

Contribution of SLR to satellite-only global gravity field model

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Combined satellite-only global gravity field models



Advantage: Compensates for the weaknesses of individual observation techniques.

Issue: The combination of solutions from different institutions can have an impact on the performance due to the use of different methods and algorithms.

Goal: Consistent computation of the next Gravity Observation Combination (GOCO) model.





Software toolkit GROOPS



Perform all evaluations with a uniform software package!





Software toolkit GROOPS









Gravity field coefficients estimation – Process flow in GROOPS



• Orbit integration using the given forces (State-of-the-art models, e.g. AOD1B-RL06, GOCO06s...) → Integral Equation appr.



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program

PreprocessingVariationalEquation

• Fit the integrated orbit to initial orbit positions (e.g. CPF provided by ILRS...)

PreprocessingVariationalEquationOrbitFit





Gravity field coefficients estimation – Process flow in GROOPS



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 Process weekly SLR NP or FR observations and estimate the dynamic orbit positions, station range bias and gravity field coefficients (stored in a normal equation system). Using VCE for stations weighting and outlier detection.

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SlrProcessing





Gravity field coefficients estimation – Process flow in GROOPS



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• Accumulate to a monthly normal system and solve it. Using VCE for relative weighting between the individual normals.

🕨 🛑 program 🛛 👘 NormalsAccumulate

program NormalsSolverVCE

• Evaluate the time variable gravity field and save it as spherical harmonics

program Gravityfield2PotentialCoefficients



Gravity field coefficients estimation – Settings



Using state-of-the-art force models and following further settings:

| Satellites | Ajisai, LAGEOS-1/2, Starlette and Stella | | |
|-------------------------------|---|--|--|
| Station coordinates | ITRF2020 | | |
| Tropospheric refraction model | Mendes & Pavlis | | |
| Center of mass corrections | 1.01 m (Ajisai), 0.251 m (LAGEOS-1/2), 0.075 m (Stella, Starlette) | | |
| Normal points weighting | yes | | |
| Arc | ~ 7 days | | |
| Estimated arc parameters: | | | |
| Atmospheric drag coefficient | Once per day (constant bias for along and cross) | | |
| Empirical accelerations | Once per revolution (sin, cos) along track | | |
| Station range bias | Once per arc | | |
| Satellite state vector | Once per arc | | |
| Gravity field coefficients | Per arc, up to spherical harmonic degree and order 5, and $c_{61}^{}$, $s_{61}^{}$ | | |







2006 2008 2010 2012 2014 2016 2018 2020 2022

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Summary

- Extended the functionality of GROOPS by the feature satellite laser ranging (SLR).
- Utilisation of state-of-the-art models.
- Results of the time variable part degree 2 coefficients are promising and show similar behaviour to Centre for Space Research (CSR) and Austria Academy of Science (AAS) solutions.

Outlook

- We plan to release the GROOPS source code with the SLR functionality in the next few weeks.
- Consistently computation of the next Gravity Observation Combination (GOCO) model.

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FFG



Thank you!



Additional Slides (1)







Force models

| Static gravity field + annual + trend | GOCO06s |
|--|---------------------------|
| Atmosphere + ocean de-alising, atmospheric tides | AOD1B-RL06 |
| Astronomical tides | JPL DE432 |
| Earth tides, pole tides, relativistic effects | IERS2010 |
| Ocean tides | FES2014b |
| Ocean pole tides | Desai 2004 |
| Atmospheric density model | DTM2020 |
| Earth radiation pressure | Albedo and long wave flux |