



Traceable Radiometry Underpinning Terrestrial- and Helio- Studies (TRUTHS) – A ‘gold standard’ imaging spectrometer in space to support climate emergency research

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On behalf of the TRUTHS ESA and Science Teams

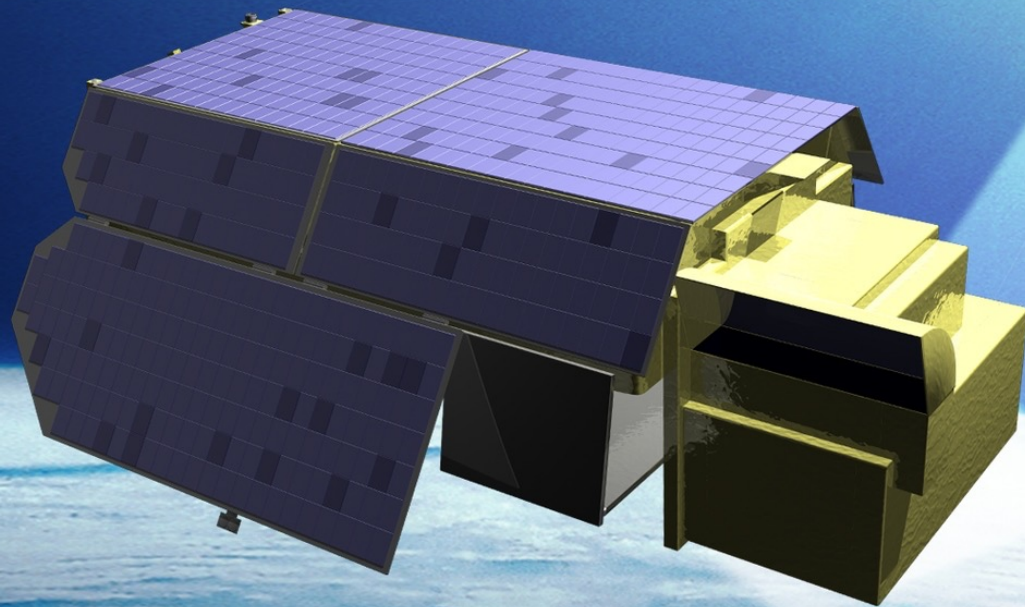
EGU23, 25 April 2023



TRUTHS

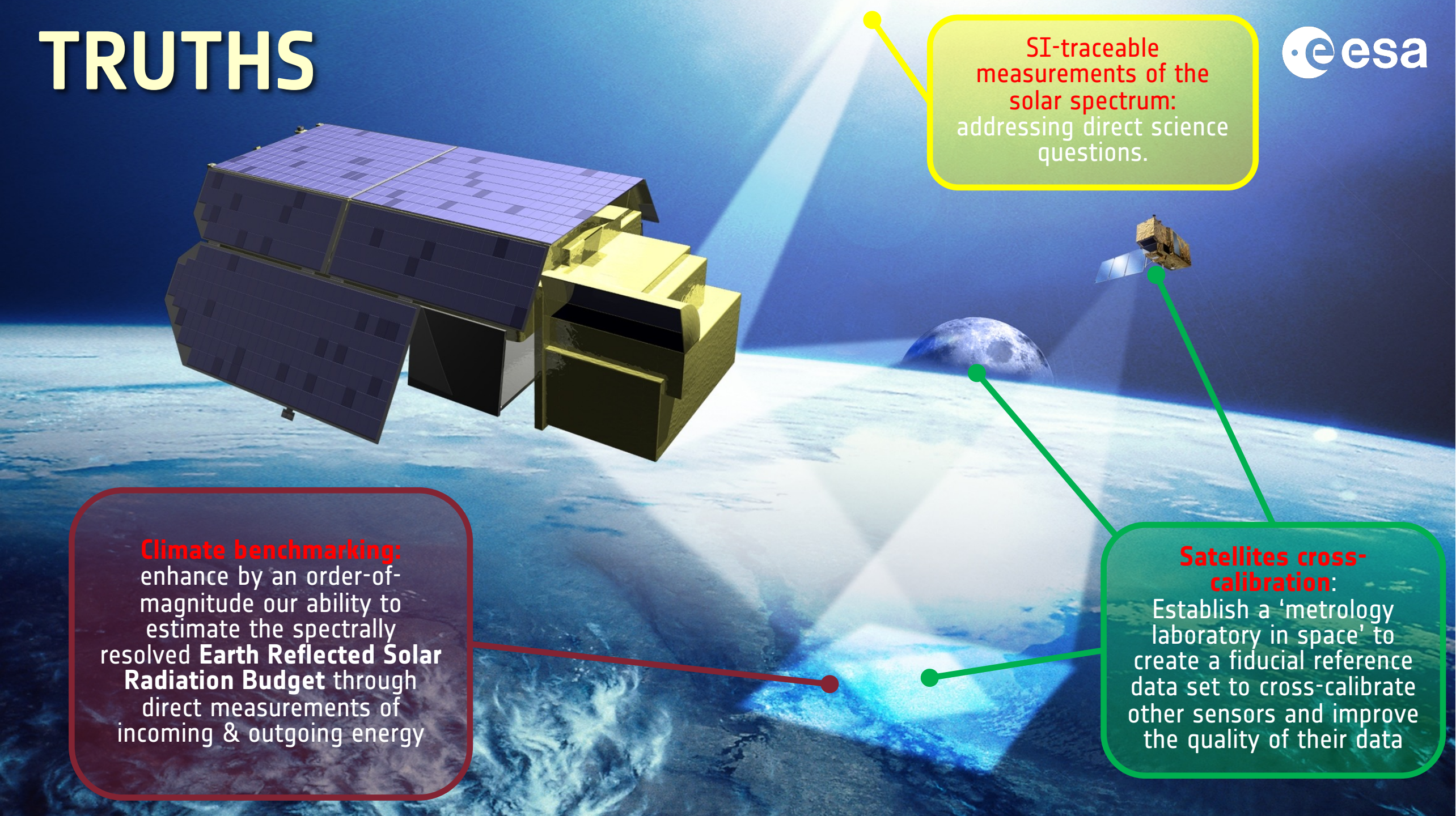


SI-traceable
measurements of the
solar spectrum:
addressing direct science
questions.

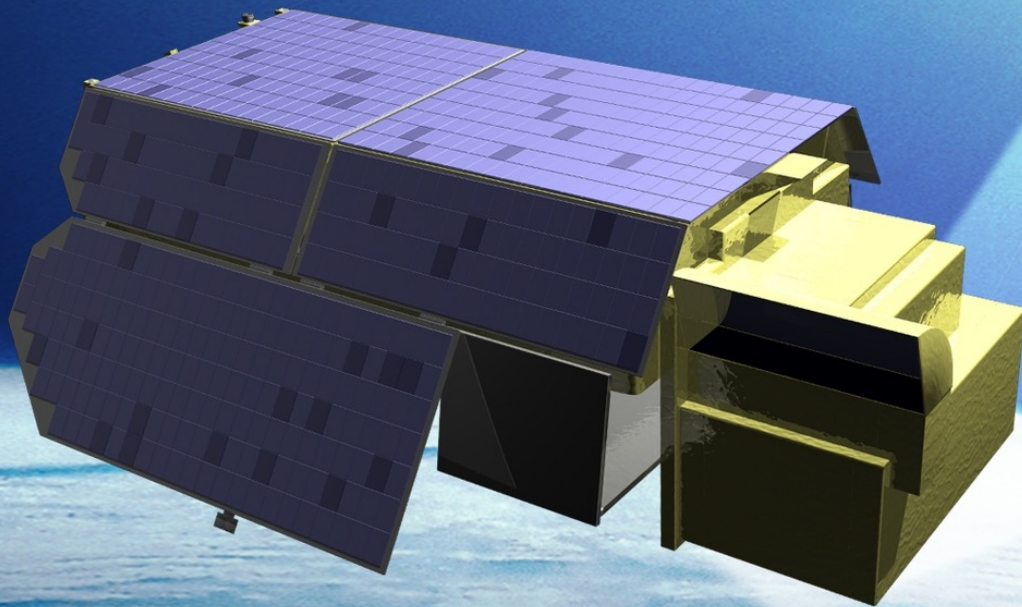


Climate benchmarking:
enhance by an order-of-
magnitude our ability to
estimate the spectrally
resolved **Earth Reflected Solar
Radiation Budget** through
direct measurements of
incoming & outgoing energy

