



Active mitigation of spaceborne radio frequency interference for VLBI

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Background



- VLBI equipment very sensitive
- ground-based **radio frequency interference** (RFI) is already a severe problem (WiFi, 5G...)
- last years, tremendous **increase of satellites** in space



- some will emit signals in VLBI frequency range
→ spaceborne RFI
- can cause **non-linearities** in our observations or might **saturate receivers**
- also planned InSAR mission that will emit strong radiation that can potentially be **damaging**

Investigated Starlink evolutions



- Three phases*
 - Phase 1: **1584** satellites
 - Phase 2: **4408** satellites
 - Phase 3: **34408** satellites

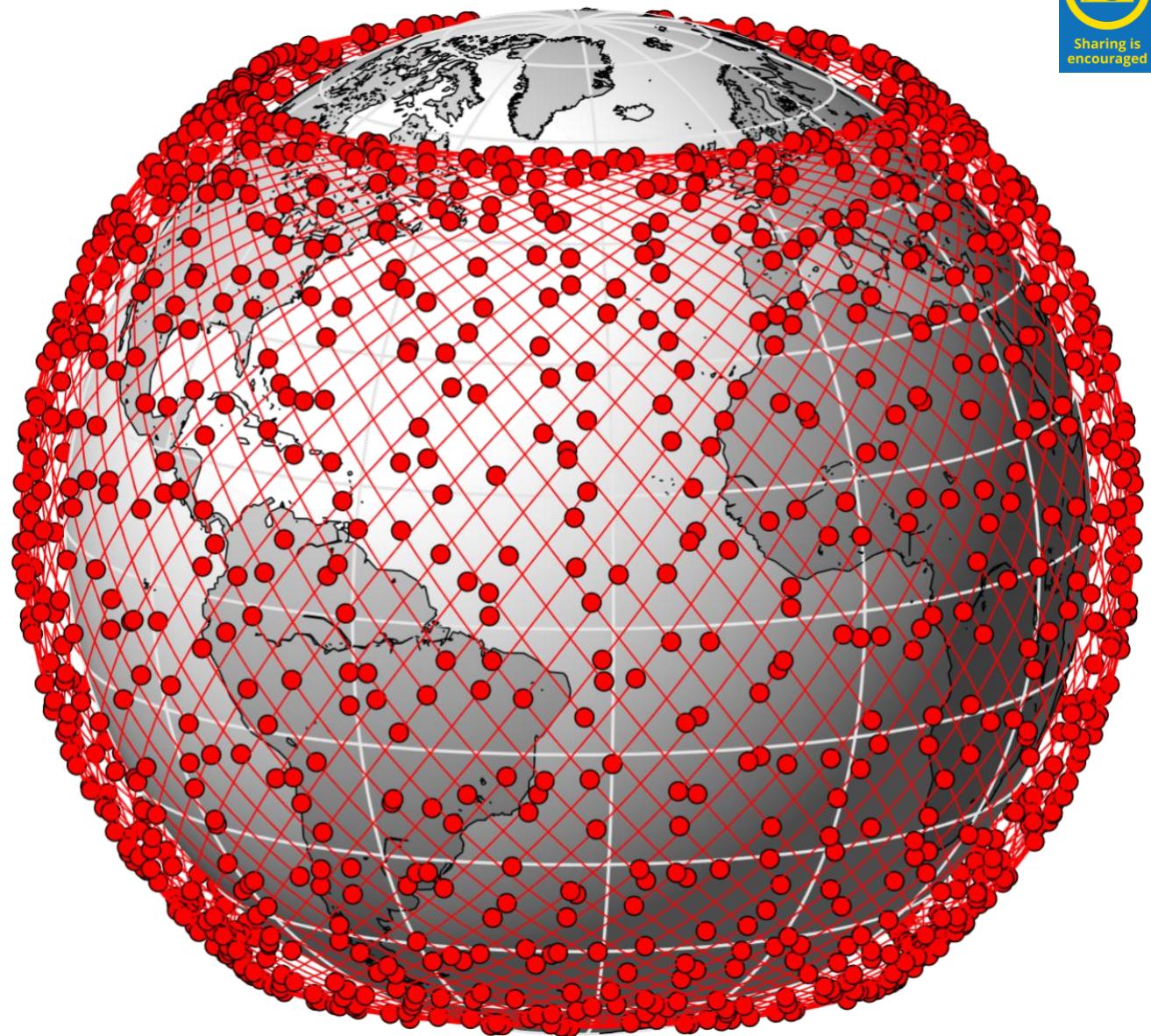
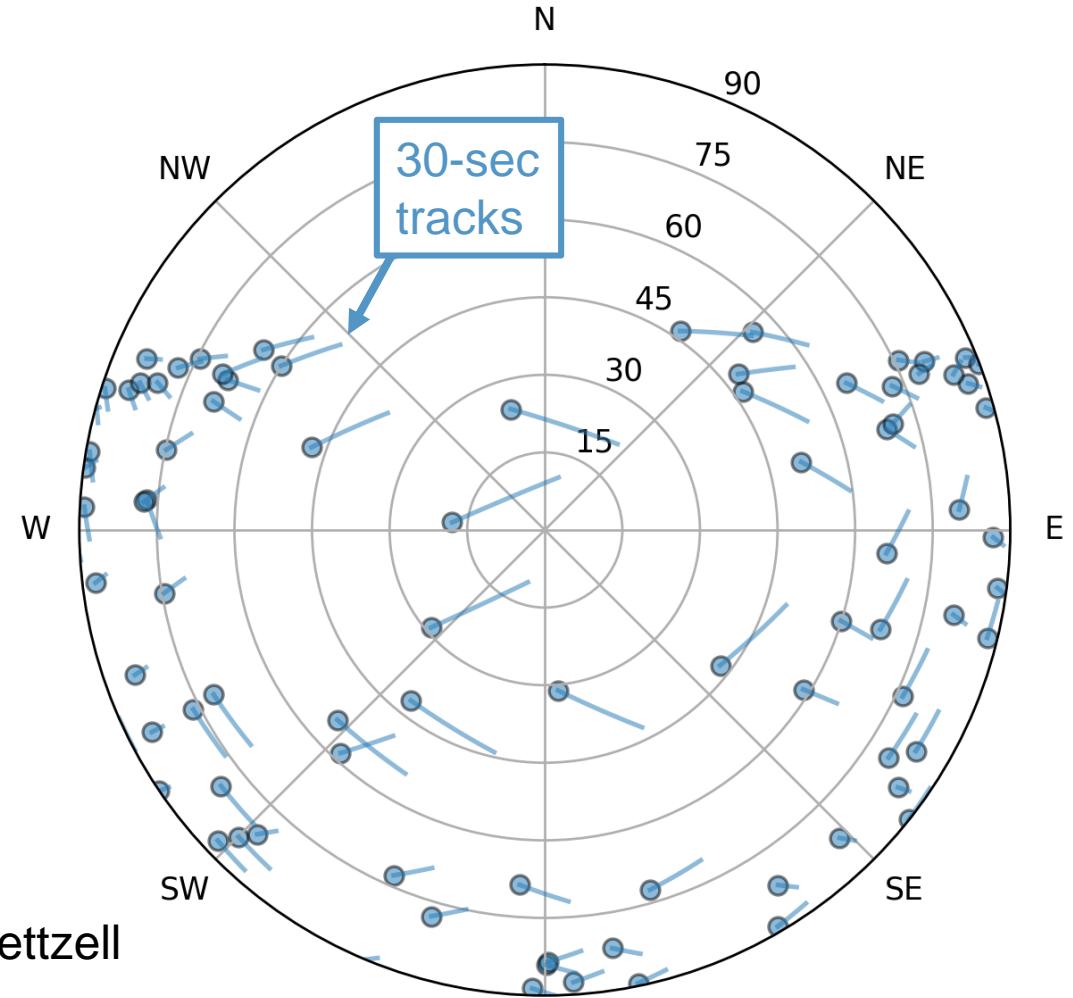
*currently, there are around 3360 Starlink satellites in orbit

	P.1	Phase 2					Phase 3							
orbits	72	72	36	6	4	7178	7178	7178	40	1998	4000	12	18	
sat/orbit	22	22	20	58	43	1	1	1	50	1	1	12	18	
satellites	1584	2824 (\rightarrow 4408 total)					30000 (\rightarrow 34408 total)							
altitude [km]	550	540	570	560	560	328	334	345	360	373	499	604	614	
inclination [°]	53	53.2	70	97.6	97.6	30	40	53	96.9	75	53	148	115.7	

