

Potential and limits of small satellite networks for temporal gravity field retrieval in the frame of the CubeGrav Project

May 26, 2022

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EGU 2022, Wien









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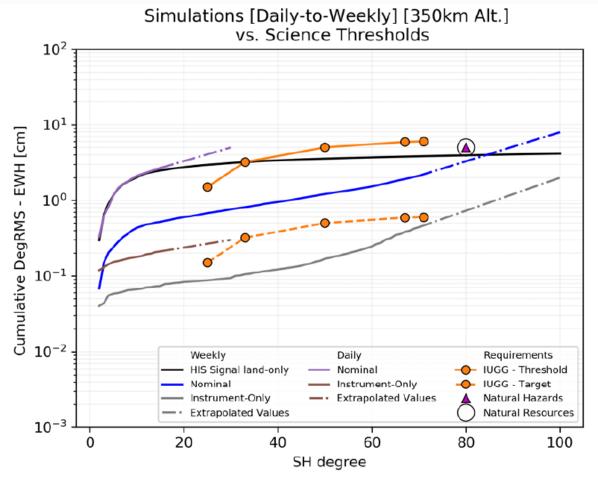
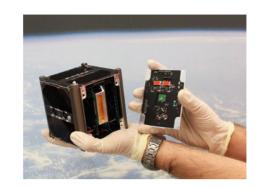


Figure 7. Cumulative errors of gravity field simulation vs. user and mission requirements expressed in EWH for daily-to-weekly time scales. The markers indicate specific target requirements.

Mission Requirement Document, MAGIC



- □ Science and User Needs on Gravity Field products not fulfilled with up-todate gravity missions → Next Generation Gravity Mission (NGGM)
- Project study "CubeGrav" as mission concepts for NGGM with small satellites including potential analysis on
 - increase of spatio-temporal resolution
 - reduction of temporal aliasing
 - acceptable cost effort for dedicated mission
- → What is the impact of different satellite formations and constellations on gravity field solutions?
- → Focus on spatial coverage and temporal resolvability



Simulation Environment



- Full Scale Numerical Closed Loop Simulations
- Gravity Field processing pair-wise
- Two main gravity-relevant instruments considered:

Accelerometer:

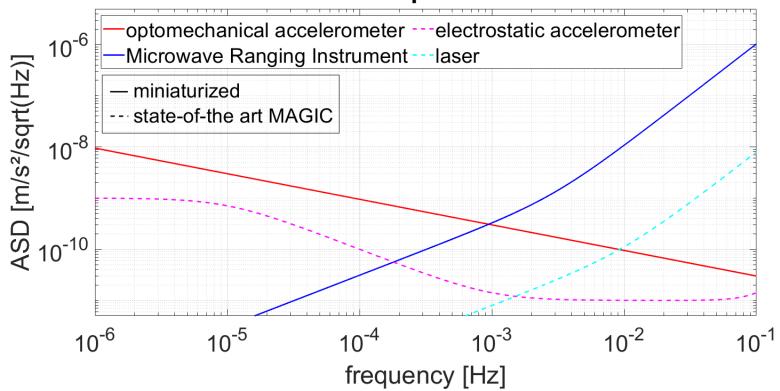
- Electrostatic Micro STAR (MAGIC) → state-of-the art NGGM
- Optomechanical sensor (Hines et al. 2020) → miniaturized footprint: 48 mm x 92 mm, mass: 26 g, power: ca. 1 W

o Inter-Satellite Ranging:

- Laser (MAGIC) → state-of-the art NGGM
- No miniaturized sensor available with < 1 µm
 → Assumption of future instrument with GRACE-FO K-Band Microwave Ranging System accuracy (Kornfeld et al. 2019)



instrument performances



Higher performance of the state-of-the art instruments



Investigation period: 1 month

Inter-satellite distance: ~220 km

Retrieved Signal	Co-estimation of sh gravity fields	AO error	OT error	
	approach	length		
HIS	-	-	yes	yes
AOHIS	"Wiese" d/o 30 (Wiese et al. 20'11)	1 day, 12 h, 6h	-	yes

Orbits	Altitude [km]	Inclination [°]	
Polar	463	89	
Inclined	432	70	

Orbital Parameters taken from MAGIC orbit set 3d_H (Massotti et al. 2021)