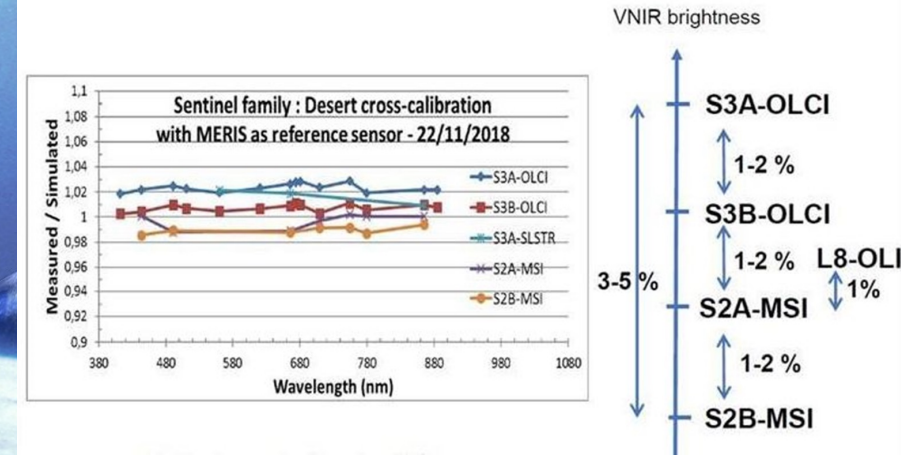
**Satellites cross-
calibration:**
Establish a 'metrology
laboratory in space'
to create a fiducial reference
data set to cross-calibrate
other sensors and improve
the quality of their data



TRUTHS

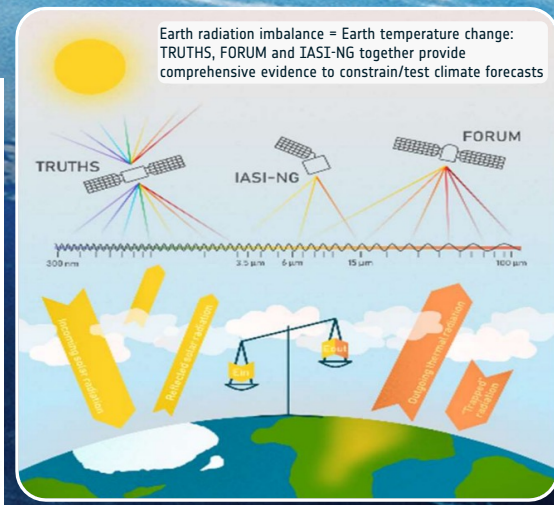
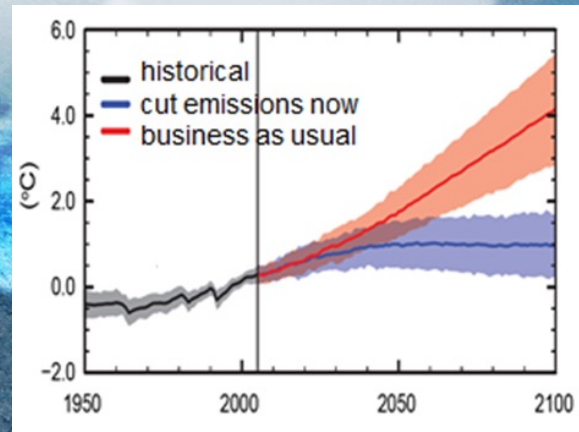


TRUTHS cross-calibration:
bias removal, improving accuracy of other sensors,
datasets re-calibration



TRUTHS climate Benchmarking:
more precise predictions

- Optical mission for measuring incoming solar and outgoing reflected radiation
- **Metrology lab in orbit:** flying a primary calibration standard traceable to SI Units



What does TRUTHS do?

Measures incoming and Earth/Moon reflected radiation from the Sun

- 320 to 2400 nm @ ~4 nm intervals (1 nm for solar UV)
- Global nadir @ 50 m ground resolution with 100 km swath (capability)
- Target radiometric uncertainty of 0.3% (k=2)

Establishing a benchmark of the radiation state of the planet at ToA (radiance/reflectance) & BoA surface reflectance to help enable:

Observations

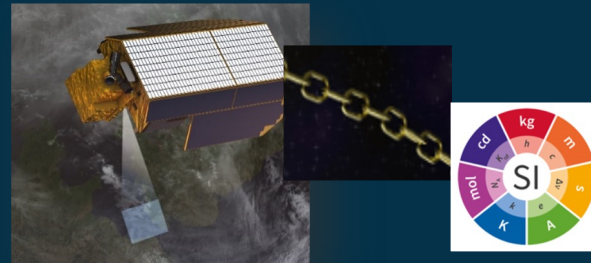
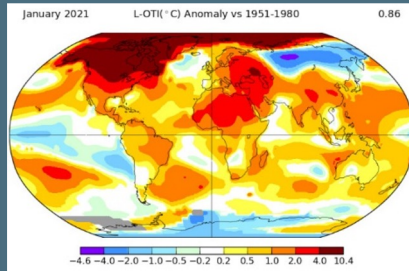
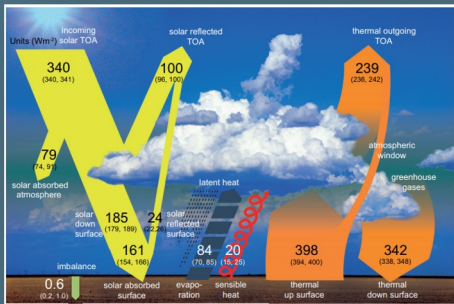
- Benchmark
- monitoring
- Litigation
- algorithm improvement

Calibration

- Interoperability
- data-gaps
- performance
- Utility

Climate action: Supporting 'Net Zero'

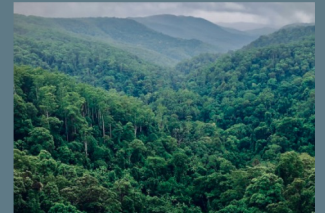
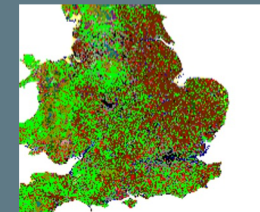
Climate sensitivity/response