Phase 1



1584 satellites

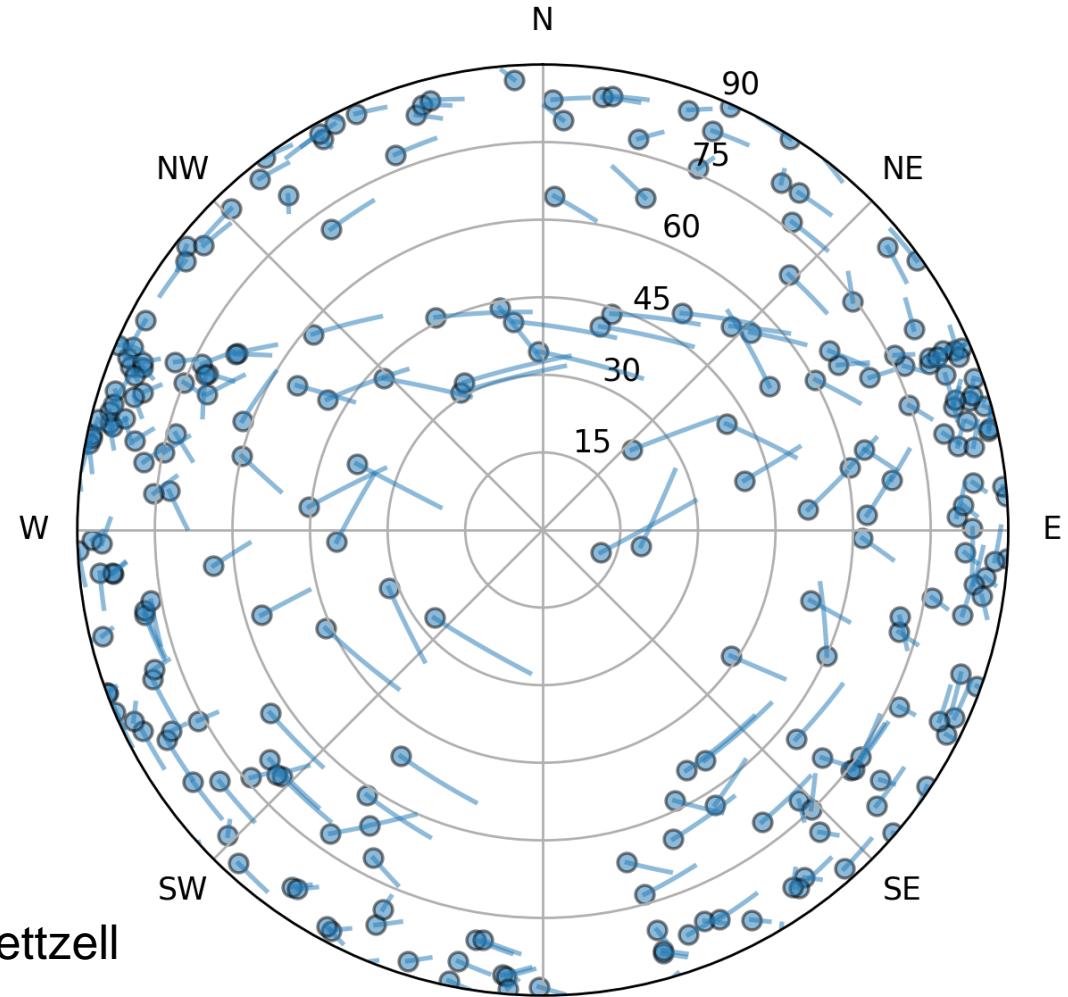


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Phase 2



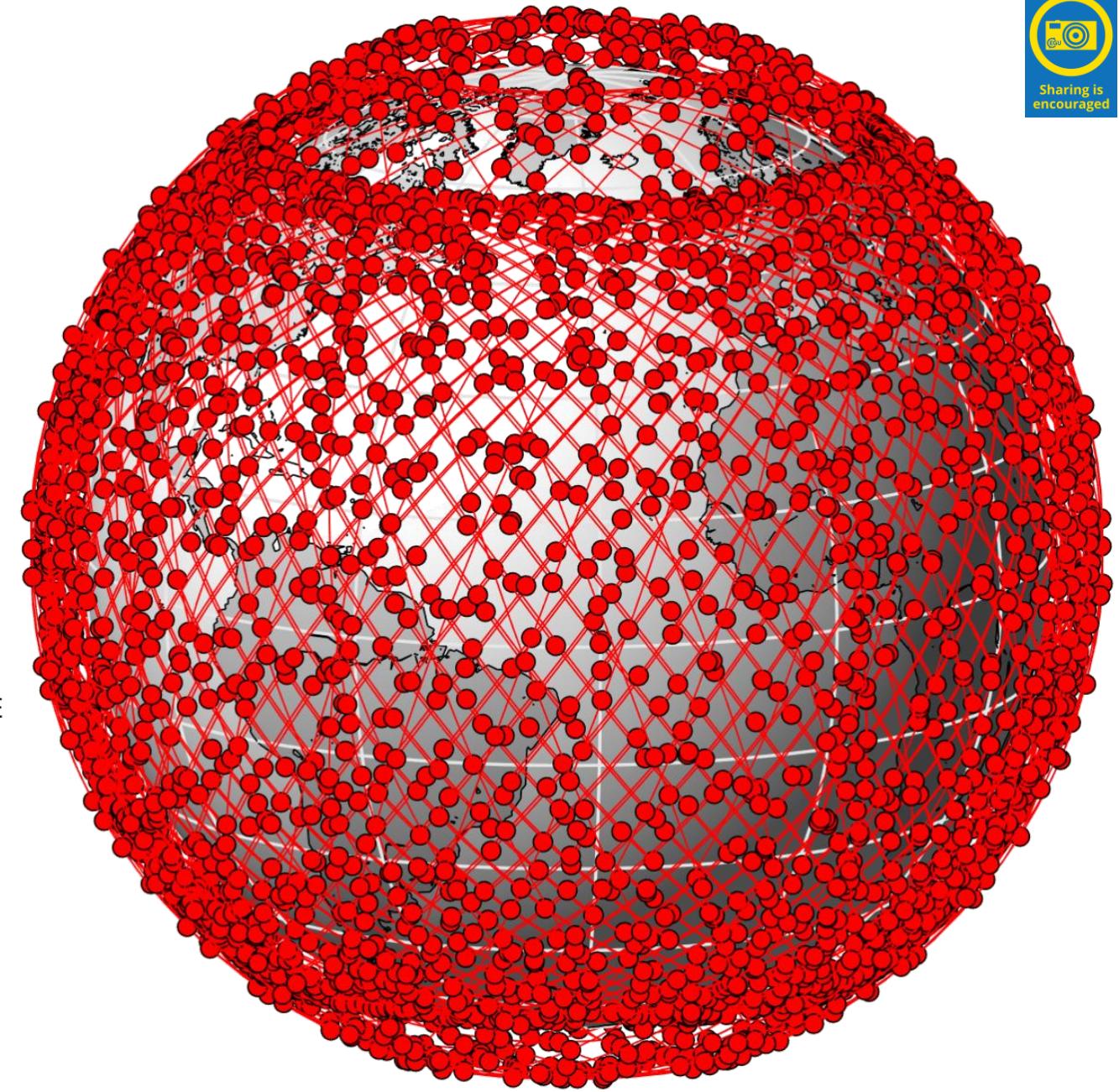
4408 satellites



Wettzell



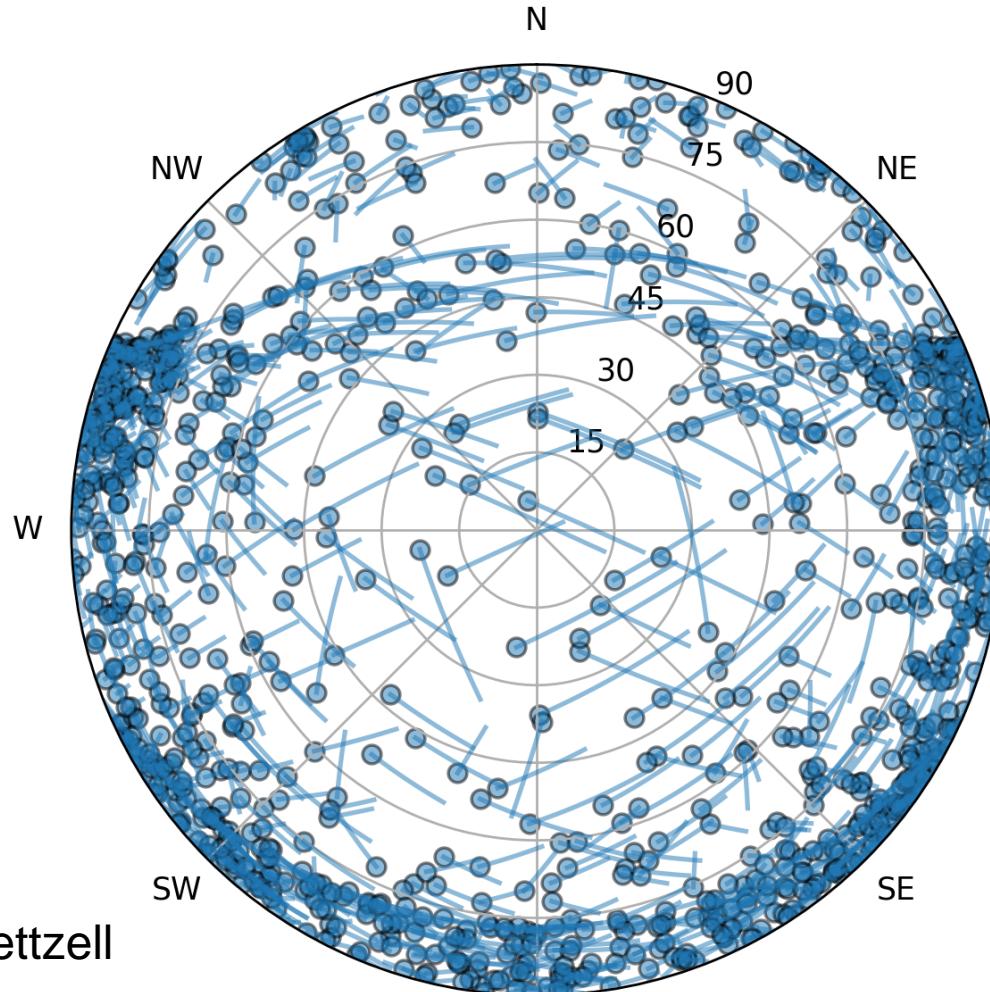
ETH zürich



Phase 3



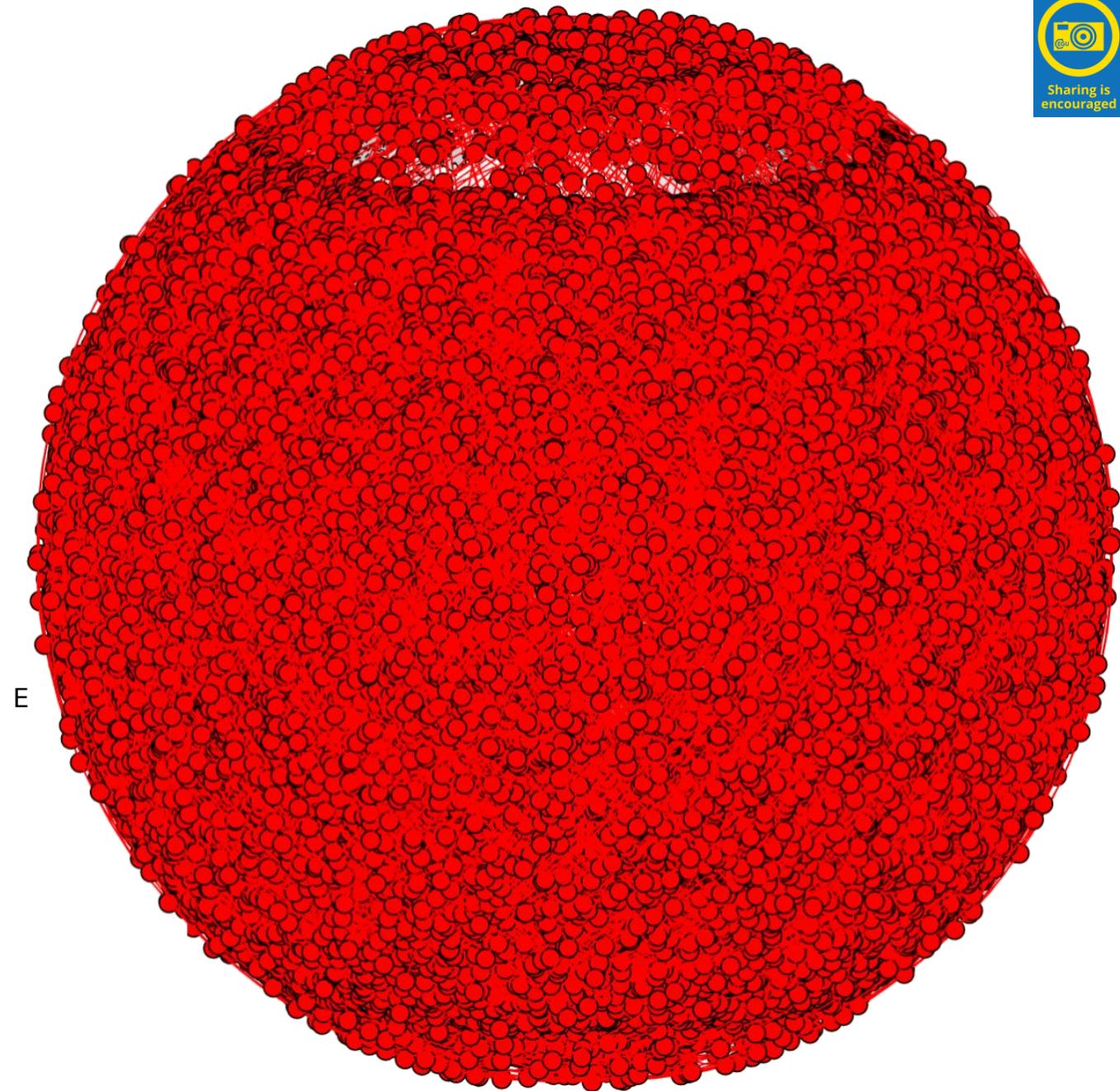
34408 satellites



Wettzell



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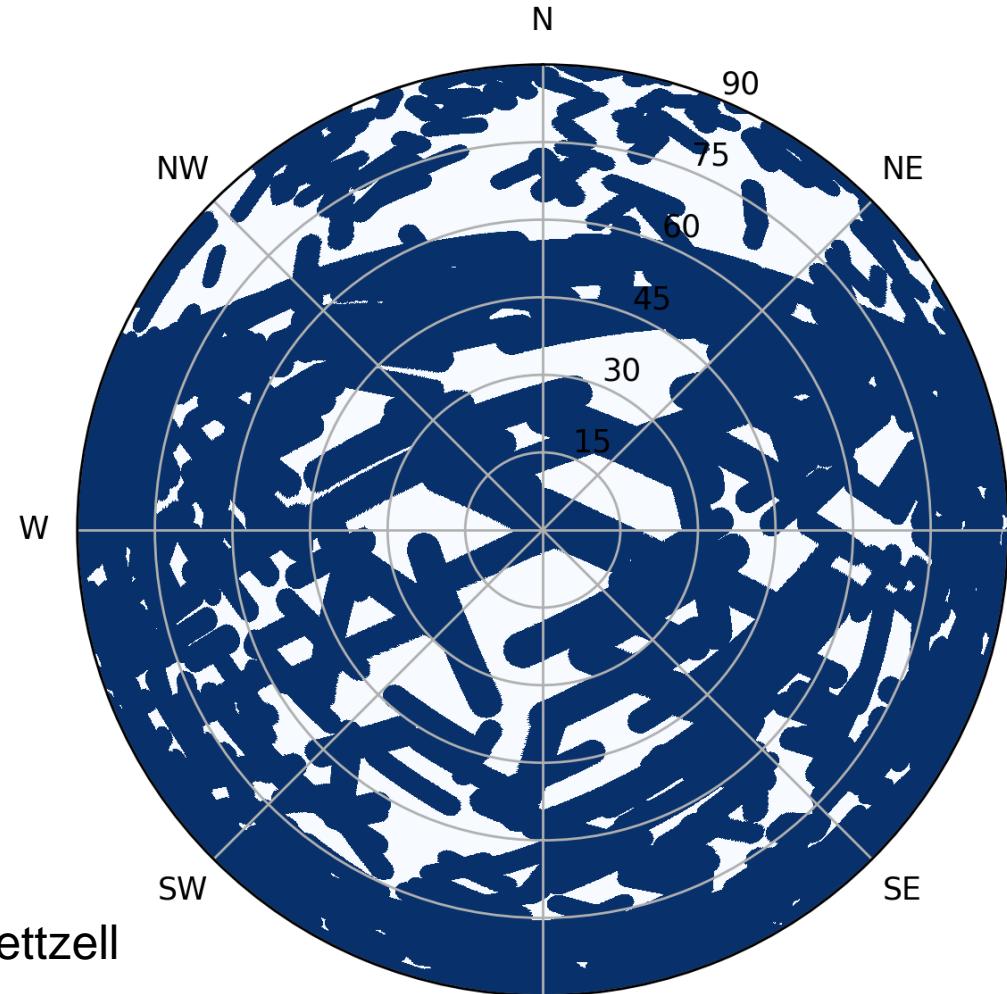