Orbital Setup



1 polar pa	air	4 polar pairs	4 polar pairs	18 polar pairs	18 polar pairs
1 orbit plane					18 orbit planes

1 polar + 1 inclined pairs

4 polar + 4 inclined pairs

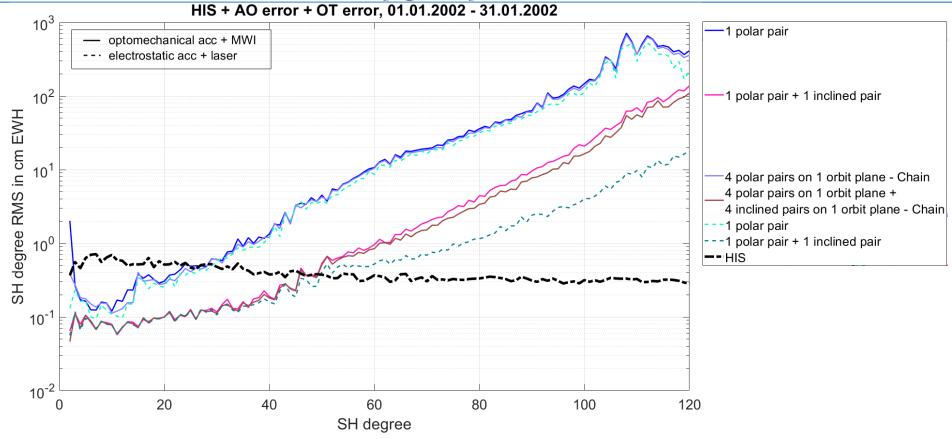
2 orbit planes

8 orbit planes

Simulation Results – monthly gravity field retrieval



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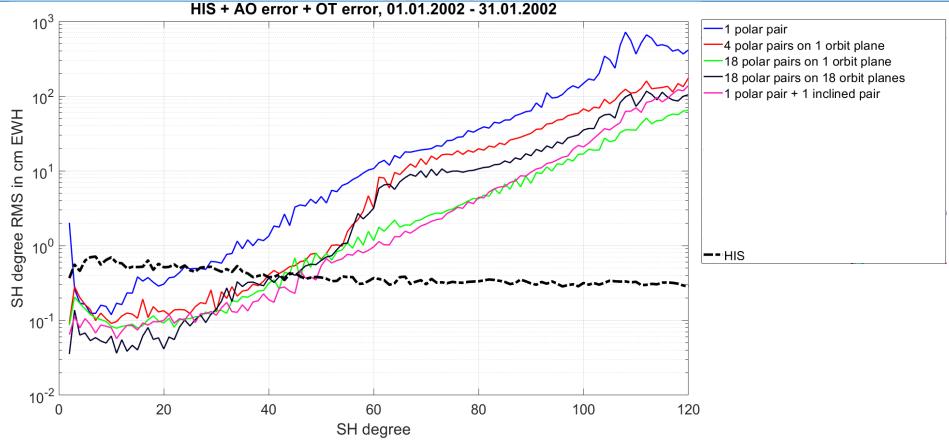


- □ Similar performance with different instruments → temporal aliasing dominating gravity field
- Satellite chains increase only redundancy due to more provided observations but no better spatio-temporal resolution

Simulation Results – monthly gravity field retrieval



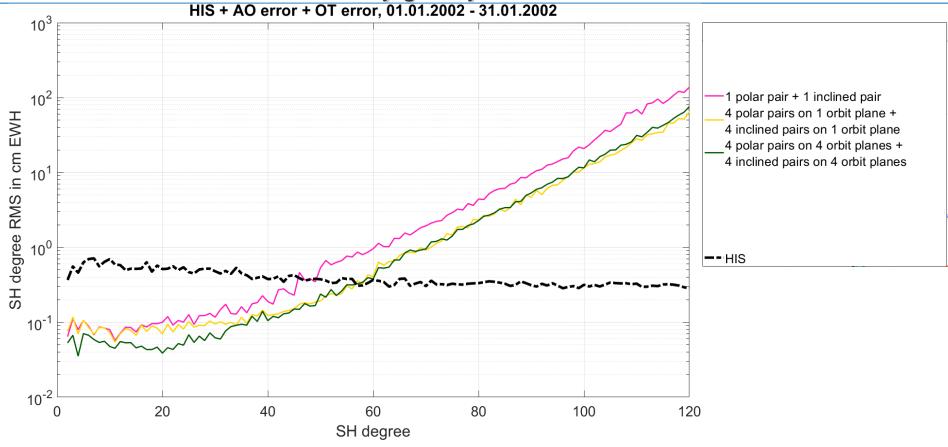
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Increasing the number of polar satellite pairs improves performance and simultaneously decreases the added value from additional observations (for monthly retrieval!)