Climate action/mitigation



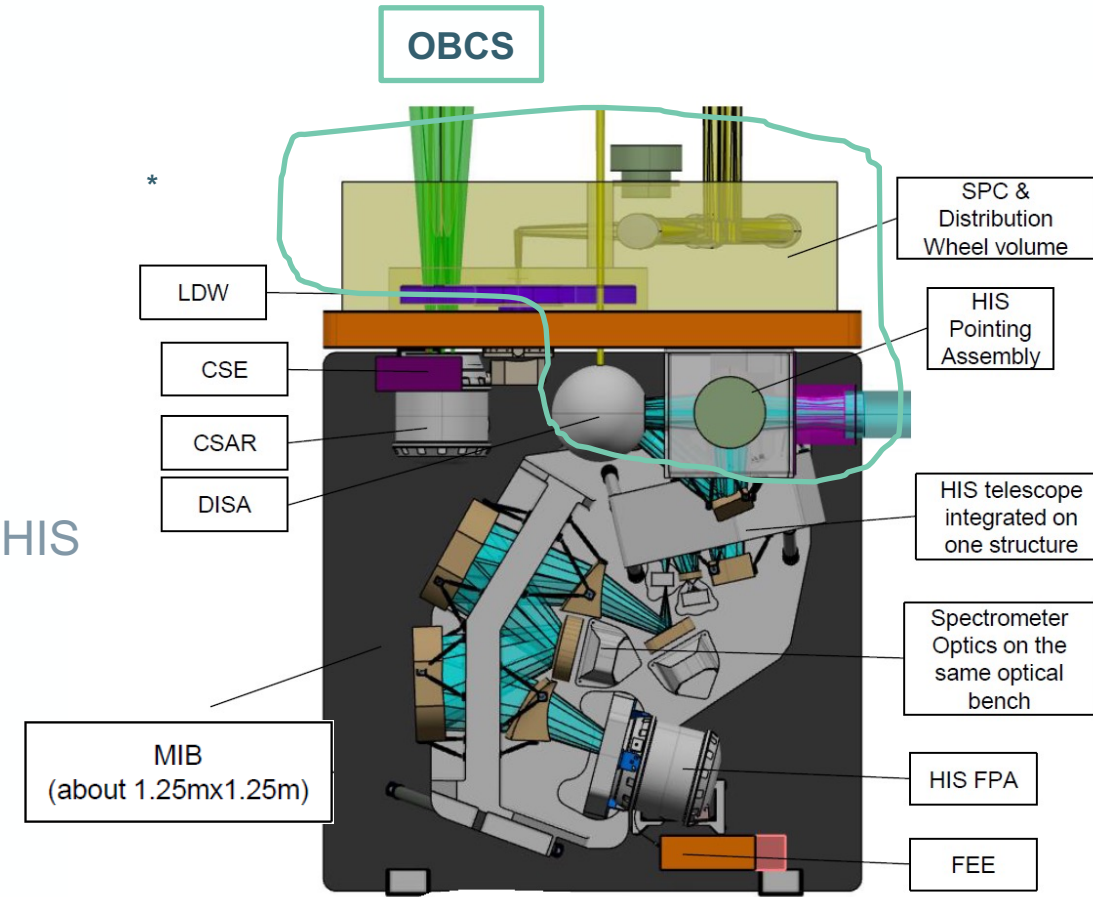
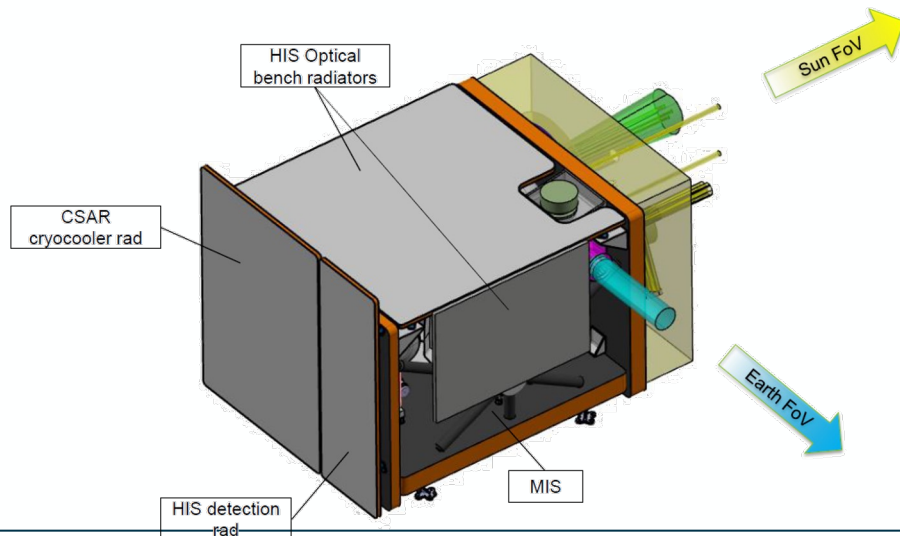
Adaptation/sustainability



Payload Overview

Payload, composed of three elements:

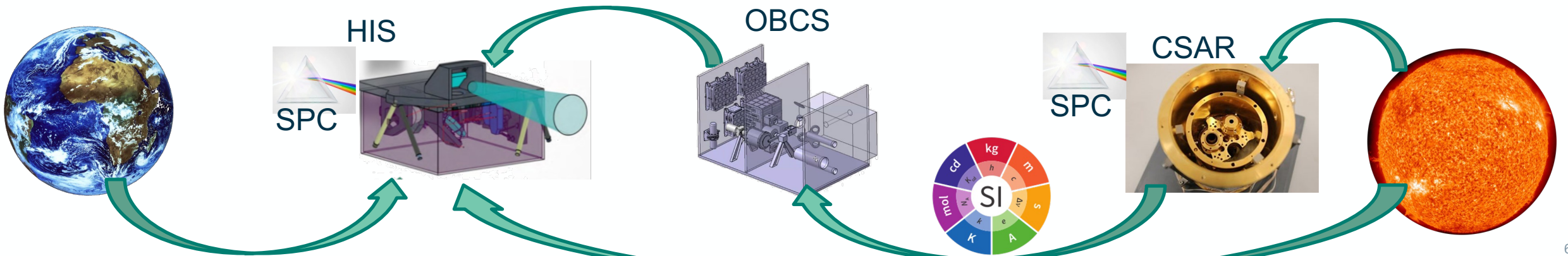
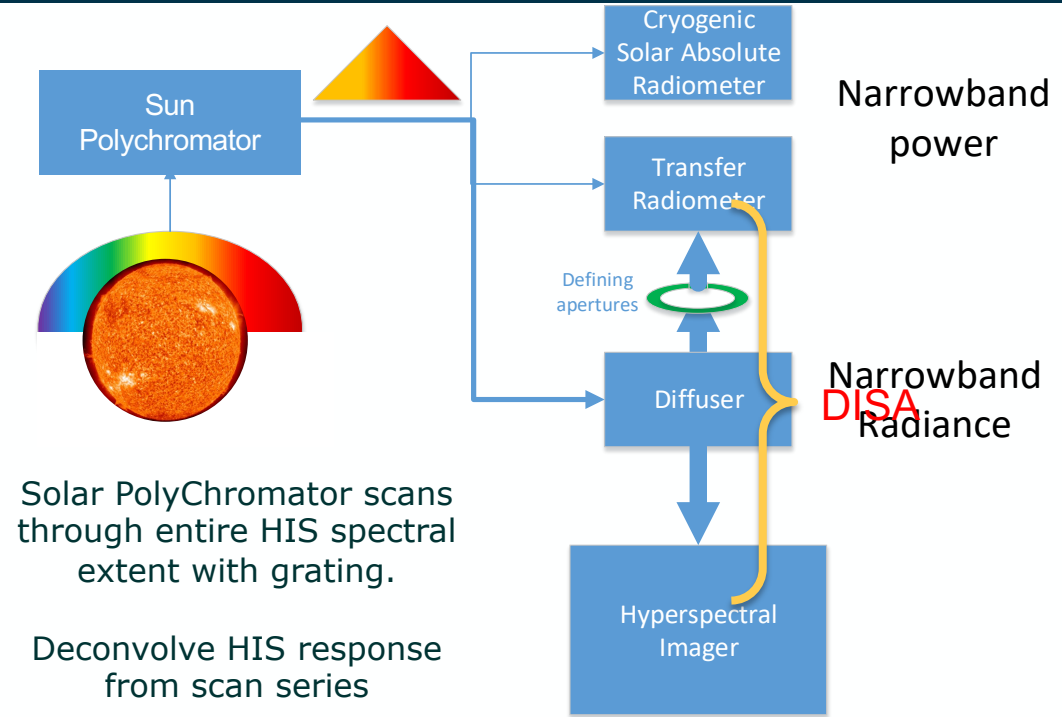
- **CSAR** (Cryogenic Solar Absolute Radiometer) – operated at 60 K (cryocooler*), the “primary standard”
- **HIS** (Hyperspectral Imaging Spectrometer) – UV to SWIR (320-2400 nm), single detector, 50 m resolution, 100 km swath. Detector actively cooled at 150 K
- **OBCS** (On-Board Calibration System) – transferring the CSAR solar absolute (SI) measurement to the HIS



*Cryo-cooler Assembly – recurrent from THRISHNA mission at ISRR baseline

TRUTHS in-orbit calibration philosophy

1. Cryogenic Absolute Solar Radiometer (CSAR)
 - Measures optical power in Sun Polychromator (SPC) output
 - SPC generates beams of 'monochromatic' radiation from sun distributed to different parts of the calibration system.
2. Double Integrating Sphere Assembly (DISA)
 - Conversion from power to radiance.
 - Calibrated, relative mode, via direct sun + HIS measurement.
 - Calibrates HIS gain via SPC output.
3. Concept of Operations based on geometric knowledge & stability
 - Calibrating out in-flight degradation only.
 - Repeatability assumptions based on mechanical and thermal control.