Phase 3

Avoidance mask for 30 sec scan



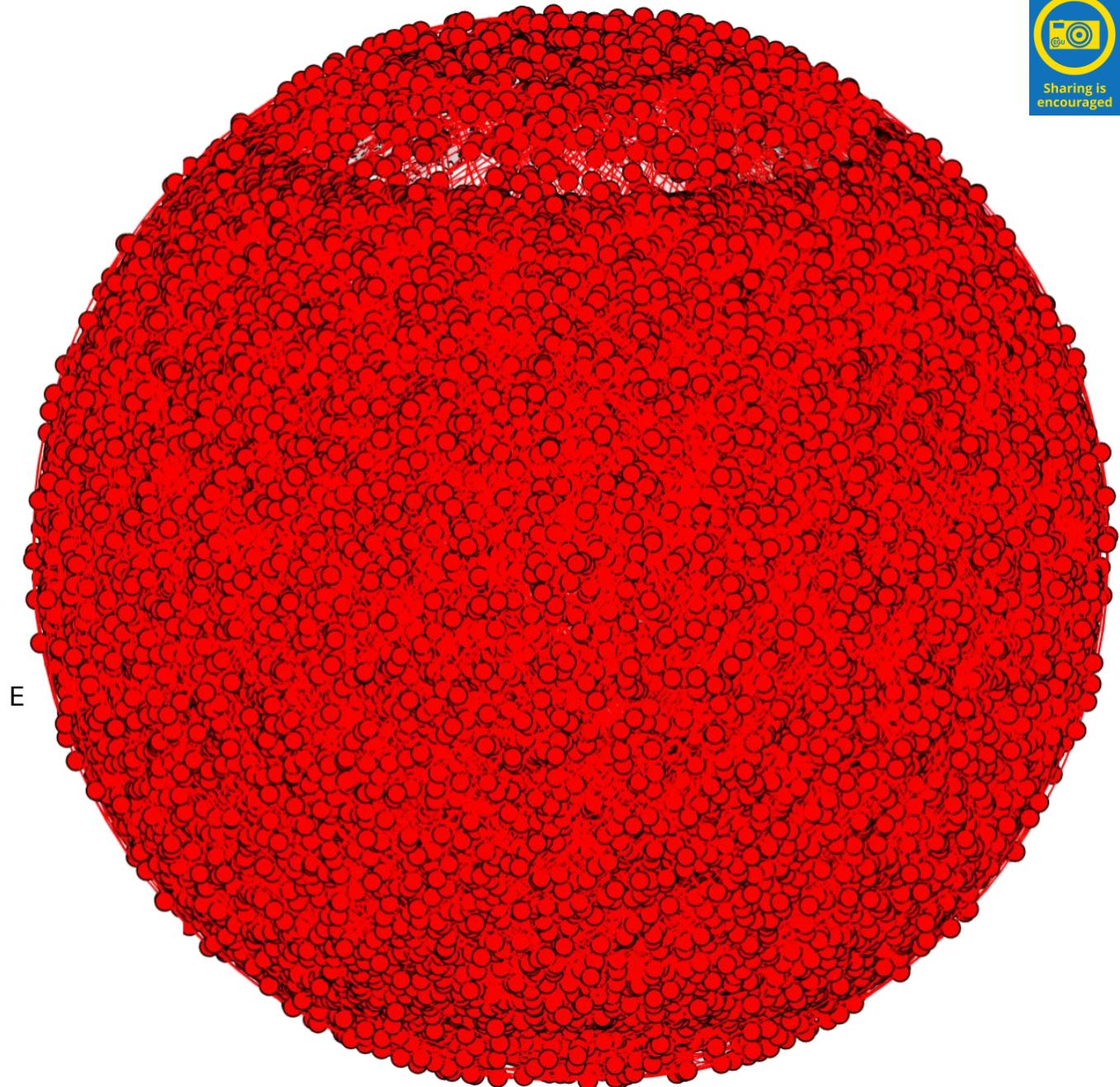
34408 satellites



Wettzell



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Impact on VLBI observations?

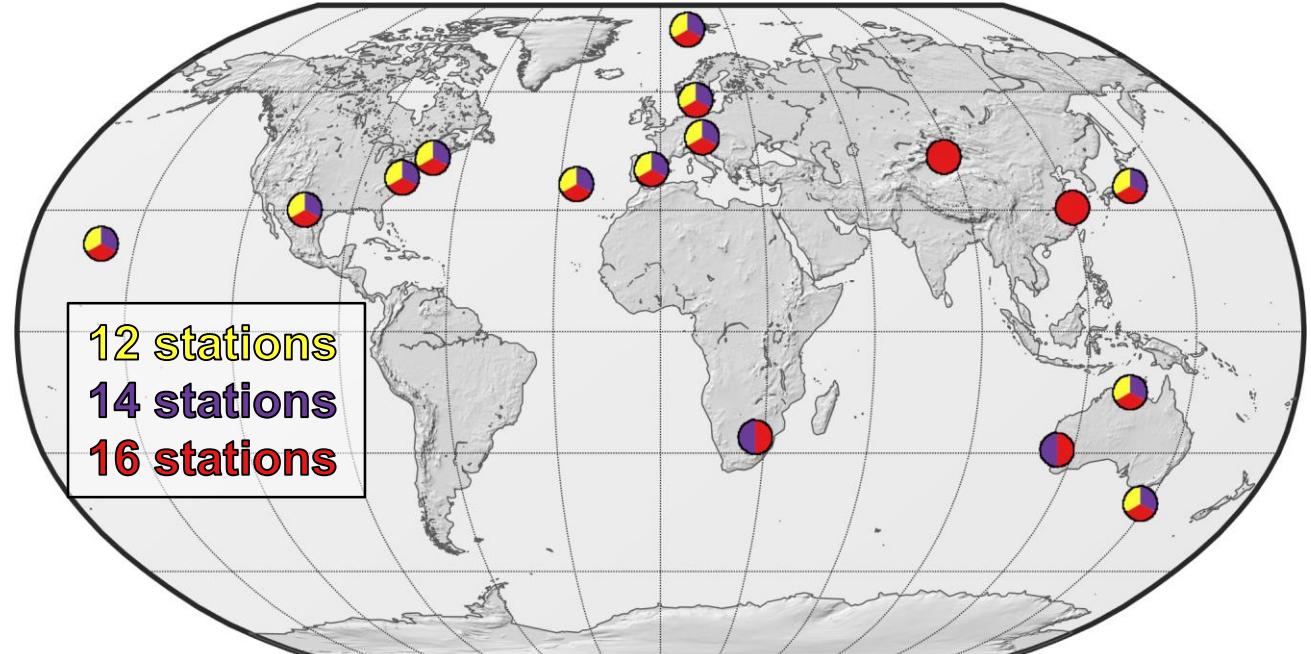
- How many observations are affected?
- Effect on geodetic products?

Method



- Generate VLBI observing plans
 - **no satellites: reference** without satellites
 - **active avoidance**: do **not schedule** sources close to satellites
 - **no active avoidance**: no active avoidance but **removal of affected observations**
- Monte-Carlo simulation

Repeated for three VLBI networks
and for three Starlink network phases
→ **nine combinations**



How close is too close?



Two terms: constant offset + error in orbits

$$\epsilon = c + f(e_0, e_i, d)$$

c = constant factor [deg]

e_0 = orbital error at epoch 0 [km]

e_i = increase of orbital error per day [km]