Simulation Results – monthly gravity field retrieval

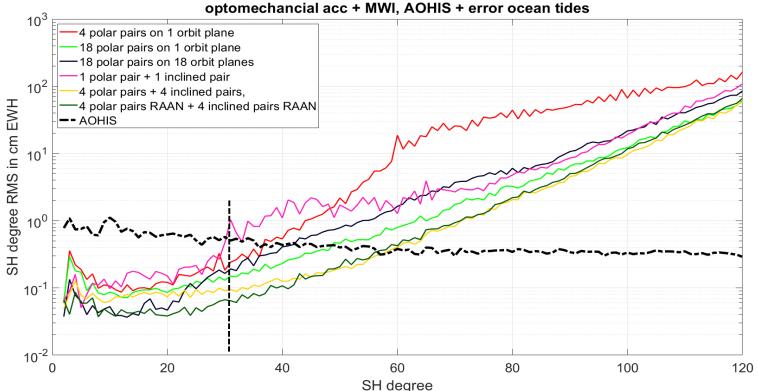




- Additional inclined satellite pairs improve the overall gravity solution
- Spatial distribution is of minor relevance for monthly field retrieval



01.01.2002 - 31.01.2002, 1 daily gravity fields, wiese do 30 optomechancial acc + MWI, AOHIS + error ocean tides

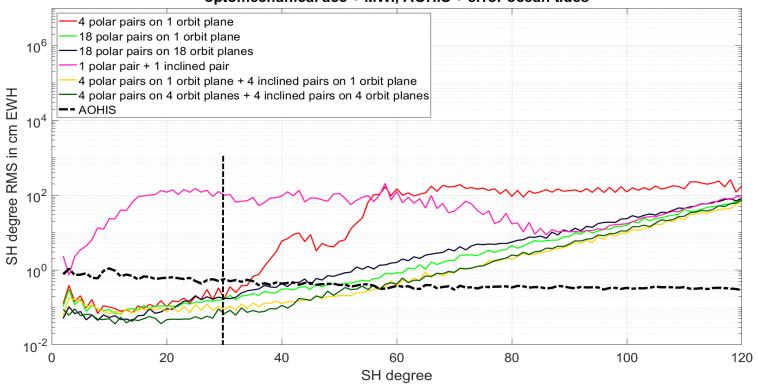


- ☐ All scenarios are able to resolve 1-daily gravity fields
- Observations from inclined orbits and simultaneously a higher amount of satellites pairs perform the best





01.01.2002 - 31.01.2002, 12 hours gravity fields, wiese do 30 optomechanical acc + MWI, AOHIS + error ocean tides

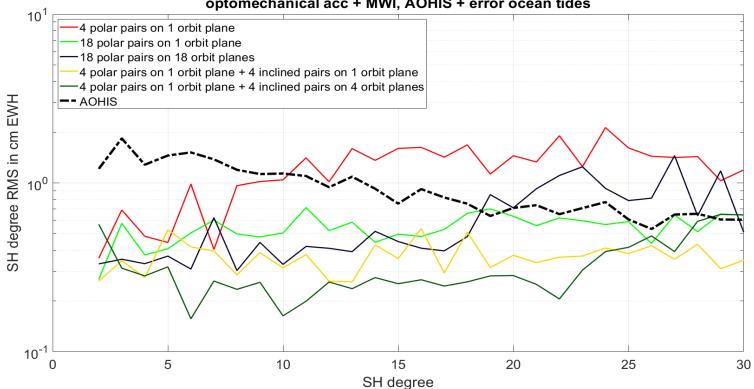


 Configurations with limited spatial coverage are not able to co-estimate half-daily gravity fields and degrade the overall monthly solution