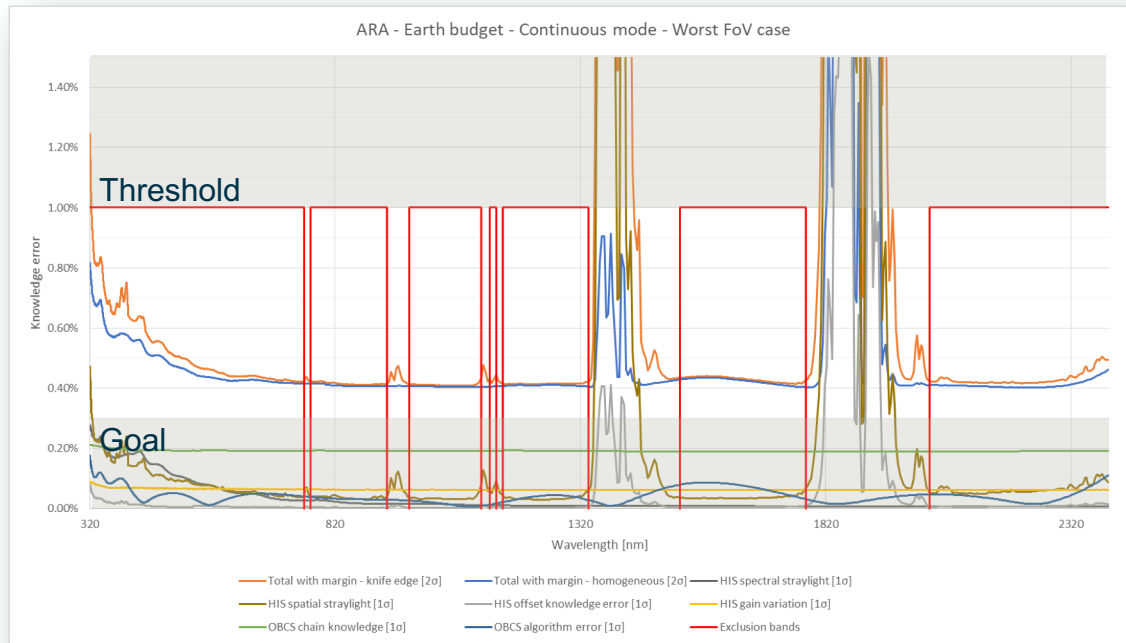


Absolute Radiometric Accuracy (ARA)

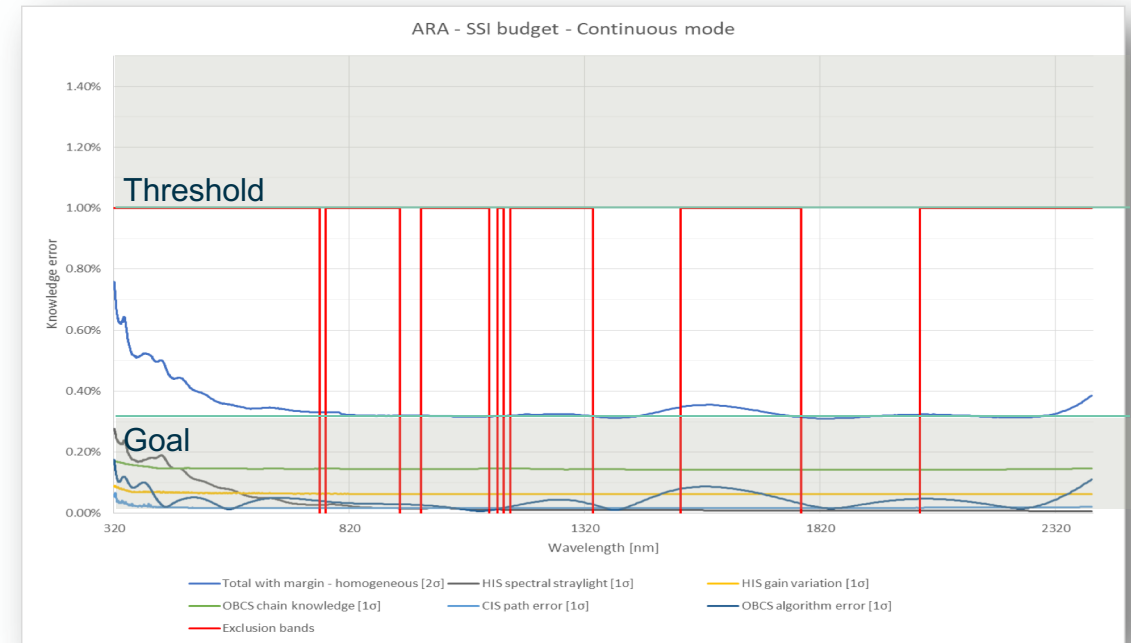
Mission requirement:

| MRD-ID | Type | Value |
|-------------|--------------------|--|
| MRD-OBS-310 | ERU ERSR, SSI, LSI | The Expanded Radiometric Uncertainty for ERSR, SSI and LSI measurements shall be better than 0.3% (G) / 1% (T). |