d = days till session start

a) $c = 1$

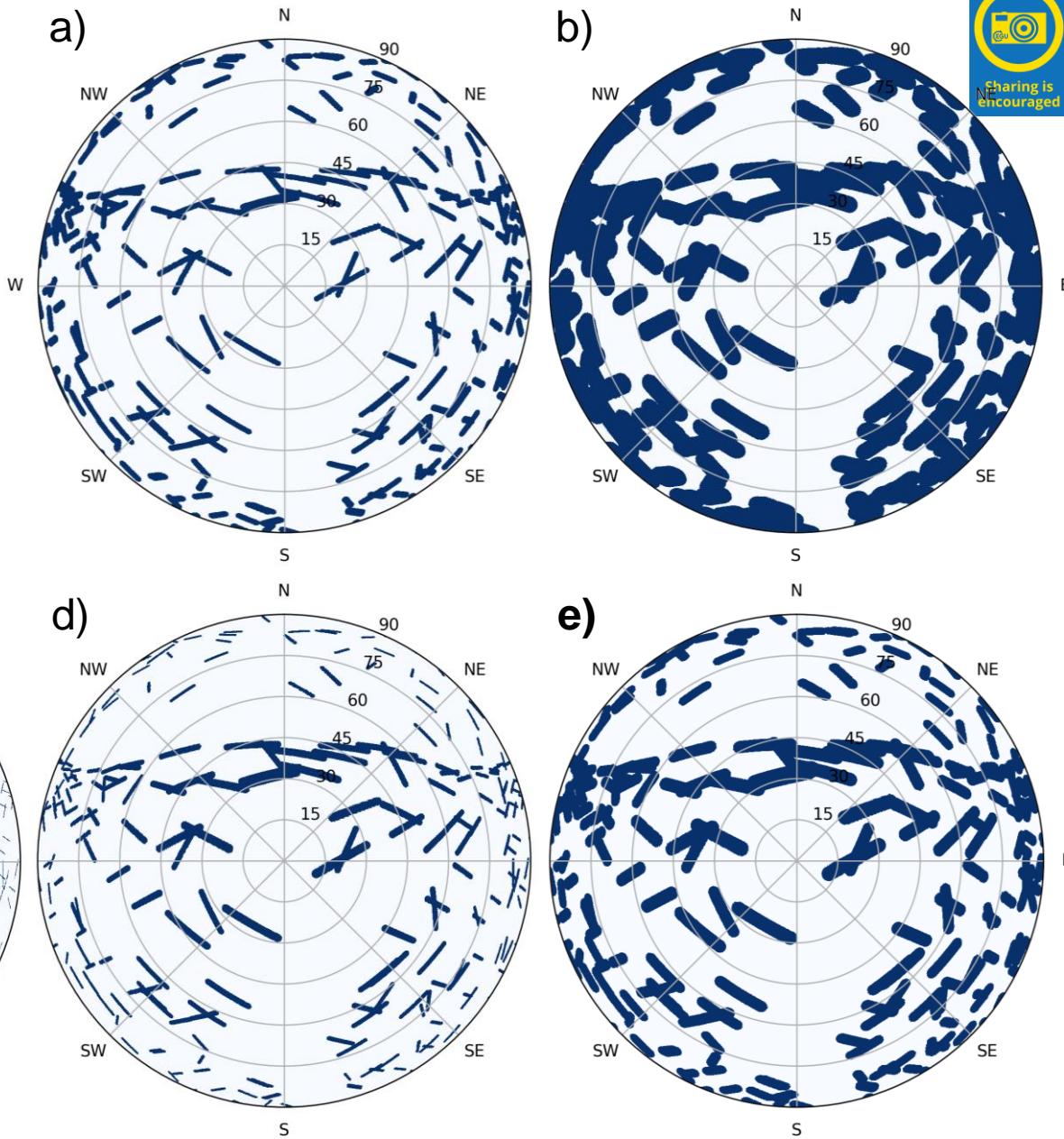
b) $c = 3$

c) $e_0 = e_i = 2; d = 1$

d) $e_0 = e_i = 2; d = 7$

e) $c = 1; e_0 = e_i = 2; d = 7$

*with min elevation of 20°



Example results*

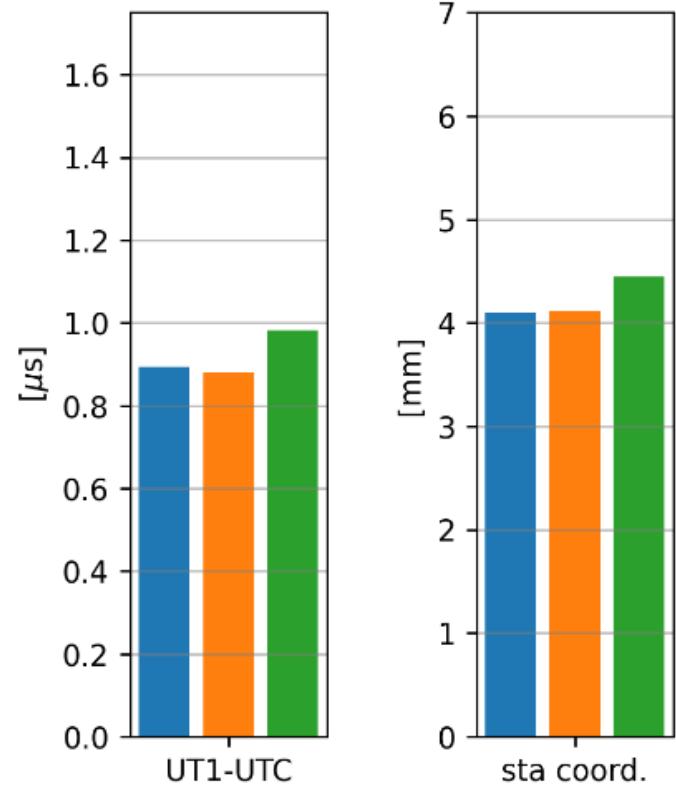
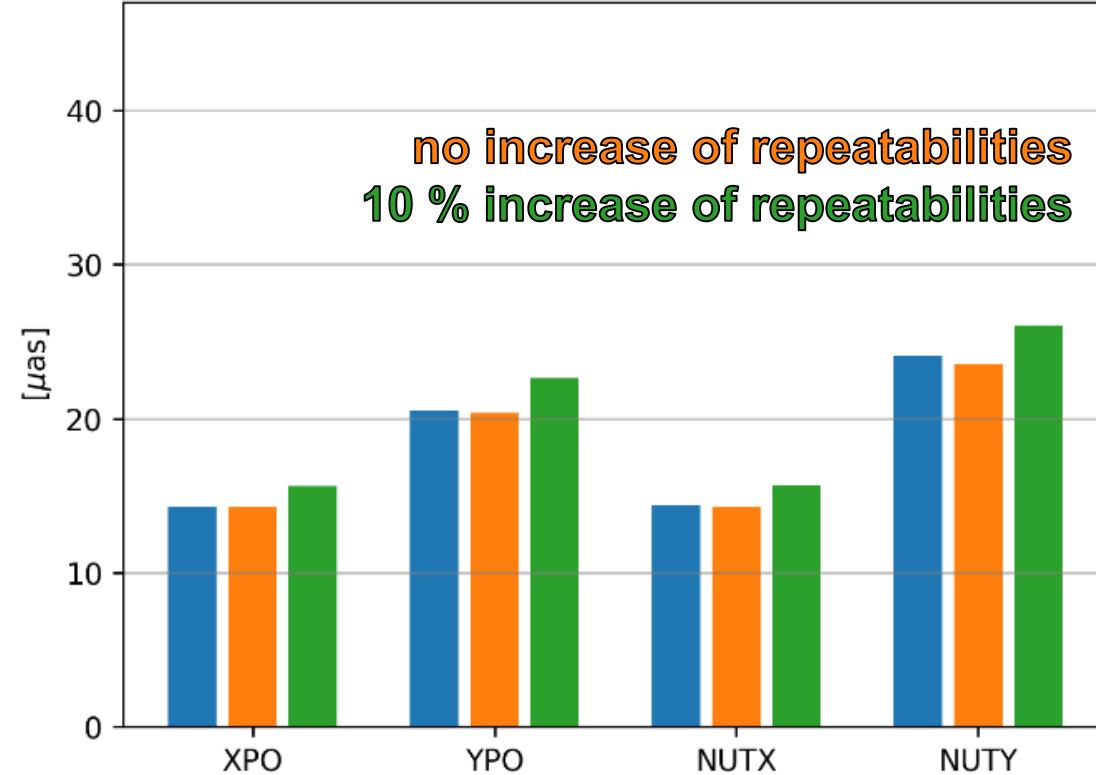
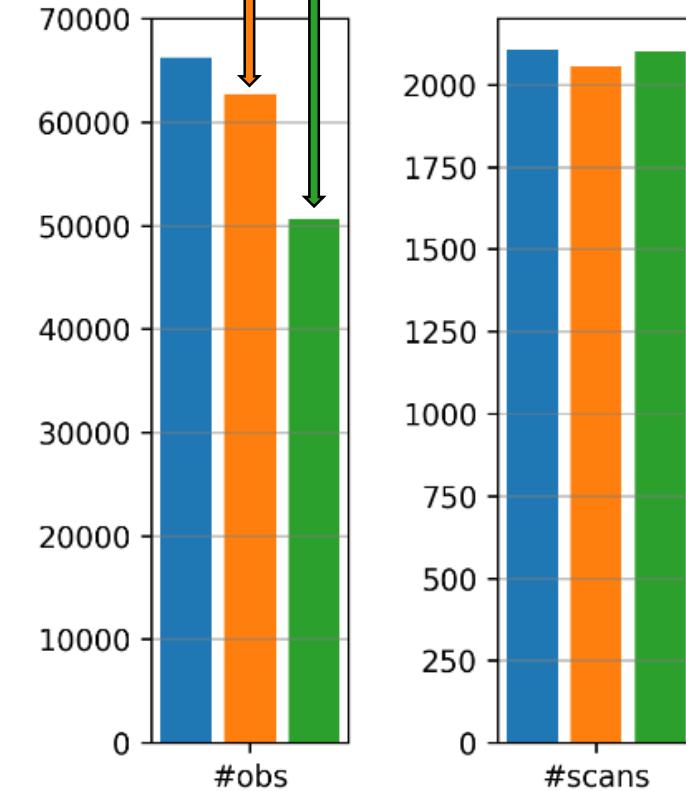


16 station VLBI network with Starlink phase 2 (4408 satellites)

no satellites
active avoidance
no active avoidance

5 % fewer observations

25 % observations are potentially affected



*all results are in the appendix



Increase of repeatability based on Monte Carlo simulations



		active avoidance				no active avoidance			
		PM	NUT	UT1	STA	PM	NUT	UT1	STA
1584 sat	12 sta. phase 1	0 %	0 %	-1 %	0 %	3 %	3 %	3 %	3 %
	14 sta. phase 1	0 %	0 %	0 %	0 %	3 %	3 %	4 %	3 %
	16 sta. phase 1	-1 %	-1 %	-1 %	0 %	4 %	3 %	4 %	3 %
4408 sat	12 sta. phase 2	0 %	1 %	0 %	1 %	9 %	9 %	10 %	8 %
	14 sta. phase 2	-1 %	0 %	-1 %	1 %	8 %	9 %	9 %	8 %
	16 sta. phase 2	-2 %	-1 %	-2 %	0 %	9 %	9 %	10 %	9 %
34408 sat	12 sta. phase 3	6 %	13 %	14 %	16 %	75 %	65 %	80 %	62 %
	14 sta. phase 3	13 %	23 %	18 %	18 %	76 %	90 %	80 %	63 %
	16 sta. phase 3	17 %	29 %	21 %	19 %	78 %	93 %	89 %	63 %

Increase of repeatability based on Monte Carlo simulations



		active avoidance				no active avoidance			
		PM	NUT	UT1	STA	PM	NUT	UT1	STA
1584 sat	12 sta. phase 1								
	14 sta. phase 1								
	16 sta. phase 1								
4408 sat	12 sta. phase 2								
	14 sta. phase 2								
	16 sta. phase 2								
34408 sat	12 sta. phase 3								
	14 sta. phase 3								
	16 sta. phase 3								

The table displays the percentage increase of repeatability for different station counts (12, 14, 16) across three phases (phase 1, phase 2, phase 3) for two scenarios: "active avoidance" (orange background) and "no active avoidance" (green background). The values are as follows:

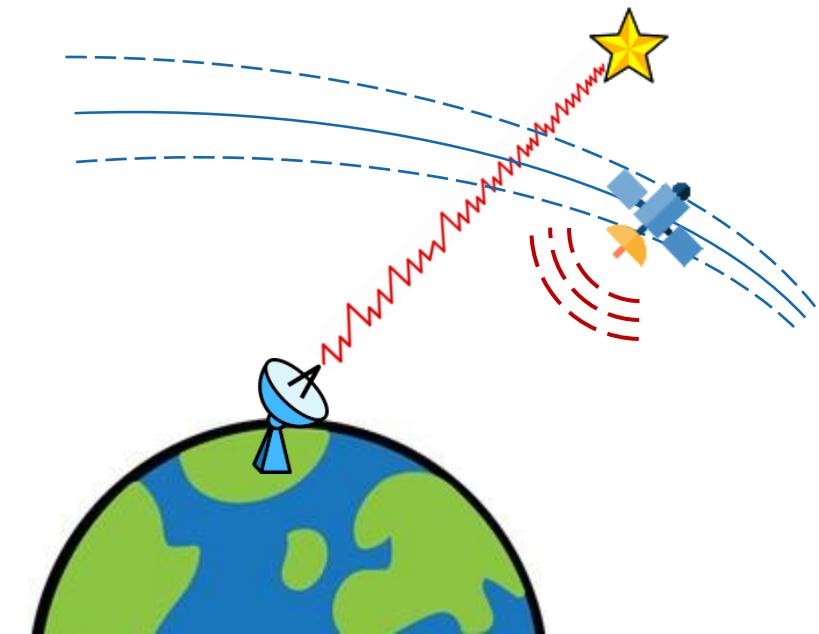
Phase	Stations	Active Avoidance (%)	No Active Avoidance (%)
1584 sat	12	0%	3%
	14	0%	9%
	16	0%	20%
4408 sat	12	0%	9%
	14	0%	20%
	16	0%	80%
34408 sat	12	0%	20%
	14	0%	80%
	16	20%	80%

Current bottlenecks

- **Implementation**
 - track thousands of satellites (every second)
 - computationally expensive
- Scheduling is an optimization problem
 - relies to some extend on brute-force testing
 - performance is critical
- **Affected area / safety margin**
 - more study needed to properly measure and quantify requirements (under way)
- Satellite orbit data and predictions
 - tens of thousands of satellites
 - launched database of TLE files at ETH
 - <https://satdb.ethz.ch/>

```
void Scan::checkSatelliteDistance( Station &sta, const shared_ptr<const AbstractSource> &source, ofstream &of ) {
    boost::optional<unsigned long> oidx = findIdxOfStationId( sta.getId() );
    int idx = oidx.get();
    unsigned int startTime = getTimes().getObservingTime();
    vector<unsigned int> times;
    unsigned int t = getTimes().getObservingTime( idx );
    while ( t < getTimes().getObservingTime( idx, Timestamp::end ) ) {
        times.push_back( t );
        t += AvoidSatellites::frequency;
    }
    times.push_back( getTimes().getObservingTime( idx, Timestamp::end ) );

    for ( auto tt : times ) {
        pair<double, double> rade_src = source->getRaDe( tt, sta.getPosition() );
        for ( const auto &sat : AvoidSatellites::satellitesToAvoid ) {
            auto rade_sat = sat->calcRaDeDistTime( tt, sta.getPosition() );
        }
    }
}
```



Conclusion



- Number of satellites in space is rising dramatically
- VLBI must **prepare** for potential threats from space borne RFI
- Presented in this work: **active mitigation strategy**
 - avoid observations close to satellites
 - passive strategies (e.g. filters) also exist
- **Few thousand** satellites can be **handled effectively** by this approach
 - no noticeable effect on the geodetic products based on simulations
- **Orbit prediction accuracy** drives safety margin size
 - Currently based on TLE files
 - problematic if schedules need to be posted 7 days in advance
- **More work is required**