12:00 pm - 12:00 am on 01.01.2002, 12 hours gravity fields, wiese do 30 optomechanical acc + MWI, AOHIS + error ocean tides

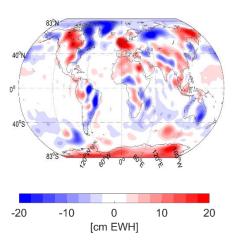


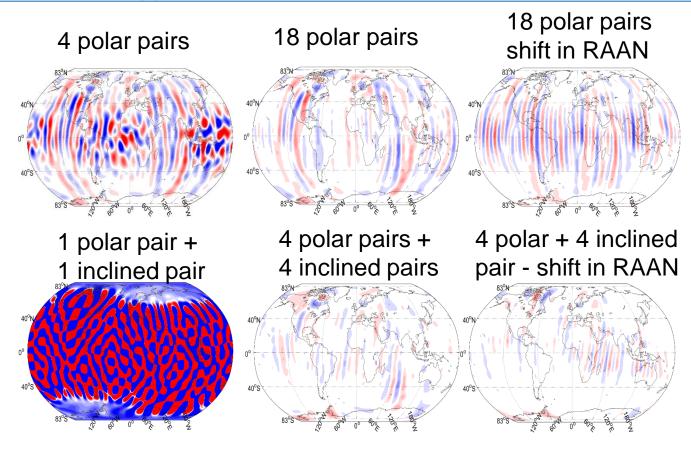
 Configurations with limited spatial coverage are not able to co-estimate half-daily gravity fields and degrade the overall monthly solution

Simulation Results – Wiese approach – 12 h co-estimation



Global estimated halfdaily Gravity Field Residuals from Simulations **AOHIS** compared to Equivalent Signal in Water Height (EWH) on 01.01.2002 12:00 h to 24:00 h

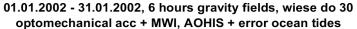


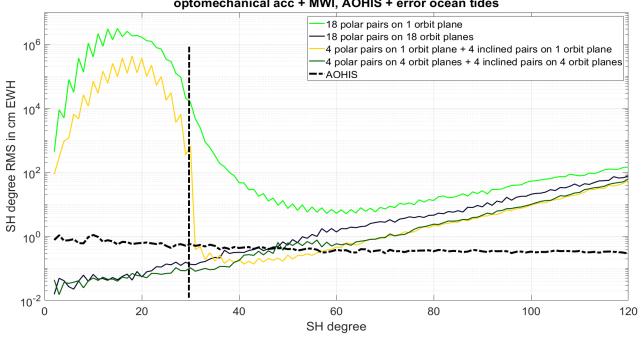


- Configurations with limited spatial coverage are not able to retrieve the
 12 h AOHIS signal
- Reduction of striping pattern with inclinced observations

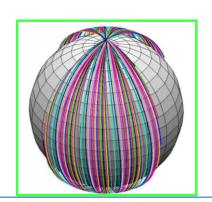
Simulation Results – Wiese approach – 6 h co-estimation



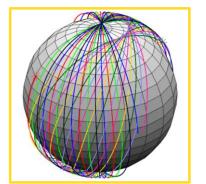


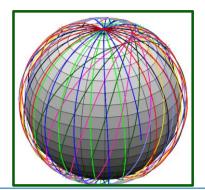


Ground track coverage after 6 hours:











- □ Miniaturized optomechanical accelerometer suitable for a potential CubeSat gravity mission but no candidate available for inter-satellite ranging with < 1 µm</p>
- Co-estimation of short periodic (24 hous, 12 hours and 6 hours) standalone gravity fields
- Importance level of different criteria for NGGMs (application-dependent):

Added value

Number of inter-satellite links

Number of satellite pairs

Spatial distribution of the satellites

Observations from different inclined orbits



- Optimisation of ground track coverage with adapted orbits (sub-cycle)
- O What are potential candidates for ranging instruments?
 - → Any ideas and suggestions are highly appreciated



References



- https://www.asi.it/wpcontent/uploads/2021/03/MAGIC_NGGM_MCDO_MRD_v1_0-signed2.pdf (MRD MAGIC)
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References



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