Earth Reflected Solar Radiance



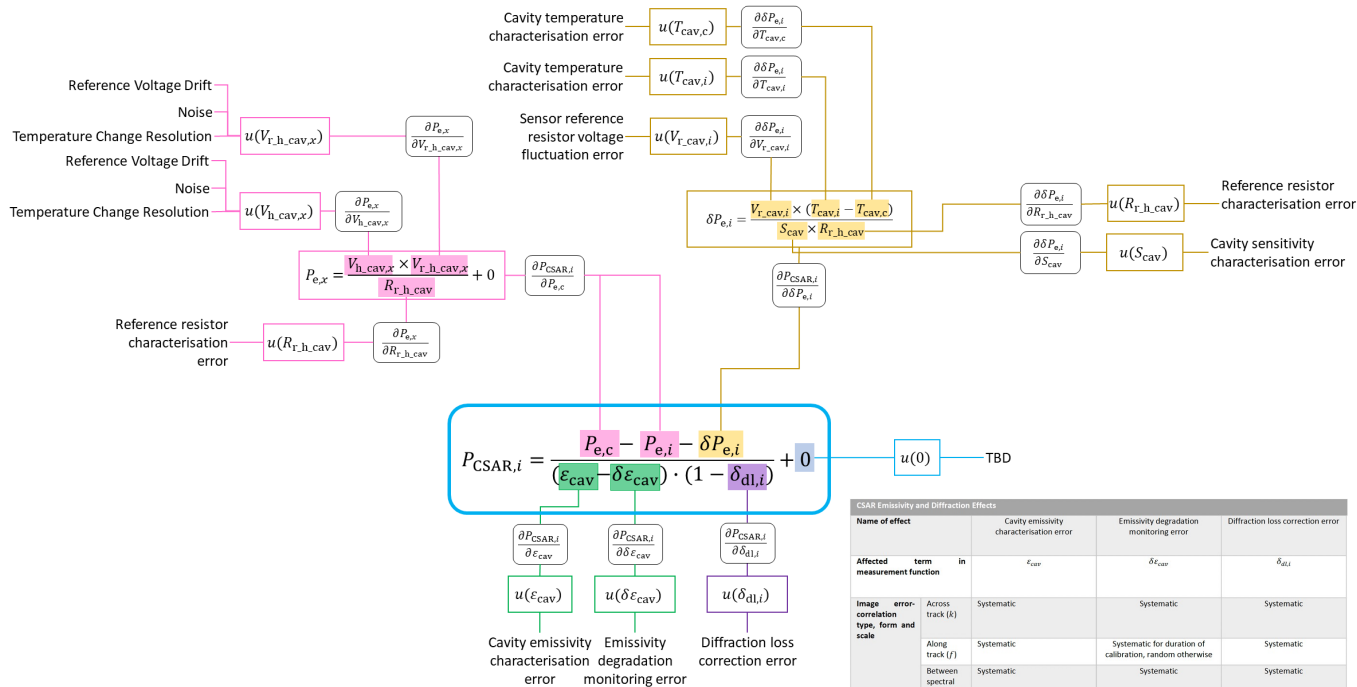
Spectral Solar Irradiance



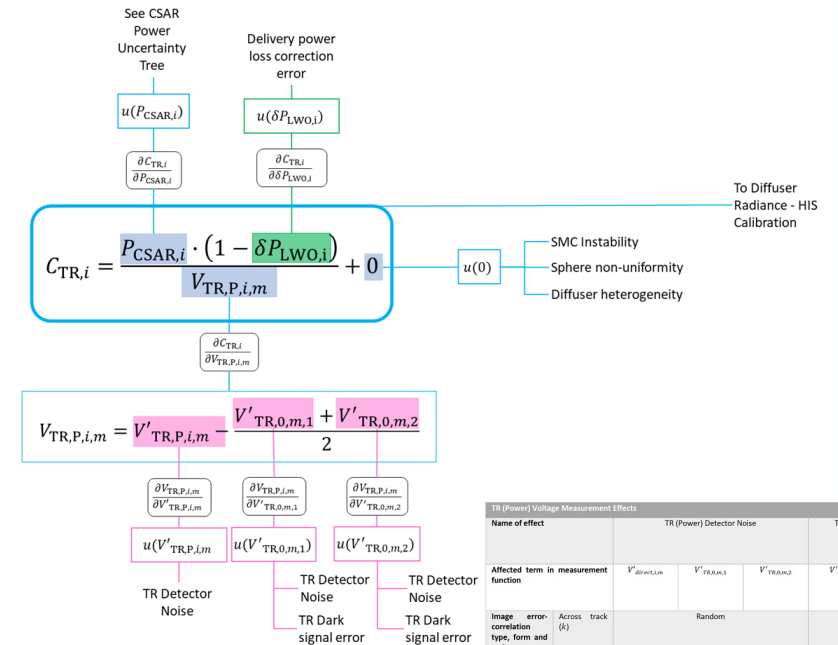
Application of Metrological principles

FIDUCEO like analysis of end to end traceability and uncertainties establishing measurement equation and errors sources together with associated uncertainties for the end to end measurement.

CSAR power calibration



Transfer Radiometer power transfer (to be updated)

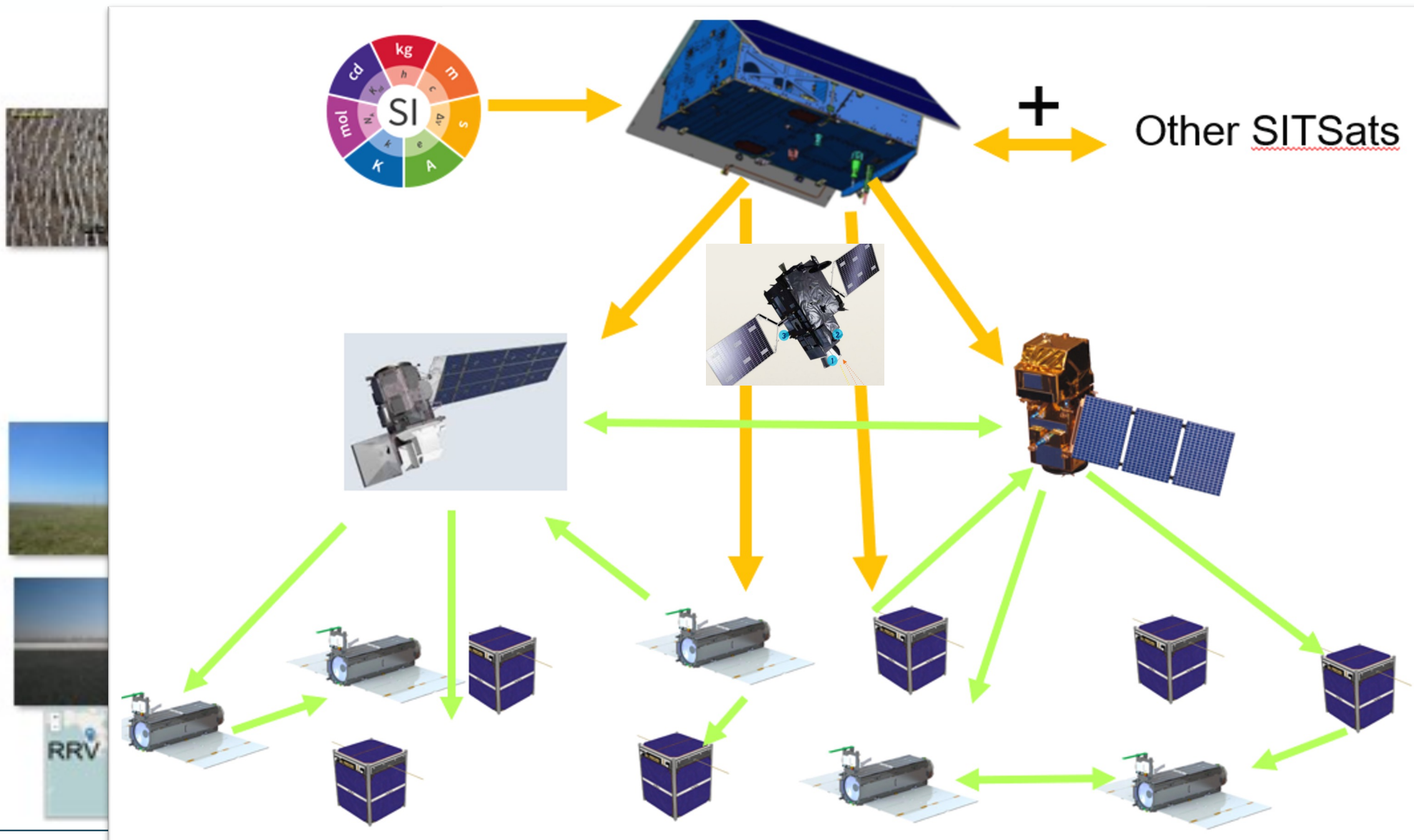


| CSAR Emissivity and Diffraction Effects | | | |
|--|--|---|---|
| Name of effect | Cavity emissivity characterisation error | Emissivity degradation monitoring error | Diffraction loss correction error |
| Affected term in measurement function | ϵ_{cav} | $\delta \epsilon_{cav}$ | δ_{dl} |
| Image error-correlation type, form and scale | Across track (k): Systematic Along track (l): Systematic Between spectral pixels (i): Systematic | Systematic | Systematic |
| Uncertainty | PDF shape: Rectangular units: - magnitude: 0.01% | Rectangular | Rectangular |
| Source | [RD-10] | [RD-10] | [RD-10] |
| Sensitivity coefficient | $\frac{\partial L_{k,l}}{\partial \epsilon_{cav}}$ | $\frac{\partial L_{k,l}}{\partial \delta \epsilon_{cav}}$ | $\frac{\partial L_{k,l}}{\partial \delta_{dl}}$ |

| TR (Power) Voltage Measurement Effects | | | | | |
|--|---|---|---|---|--|
| Name of effect | TR (Power) Detector Noise | | | TR (Power) Dark Signal Error | |
| | $V'_{TR,0,m,1}$ | $V'_{TR,0,m,1}$ | $V'_{TR,0,m,2}$ | $V'_{TR,0,m,1}$ | $V'_{TR,0,m,2}$ |
| Affected term in measurement function | | | | | |
| Image error-correlation type, form and scale | Across track (k): Random | Along track (l): Systematic for duration of calibration, random otherwise | Between spectral pixels (i): Random | Random | Systematic for duration of calibration, random otherwise |
| Uncertainty | PDF shape: - | Rectangular | Rectangular | Rectangular | Rectangular |
| units | - | pA | pA | pA | pA |
| magnitude | TBD | 5 | 5 | 5 | 5 |
| Source | [RD-9] | [RD-9] | [RD-9] | [RD-9] | [RD-9] |
| Sensitivity coefficient | $\frac{\partial L_{k,l}}{\partial V'_{TR,0,m,1}}$ | $\frac{\partial L_{k,l}}{\partial V'_{TR,0,m,1}}$ | $\frac{\partial L_{k,l}}{\partial V'_{TR,0,m,2}}$ | $\frac{\partial L_{k,l}}{\partial V'_{TR,0,m,1}}$ | $\frac{\partial L_{k,l}}{\partial V'_{TR,0,m,2}}$ |