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Space Geodesy

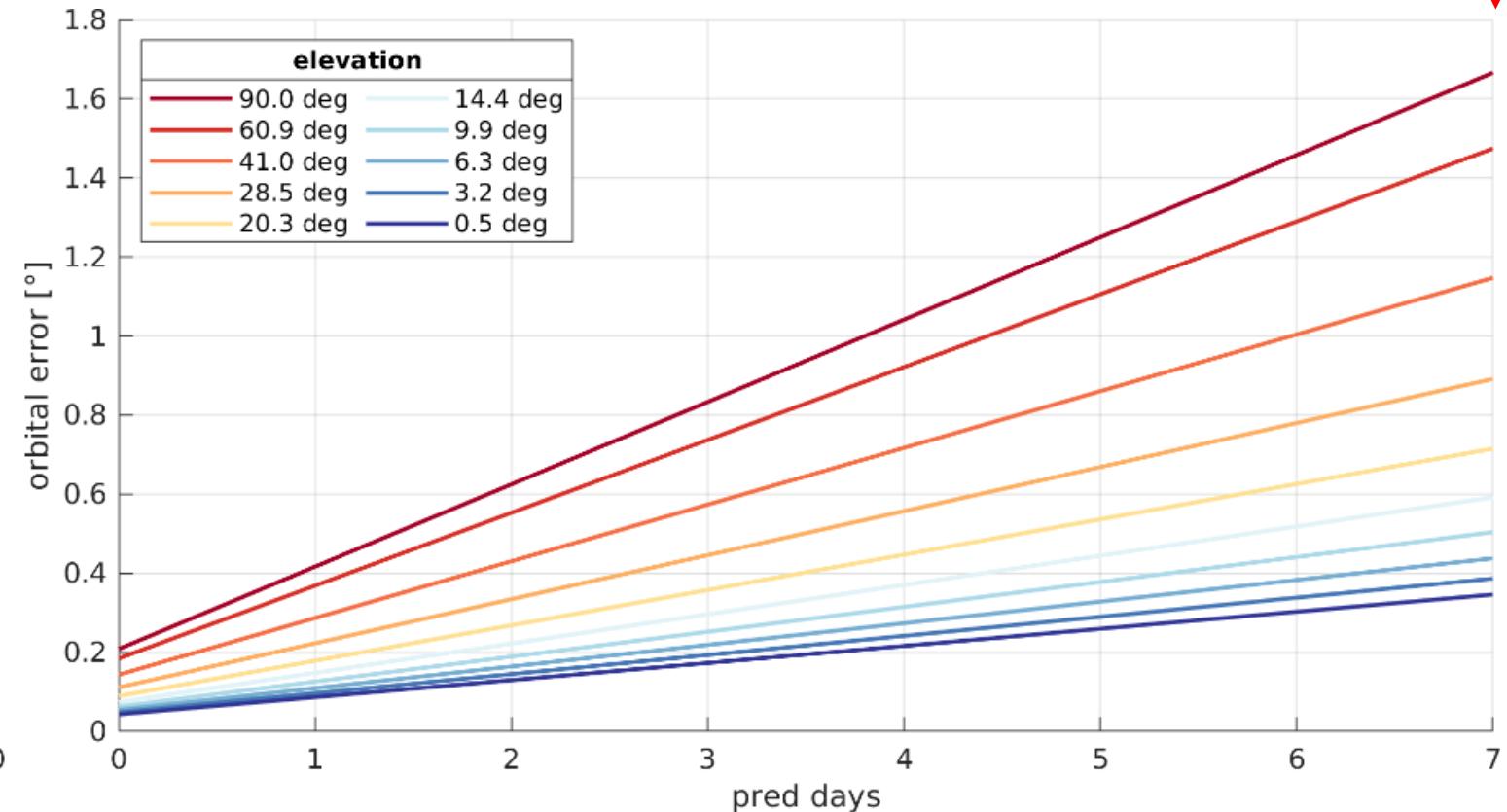
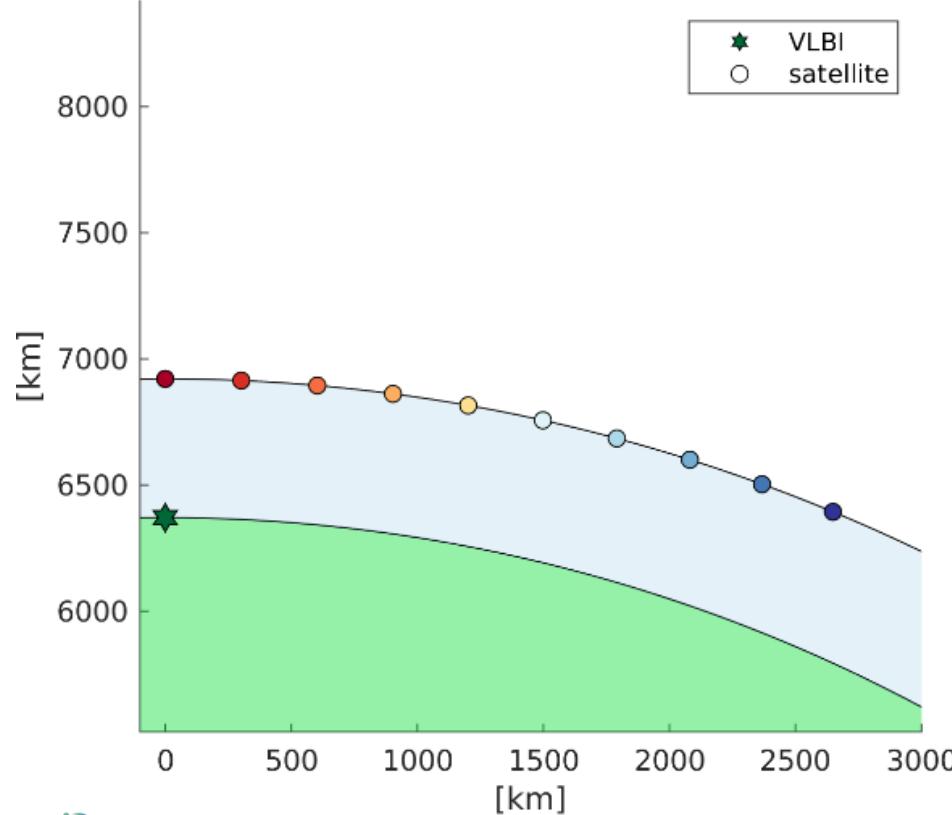
HPV G 53
Robert-Gnehm-Weg 15
8093 Zürich

When are observations affected? Safety margin for active avoidance?

Big part is orbit prediction accuracy

assumption: 500 km altitude, error at epoch 0 is 2 km, increased by 2 km per day

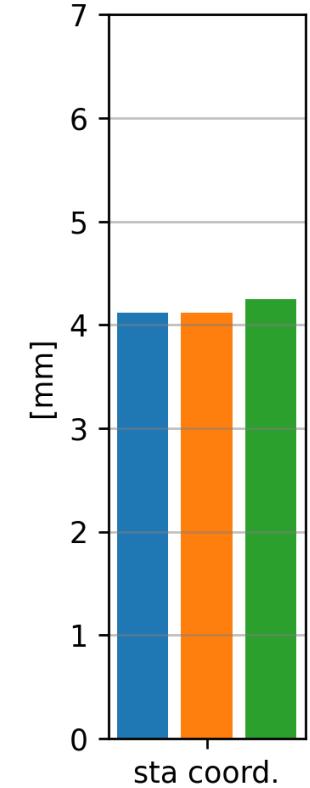
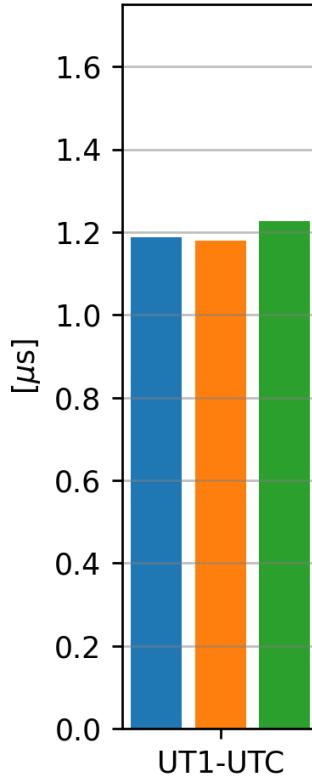
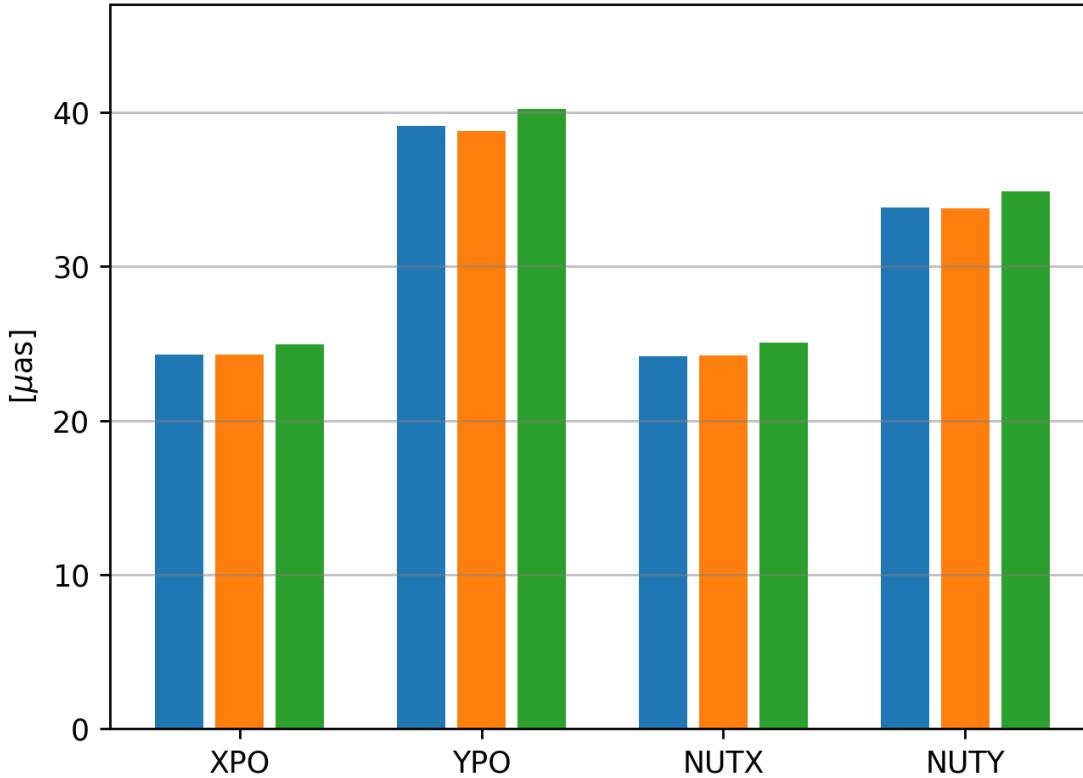
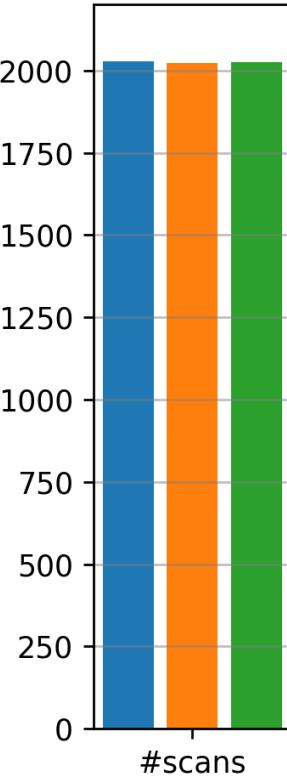
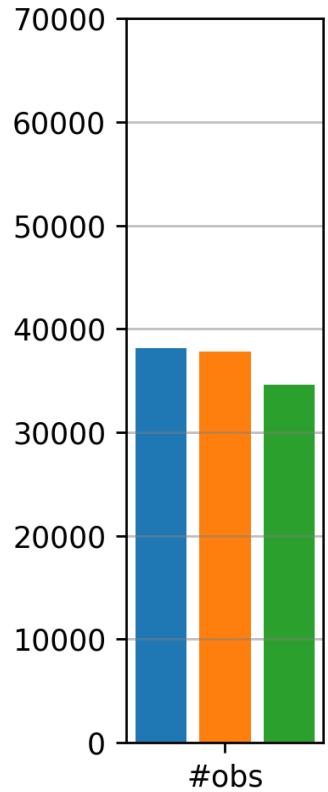
IVS requirement



12 station VLBI network + Starlink phase 1 (12 sta. phase 1)



