well** also under flight conditions.
- Two demonstration flights suggest that the **plume cross-sectional flux can be determined with an error below 10%** (considering only instrumental effect).
- The wind measurement dominates this error budget.