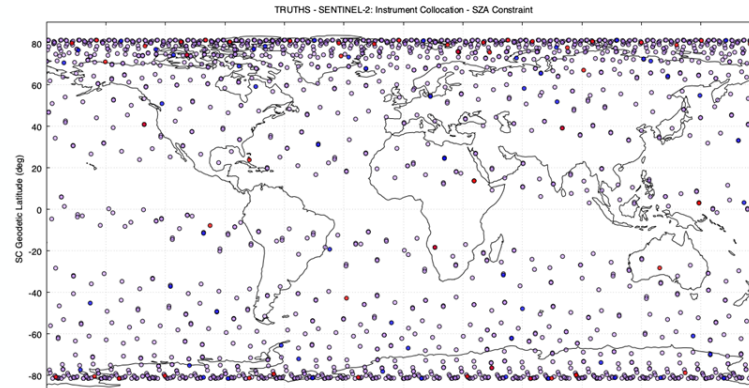
Hunt and Fahy, NPL, 2022

SI-Traceability to Cal/Val infrastructure

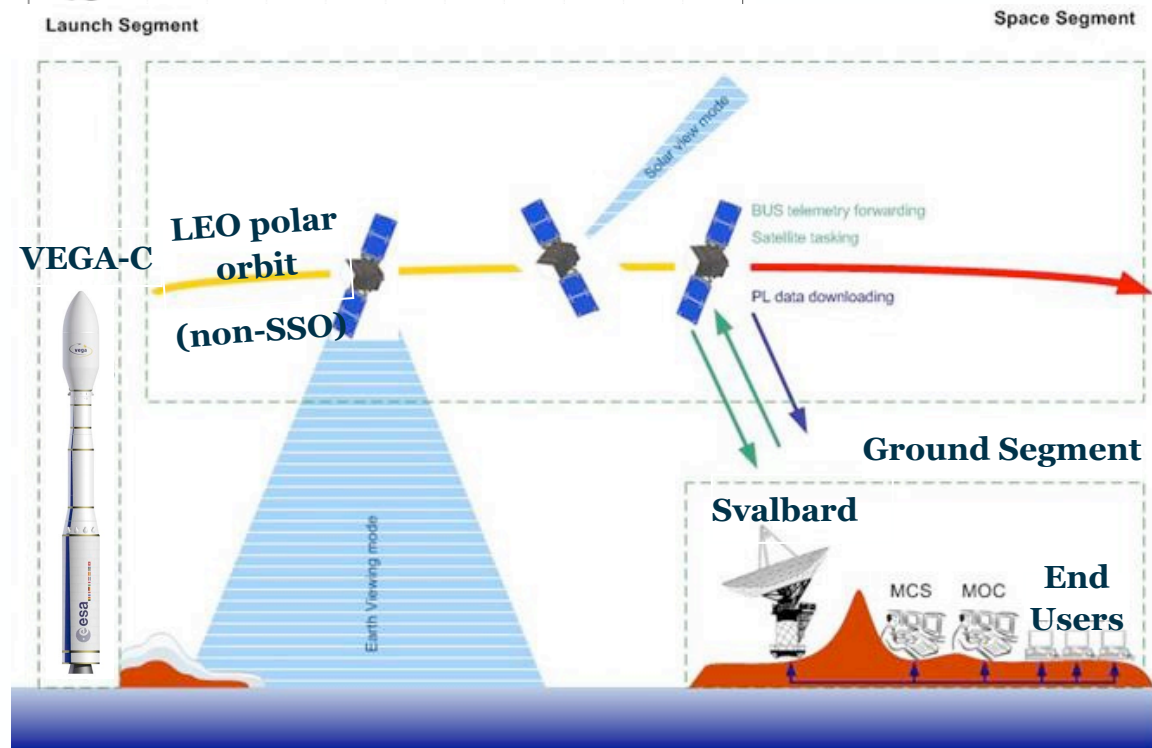
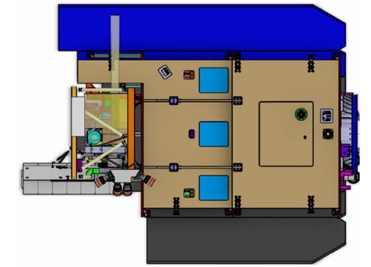


System Architecture Overview

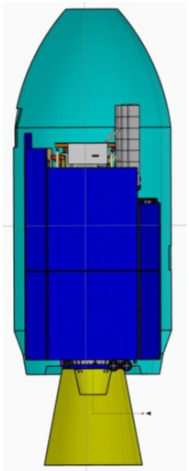
- Lifetime 5 years + 3 extension
- Launch foreseen in 2030
- Space Segment:
 - Orbit 614 km, polar (90°) non-SSO
 - 1 satellite – agile, design for non-SSO
 - Novel Payload (CSAR, OBCS, HIS)
- Launch Segment:
 - Vega-C (-E) single launch
- Ground Segment
 - 1 polar station (baseline)
 - lossless compression (baseline)
 - Routine FOS in UK
 - PDGS in UK + data access at ESRIN
 - **ESA free and open data access policy**



Space Segment



Launch Segment



TRUTHS from 2030 onwards will help initiate a sustainable long-term international climate & calibration observatory as direct response to international requests

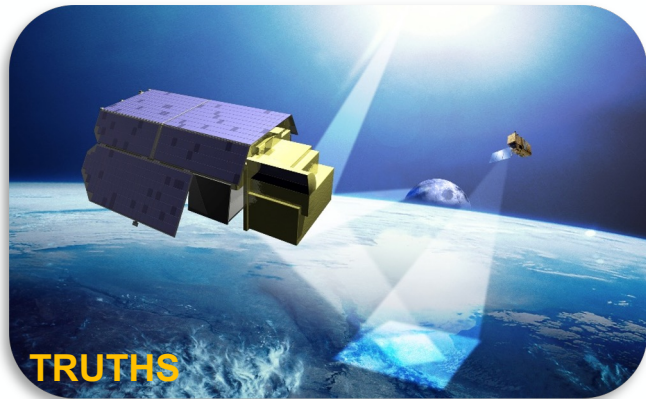
NASA CLARREO-Pathfinder (CPF) 'sister mission' which will be launched to the ISS in 2025.

- Hope for overlap!
- Also potential overlap with Chinese Libra and other SITSats

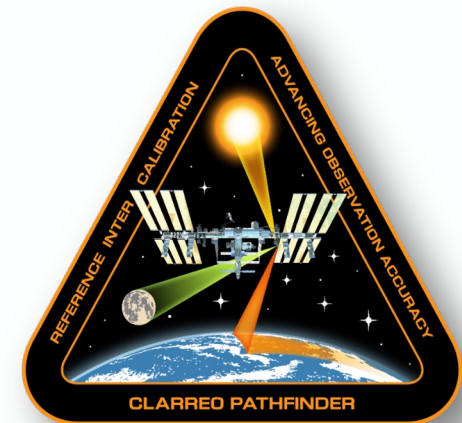
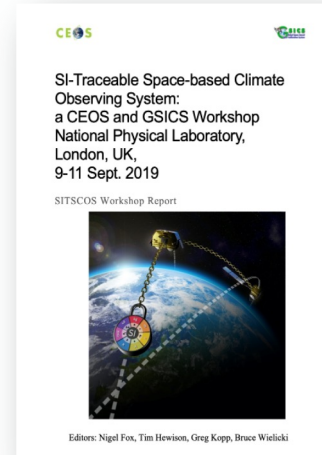
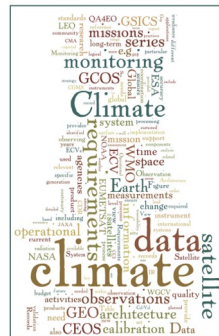


Guaranteeing the future of space activities by protecting the environment

TRUTHS & CPF SITSats will provide unique and critical information for understanding and monitoring Climate and Environmental change from space and support climate action



Strategy Towards an Architecture for Climate Monitoring from Space





TRUTHS Vision: mission assets and perspectives



• An international climate observatory

- The need to cover **long time series** and many solar cycles calls for long-term operation.
- International cooperation and data interoperability among different missions worldwide will be implemented
- NASA already agreed to a TRUTHS/CLARREO-Pathfinder framework.
- Initiating CEOS WGCV (& GSICS hopefully) task group on SITSats in general.