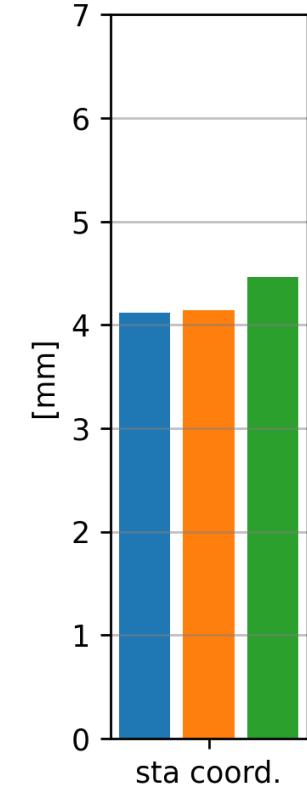
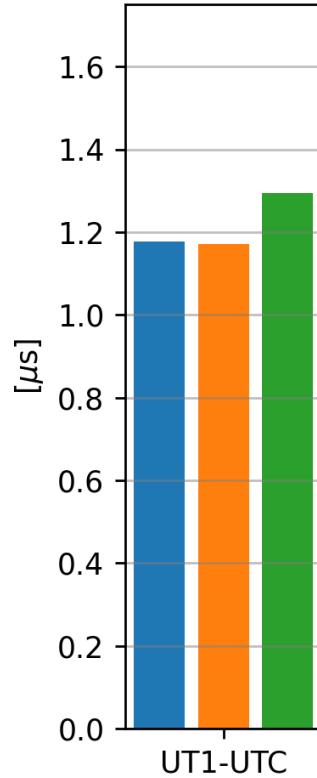
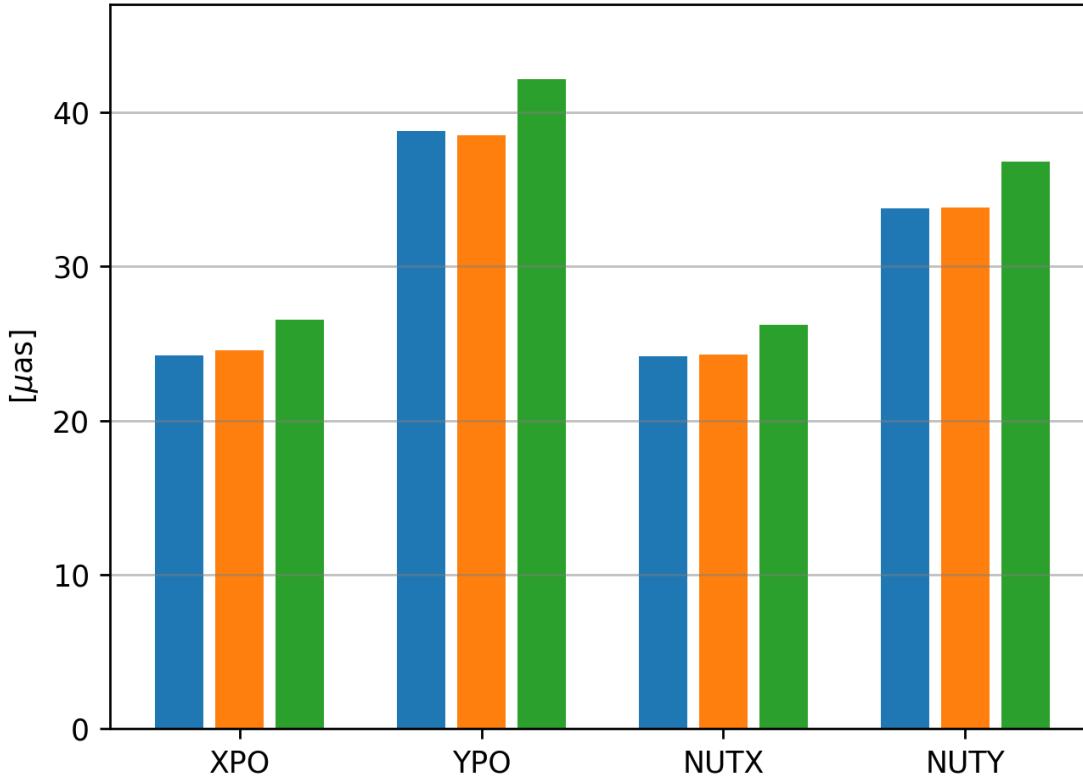
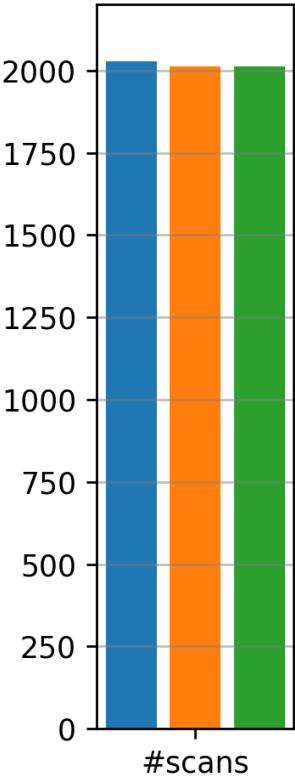
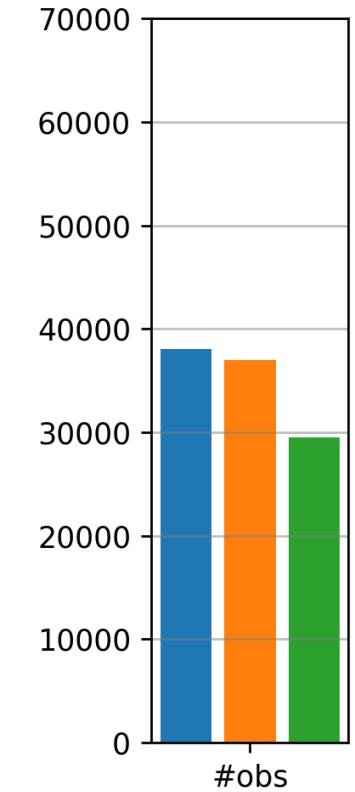
no satellites
active avoidance
no active avoidance



12 station VLBI network + Starlink phase 2 (12 sta. phase 2)



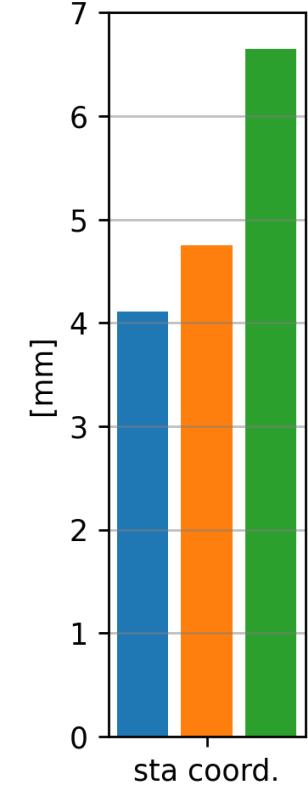
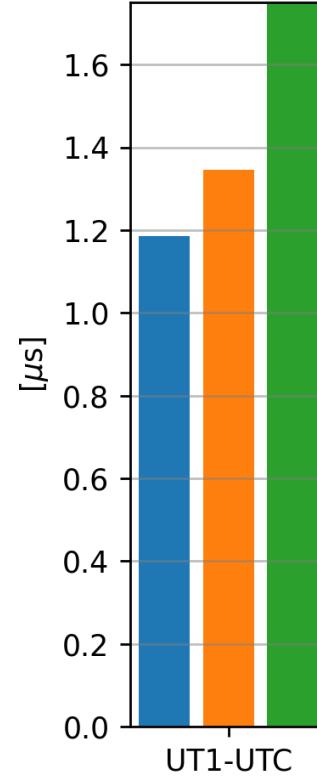
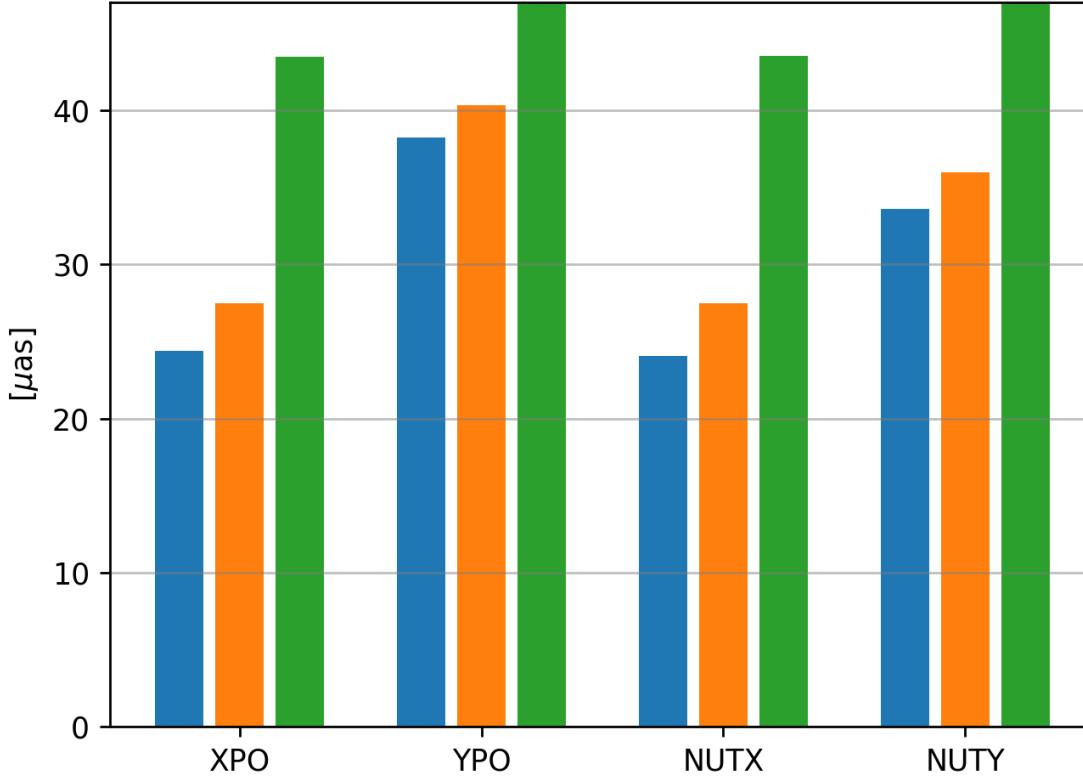
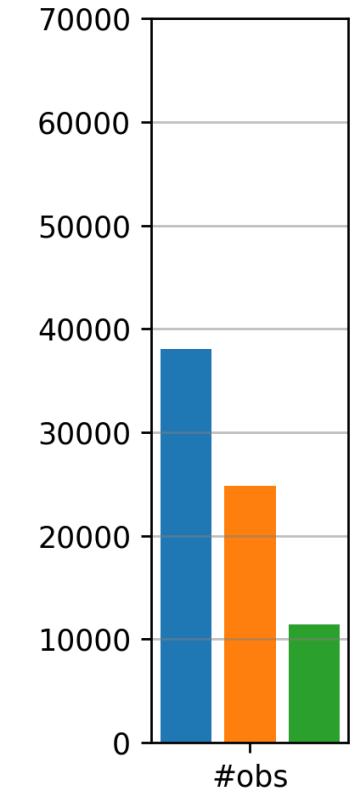
no satellites
active avoidance
no active avoidance



12 station VLBI network + Starlink phase 3 (12 sta. phase 3)



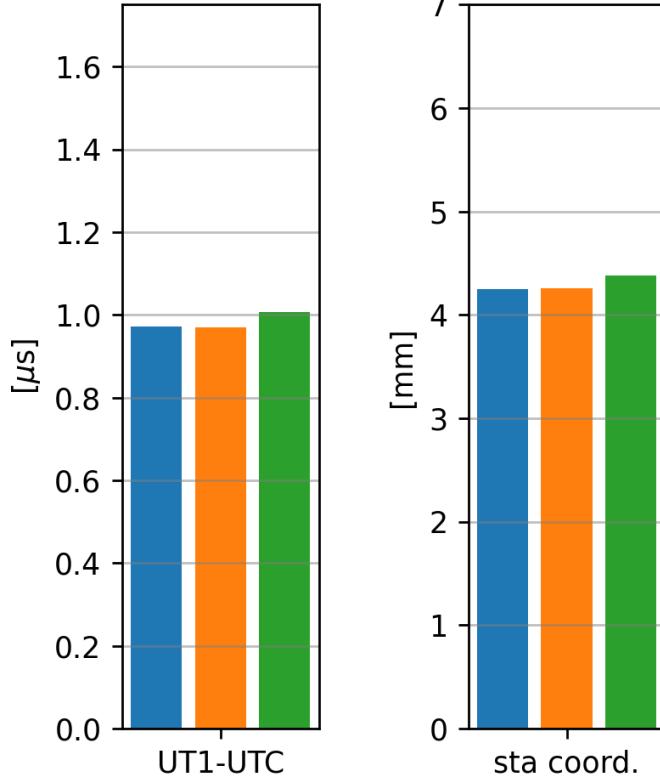
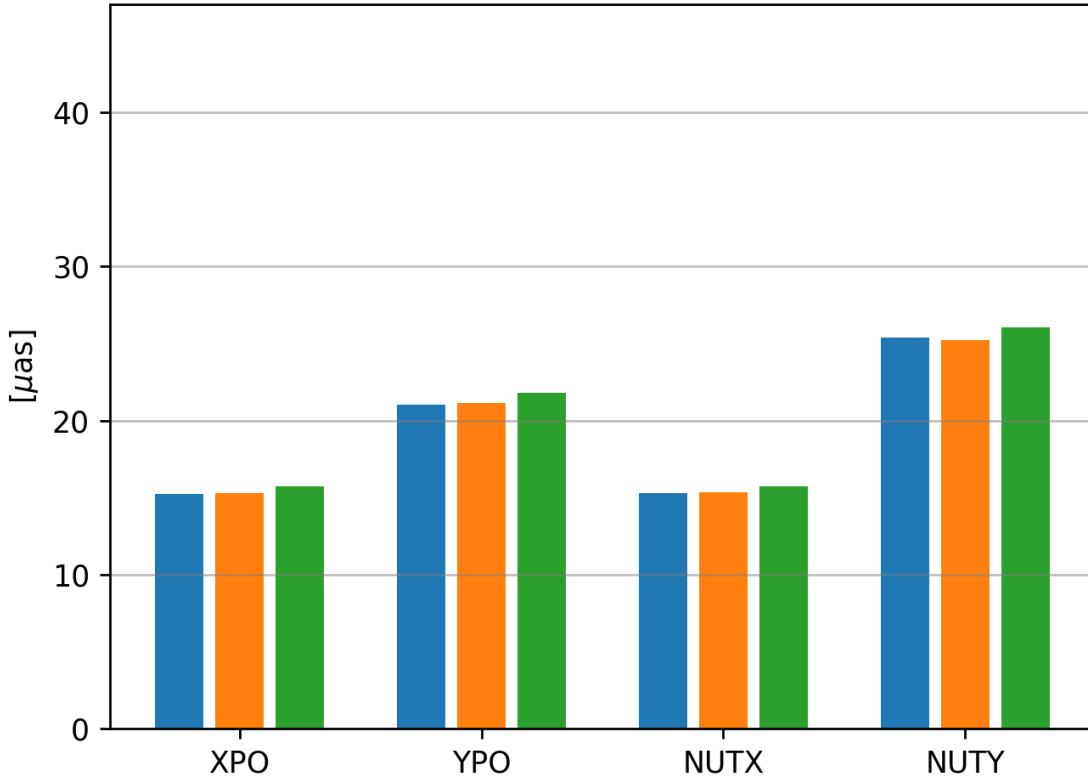
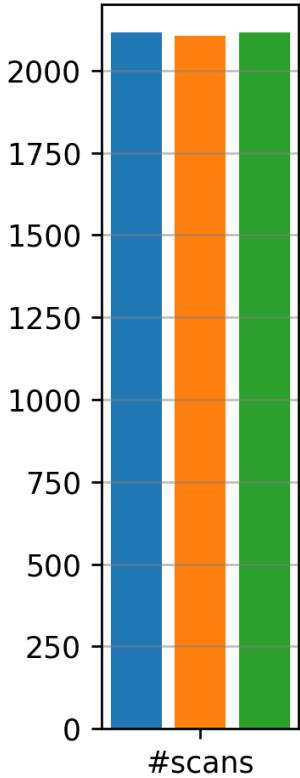
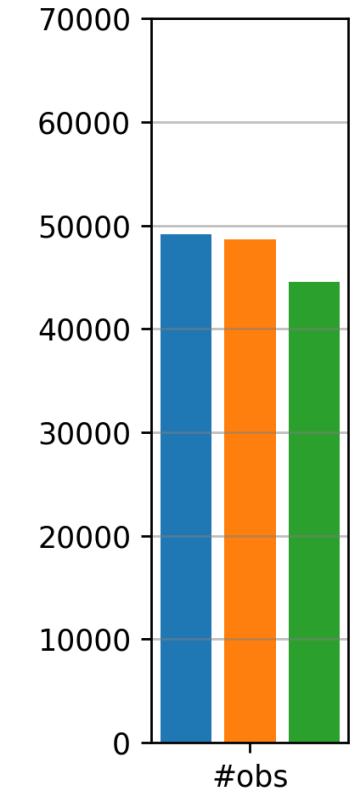
no satellites
active avoidance
no active avoidance



14 station VLBI network + Starlink phase 1 (14 sta. phase 1)



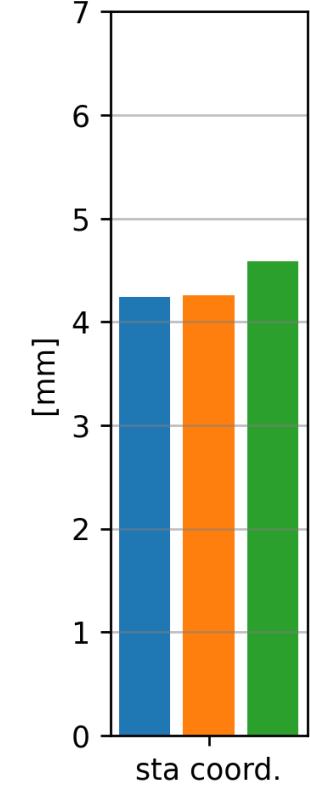
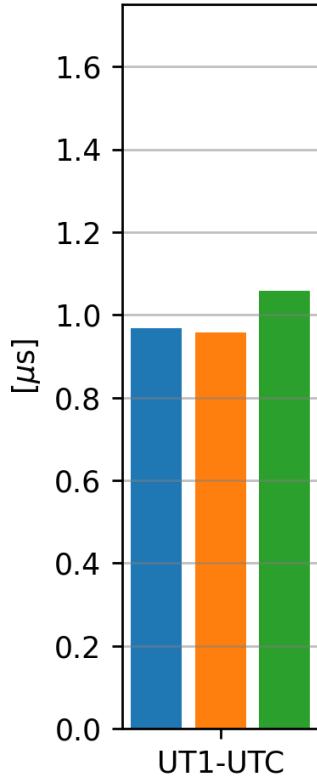
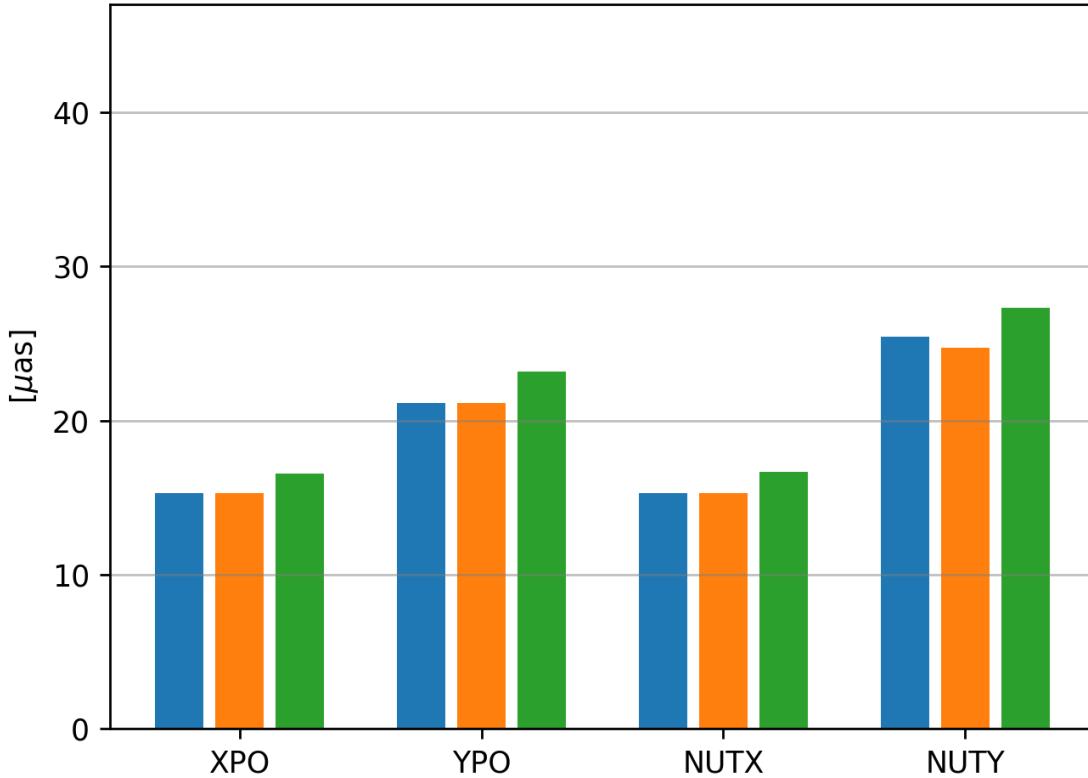
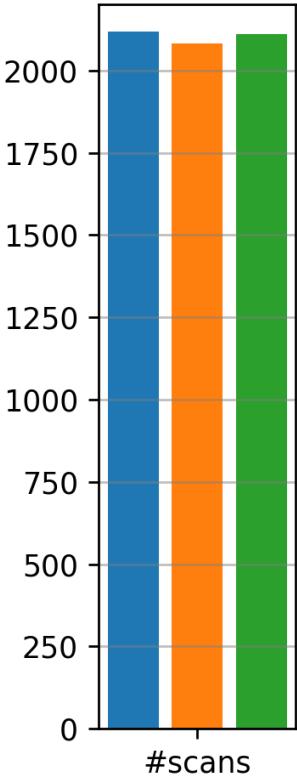
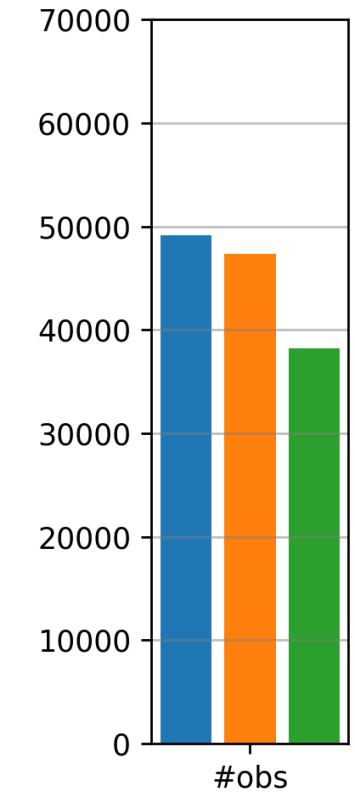
no satellites
active avoidance
no active avoidance



14 station VLBI network + Starlink phase 2 (14 sta. phase 2)



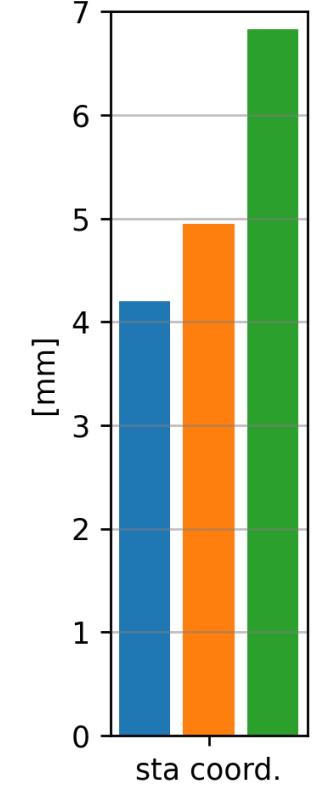
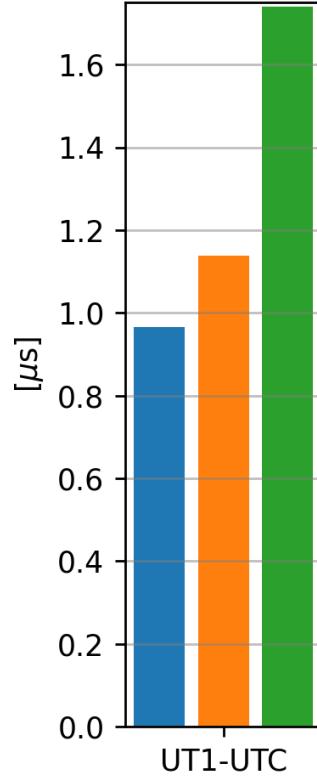
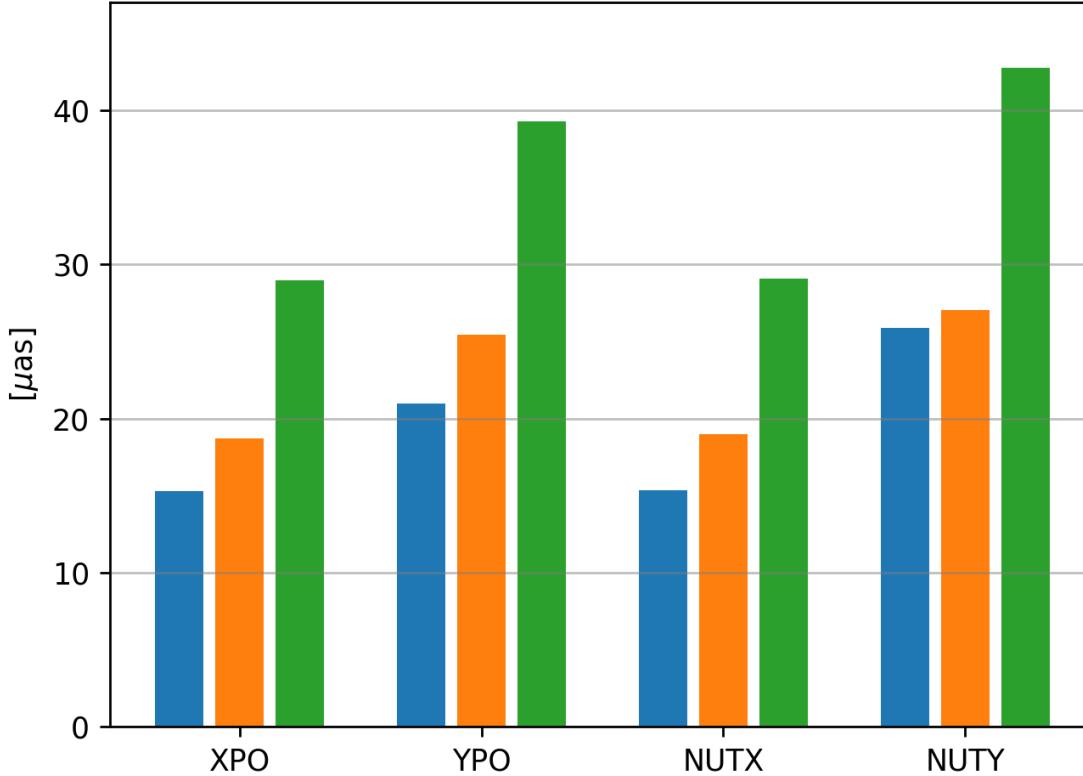
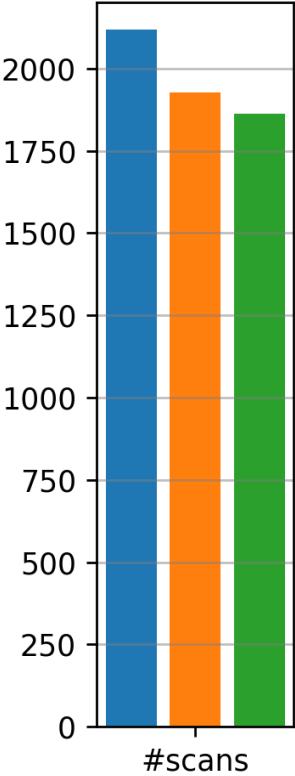
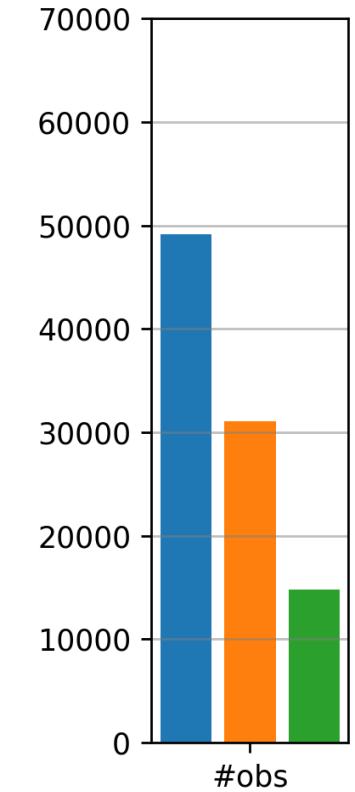
no satellites
active avoidance
no active avoidance