• An operational service for institutional and commercial satellites

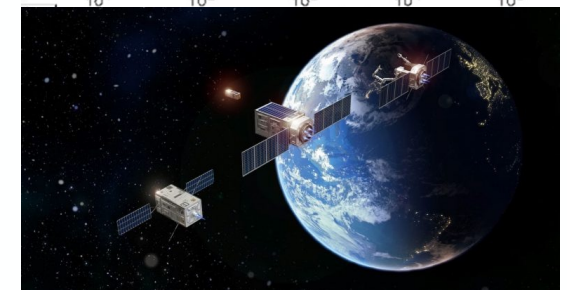
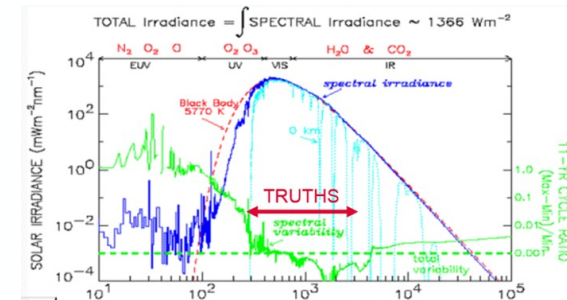
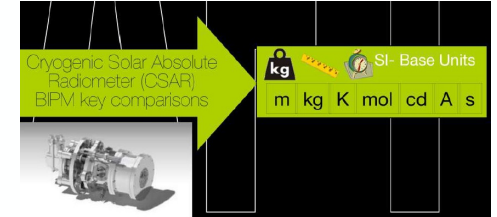
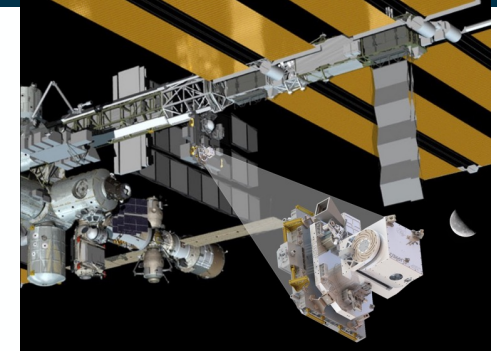
- The capability to cross-calibrate other satellites opens up to an operational service for improving the quality of other optical satellites, either institutional (e.g., Copernicus and Eumetsat series) and commercial.
- The traceability to SI units and rigorous uncertainty tracing makes the measurements unambiguous and trustworthy

• A new concept for next generation optical missions

- An optimized TRUTHS calibration system might become a novel package for optical instruments calibration.
- Smaller satellites might opt to fully rely on calibration from TRUTHS data service and optical payload can be conceived lighter and simpler,

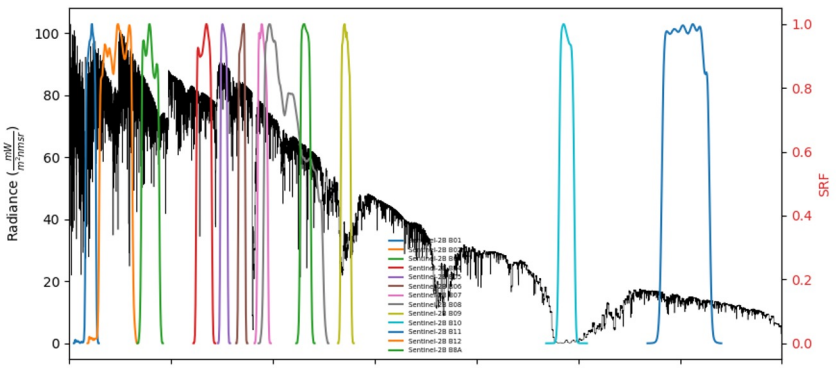
• A step towards a System of Systems

- Efficient use of space and data assets: TRUTHS improves the performance of many others and the TRUTHS data on calibration sites permit to improve new and existing datasets, even taken in the past.
- TRUTHS concepts fits in the strategic view of making space assets interconnected and result of a distributed effort



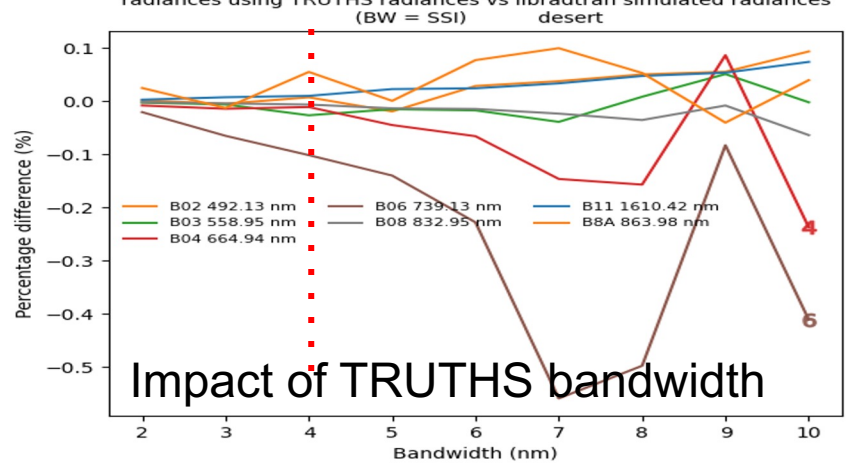
Transferring TRUTHS accuracy to other Sensors: establishing mission requirements (S2S calibration)

Graph of Sentinel-2B msi SRF overlaid on simulated TOA radiance spectrum (from libRadtran)

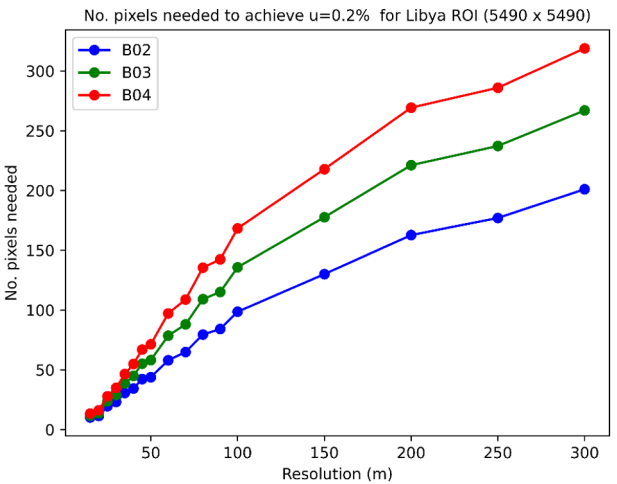
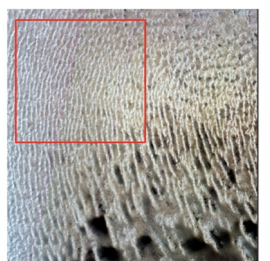
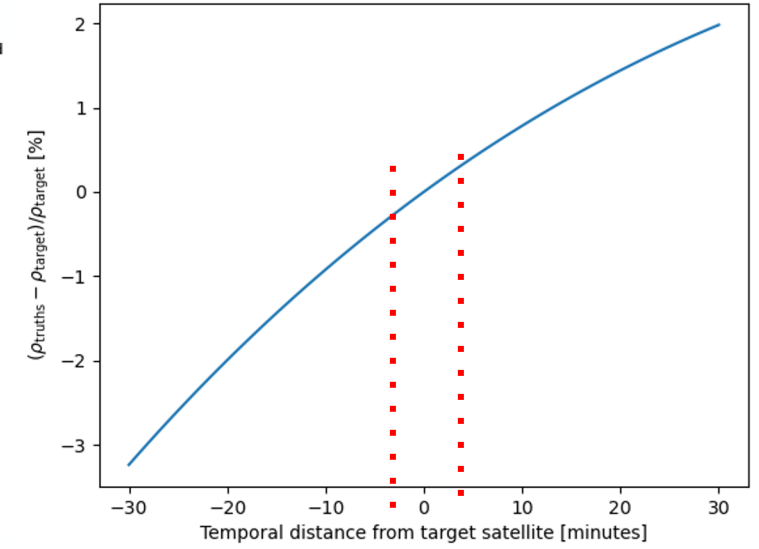


Spectral Bands of Sentinel 2

Graph of percentage difference between Sentinel-2B msi band integrated radiances using TRUTHS radiances vs libradtran simulated radiances (BW = SSI)

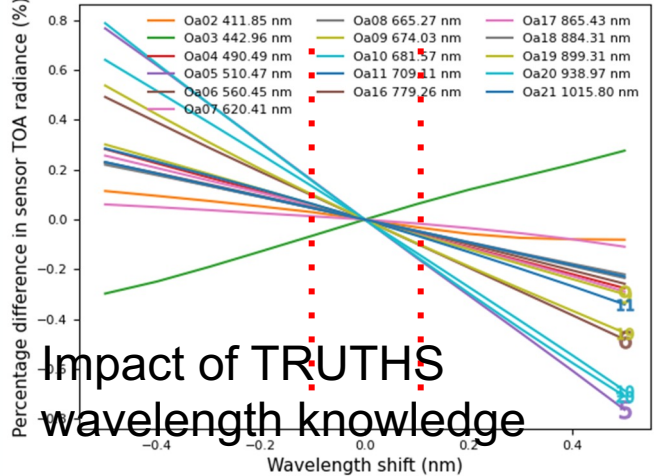


Impact of TRUTHS bandwidth

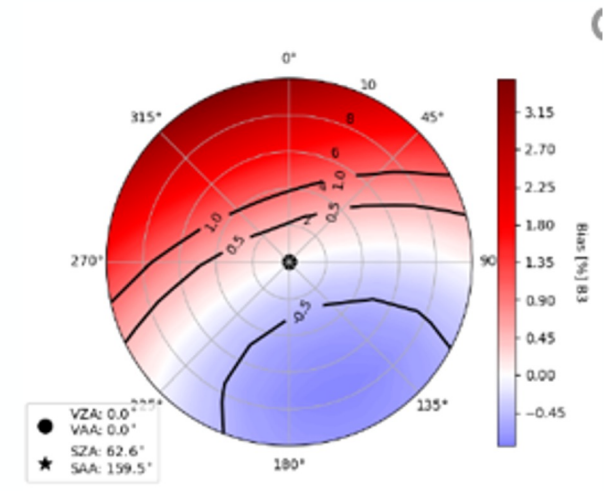


Uniformity of Cal Target (Libya 4) area to be sampled (2.5 km to achieve 0.2% @ 50 m)

Graph of percentage difference in Sentinel-3A olici TOA radiance due to wavelength shift in TRUTHS (SSI = 4 nm, BW = 4 nm): Ocean



Impact of TRUTHS wavelength knowledge

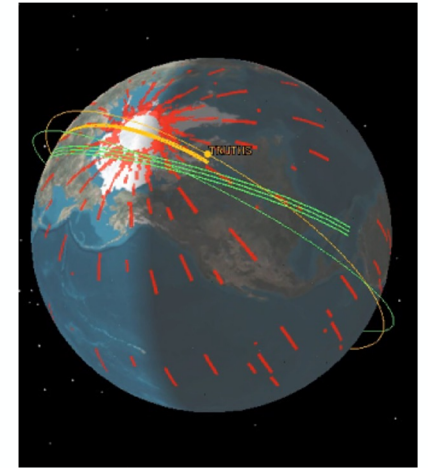
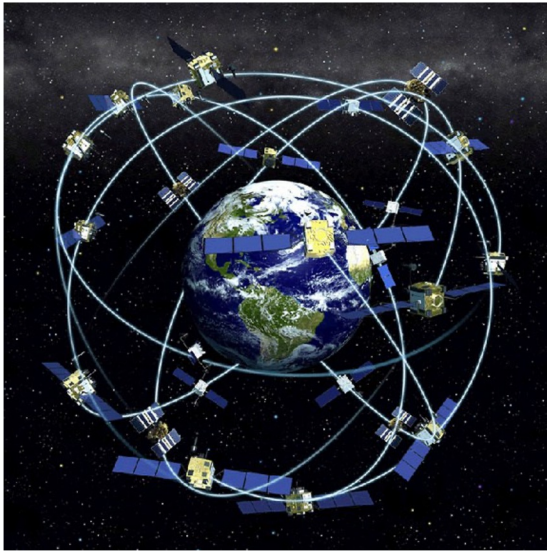


(d) MODIS B3 (460nm); AOT (550nm) = 0.4

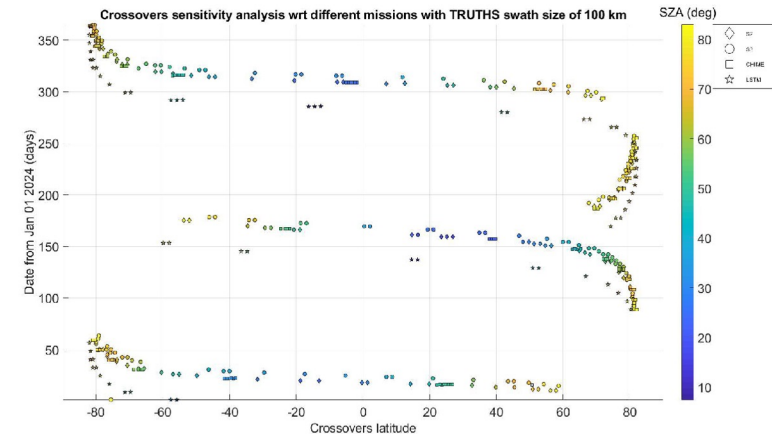
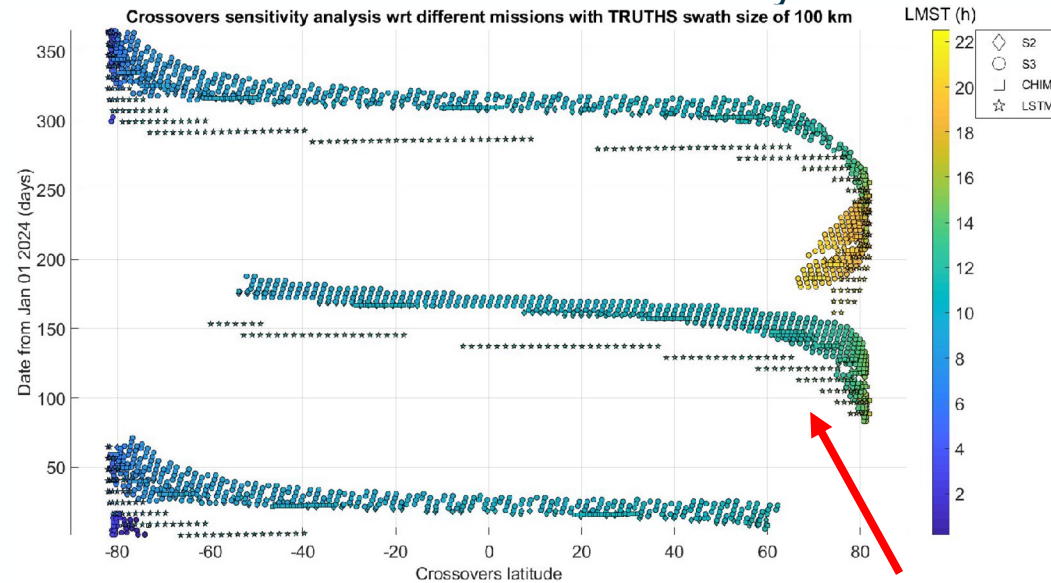
Reference Calibration

- Enables interoperability & Harmonisation
 - Prospect of 'certified calibration'

TRUTHS 90° pole to pole orbit, observing through the diurnal cycle, allows many opportunities to overpass orbit of sun-synchronous sensors



Summary after 6 months



TRUTHS provides the means to transform global EO system, including constellations of micro-sats so they deliver traceable scientific/climate quality observations -

<30 s time difference Swath overlap

1 year of near perfect nadir overlaps for TRUTHS & satellite under test

(<1° (no pointing) <30 s time difference