14 station VLBI network + Starlink phase 3 (14 sta. phase 3)



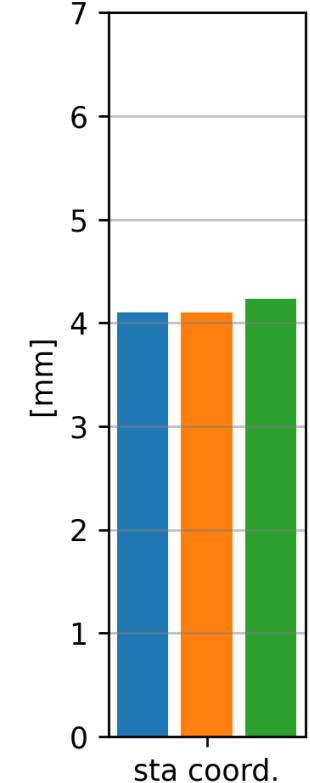
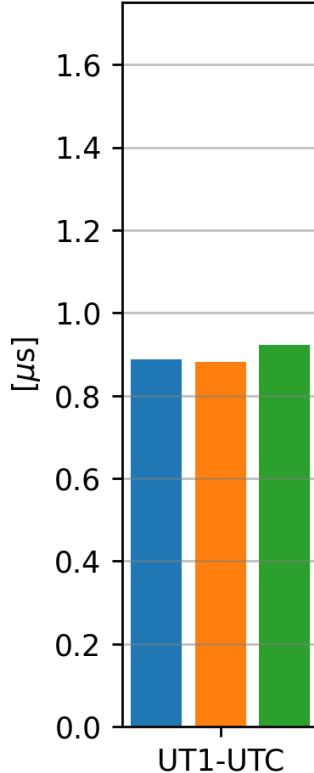
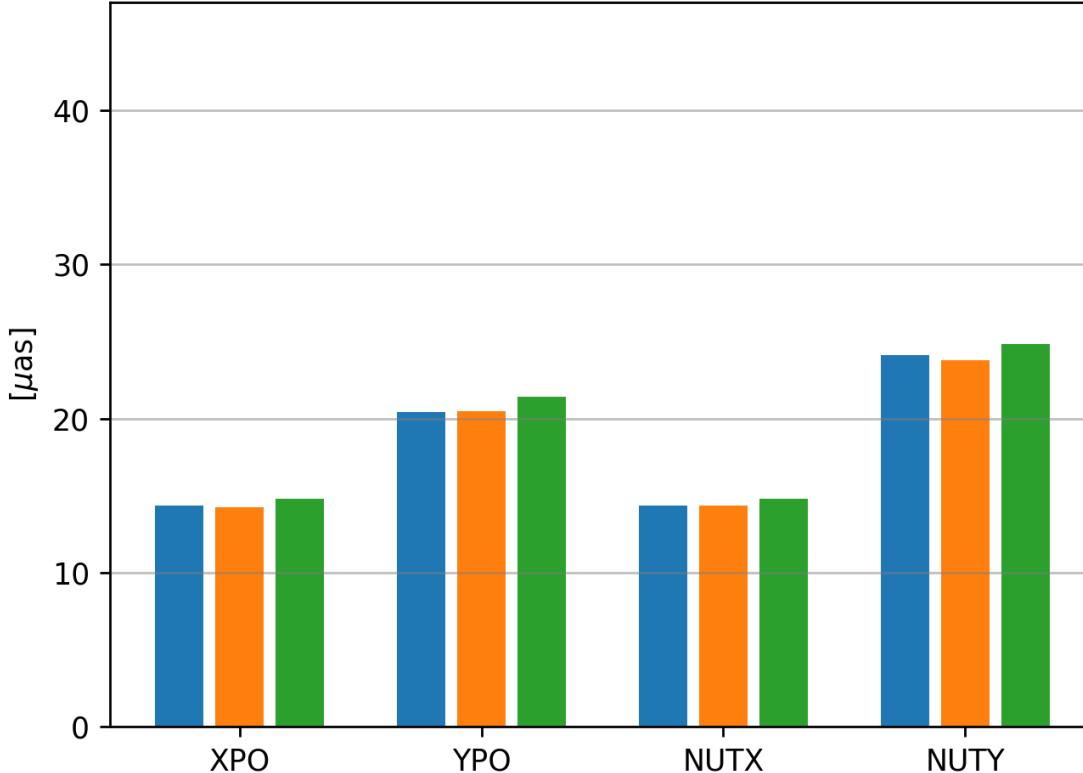
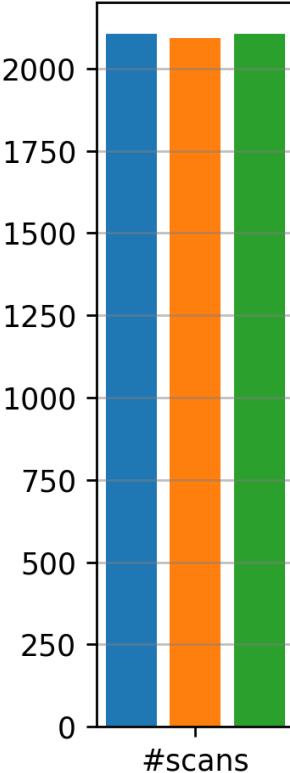
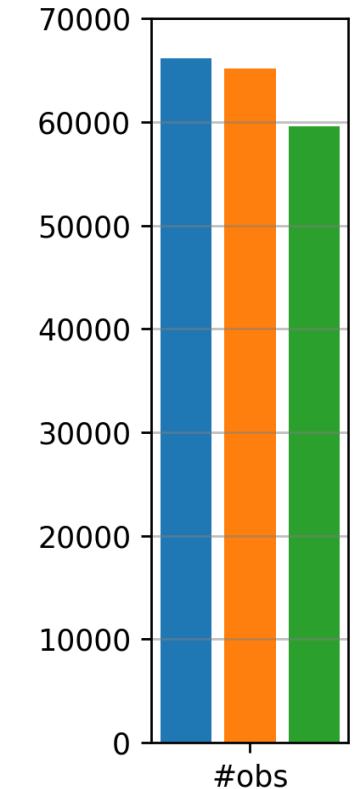
no satellites
active avoidance
no active avoidance



16 station VLBI network + Starlink phase 1 (16 sta. phase 1)



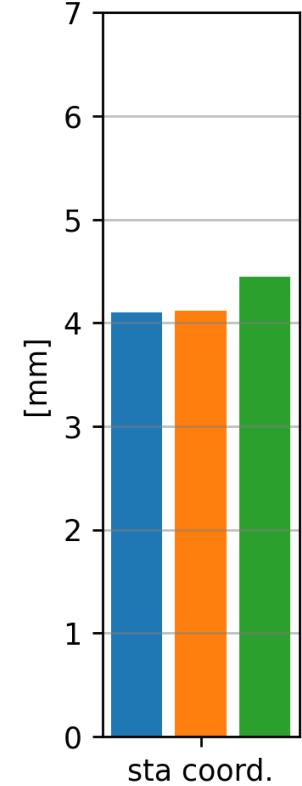
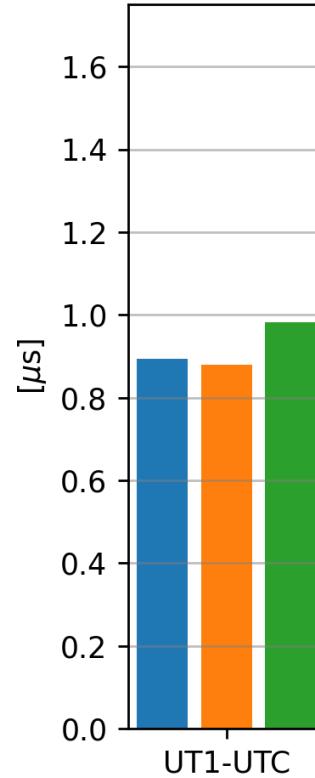
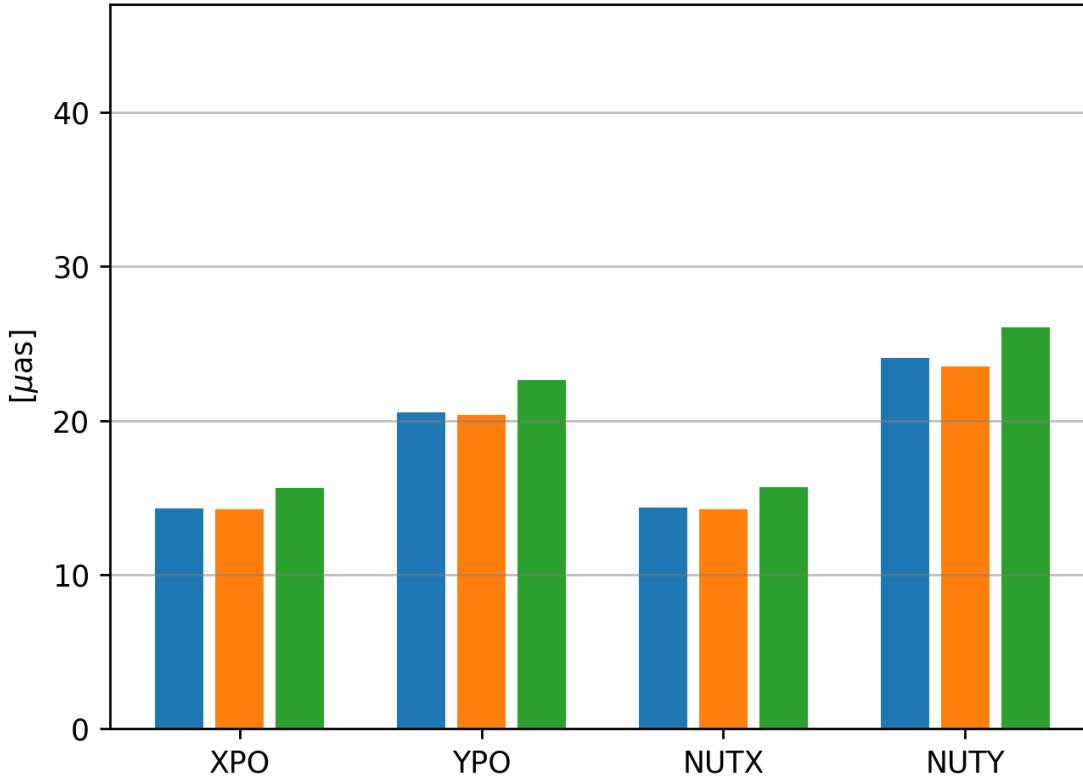
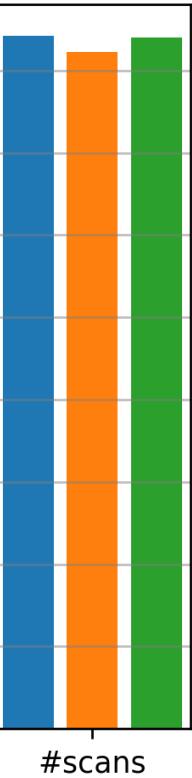
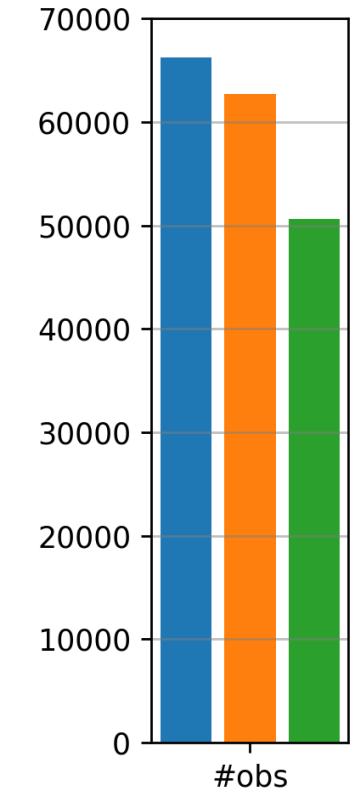
no satellites
active avoidance
no active avoidance



16 station VLBI network + Starlink phase 2 (16 sta. phase 2)



no satellites
active avoidance
no active avoidance



16 station VLBI network + Starlink phase 3 (16 sta. phase 3)



no satellites
active avoidance
no active avoidance

